University of Engineering & Technology Peshawar, Pakistan



CE301: Structure Analysis II

Module 03: Analysis of S.I Frames Using Flexibility method

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Topics to be Covered

- Introduction
- Prerequisites for using flexibility method for Frames
- Revision of Unit Load Method
- Flexibility method procedure for frame analysis
- Analysis of frame Example 1
- Example 2
- Example 3
- Assignment

□ Introduction:

Frames are analyzed with flexibility method due to

- To solve the problem in matrix notation, which is more systematic
- To compute reactions at all the supports.
- To compute internal resisting axial, shear & bending moment at any section of the indeterminate Frame.

The force method of analysis can readily be employed to analyze the indeterminate frames. The basic steps in the analysis of indeterminate frame by force method are the same as that discussed in the analysis of indeterminate beams in the previous lessons. Under the action of external loads, the frames undergo axial and bending deformations. Since the axial rigidity of the members is much higher than the bending rigidity, the axial deformations are much smaller than the bending deformations and are normally not considered in the analysis. The compatibility equations for the frame are written with respect to bending deformations only.

Prerequisites for Analysis with Flexibility method:

- It is necessary that students must have strong background of the following concepts before starting analysis with flexibility or any other matrix method.
- Enough concept of Matrix Algebra
- Must be able to find the Statical Indeterminacy
- Concept of Deflection methods (Unit load method for displacement)

Unit Load Method for Displacement

- This Method is based on the principle of virtual work.
- Best suitable to find the slope & deflection of primary structure when subjected to applied loads .
- Displacement due to external loads at any point *i* is given by

$$\Delta_n = \sum_{i=1}^m \frac{M_i \, mni}{E_i I_i}$$

• Displacement at any point *i* due to a unit load at j is given by (also called flexibility coefficient)

 Where M are the equations of bending moment in primary structure subjected to external loads & m are the equations of BM subjected to unit loads.

$$\delta_{ij} = \sum_{i=1}^{n} \frac{m_i mj}{E_i I_i} dx$$

□ Analysis Procedure

The following steps should be followed to analyze a statical indeterminate frame using flexibility method. Start with finding the degree of statical indeterminacy S.I.

Step # 01: Identify the redundants and obtain BDS also compute
 [DRS] values.
 [DRS] values.
 Assume # of redundants = 2
 No. of redundants = D.S.I
 Just to understand the procedure

Note: for choosing the redundant actions there may be more than one possible options but choose that one as redundant which makes the calculation handy. DRS is initial support rotation or settlement corresponding to redundant locations

- **Step # 02:** Analyze the BDS (primary structure) under the following loading conditions
- Analyze the BDS when acted upon by the external loads and write the equations of bending moment M for different segments.
- ii. Analyze the BDS when acted upon by the unit load at redundant location 1 & write the equations of BM m_1 for different segments.
- iii. Analyze the BDS when acted upon by the unit load at redundant location 2 & write the equations of BM m_2 for different segments.
- iv. And so on if redundants are more than 2.

Note: To write an equation take a section in each segment of loading & then write BM equation with proper sign convention.

Step # 03: Develop Bending moment equation table . BM equations table will be consist of following things which will make the calculation work easy.

Sagmanta	Segments means different members of frame like		
Segments	AB	BC	
Origin	A or B any one can be the origin	e origin B or C any one can be the origin	
Limits (length of segment)	0 to x feet	0 to y feet	
I (Moment of inertia)	Value of I for segment AB	Value of I for segment BC	
М	Equation of M for segment AB	Equation of M for segment BC	
m ₁	Equation of m_1 for segment ABEquation of m_1 for segment B		
m ₂	Equation of m_2 for segment AB Equation of m_2 for segment BC		

• **Step # 04:** Find BDS Displacements due to external loads or Compute the values of DRL

$$DRL_1 = \sum_{i=1}^n \int \frac{M m_1}{E_i I_i} dx$$

$$DRL_2 = \sum_{i=1}^n \int \frac{M m_2}{E_i I_i} dx$$

 Step # 05: Find BDS Displacements due to unit loads or Compute the values of flexibility coefficients & flexibility matrix.

$$f_{11} = \sum_{i=1}^{n} \int \frac{m_1 m_1}{E_i I_i} dx \qquad f_{22} = \sum_{i=1}^{n} \int \frac{m_2 m_2}{E_i I_i} dx$$
$$f_{12} = f_{21} = \sum_{i=1}^{n} \int \frac{m_1 m_2}{E_i I_i} dx$$
$$[f] = \begin{bmatrix} f_{11} & f_{12} \\ f_{21} & f_{22} \end{bmatrix} \qquad \text{Note:} f_{12} = f_2$$

Flexibility Method for Frames Analysis Step # 06: Apply compatibility and principle of superposition at the location of the removed redundants and solve the matrices to compute the values of redundant actions AR. As we know

 $[DRS] = [DRL] + [f] \cdot [AR]$ $[AR] = [f]^{-1} \cdot [DRS - DRL]$ $\begin{bmatrix}AR_{1}\\AR_{2}\end{bmatrix} = \begin{bmatrix}f_{11} & f_{12}\\f_{21} & f_{22}\end{bmatrix}^{-1}\begin{bmatrix}DRS_{1} - DRL_{1}\\DRS_{2} - DRL_{2}\end{bmatrix}$

From this

Step # 07: Compute the member end actions & draw shear force and bending moment diagrams.

[AM] = [AML] + [AMR][AR]

Problem 01: Analyze the given frame using flexibility method.



• **Step # 01:** Identify the redundants and obtain BDS also compute [DRS] values.



Vertical & horizontal reaction at D are chosen as redundant actions



Basic determinate structure (BDS) or Primary structure or Released structure

- **Step # 02:** Analyze the BDS (primary structure) under the following loading conditions
- Analyze the BDS when acted upon by the external loads and write the equations of bending moment M for different segments.
- ii. Analyze the BDS when acted upon by the unit load at redundant location 1 & write the equations of BM m_1 for different segments.
- iii. Analyze the BDS when acted upon by the unit load at redundant location 2 & write the equations of BM m₂ for different segments.

Note: To write an equation take a section in each segment of loading & then write BM equation with proper sign convention.

i. BDS acted upon by the external loads



i. BDS acted upon by unit load at redundant location 1.



i. BDS acted upon by unit load at redundant location 2.



• **Step # 03:** Develop Bending moment equation table.

Segment	AB	BC	CD
Origin	А	С	D
Limits	0 to 15 ft	0 to 30 ft	0 to 15ft
Moment of inertia, I	Ι	21	Ι
M values	5x - 975	-x ²	0
m ₁ values	30	X	0
m ₂ values	Х	15	X

• **Step # 04:** Find BDS Displacements due to external loads or Compute the values of DRL matrix.

$$DRL_1 = \sum_{i=1}^n \int \frac{M m_1}{E_i I_i} dx$$

$$DRL_{1} = \frac{1}{EI} \left[\int_{0}^{15} (5x - 975)(30) dx + \int_{0}^{30} \frac{(-x^{2})(x)}{2} dx + 0 \right]$$
$$DRL_{1} = \frac{-523125}{EI}$$

$$DRL_2 = \sum_{i=1}^n \int \frac{M m_2}{E_i I_i} dx$$

$$DRL_{2} = \frac{1}{EI} \left[\int_{0}^{15} (5x - 975)(x) dx + \int_{0}^{30} \frac{(-x^{2})(15)}{2} dx + 0 \right]$$
$$DRL_{2} = \frac{-171562.5}{FI}$$

$$DRL_{1} = \frac{-523125}{EI} \qquad DRL_{2} = \frac{-171562.5}{EI}$$
$$[DRL] = \begin{bmatrix} DRL_{1} \\ DRL_{2} \end{bmatrix} = \frac{1}{EI} \begin{bmatrix} -523125 \\ -171562.5 \end{bmatrix}$$

• **Step # 05:** Find BDS Displacements due to unit loads

or Compute the values of flexibility coefficients

& flexibility matrix.

$$f_{11} = \sum_{i=1}^{n} \int \frac{m_1 m_1}{E_i I_i} dx$$

$$f_{11} = \frac{1}{EI} \left[\int_0^{15} (30)(30)dx + \int_0^{30} \frac{(x)(x)}{2} dx + 0 \right]$$
$$f_{11} = \frac{18000}{EI}$$

$$f_{22} = \sum_{i=1}^{n} \int \frac{m_2 m_2}{E_i I_i} dx$$

$$f_{12} = f_{21} = \frac{1}{EI} \left[\int_0^{15} (30)(x) dx + \int_0^{30} \frac{(15)(x)}{2} dx + 0 \right]$$

$$f_{12} = f_{21} = \frac{0750}{EI}$$
$$f_{12} = f_{21} = \sum_{i=1}^{n} \int \frac{m_1 m_2}{E_i I_i} d.$$

$$f_{22} = \frac{1}{EI} \left[\int_0^{15} (x)(x) dx + \int_0^{30} \frac{(15)(15)}{2} dx + \int_0^{15} (x)(x) dx \right]$$
$$f_{22} = \frac{5625}{EI}$$

EI

$$f_{11} = \frac{18000}{EI} \qquad \qquad f_{22} = \frac{5625}{EI}$$

$$f_{12} = f_{21} = \frac{6750}{EI}$$

$$[f] = \begin{bmatrix} f_{11} & f_{12} \\ f_{21} & f_{22} \end{bmatrix}$$

$$[f] = \frac{1}{EI} \begin{bmatrix} 18000 & 6750 \\ 6750 & 5625 \end{bmatrix}$$

Flexibility Method for Frames Analysis Step # 06: Apply compatibility and principle of superposition at the locations of the removed redundants and solve the matrices to compute the values of redundant actions AR. As we know

From this

 $[DRS] = [DRL] + [f] \cdot [AR]$ $[AR] = [f]^{-1} \cdot [DRS - DRL]$ $\begin{bmatrix}AR_{1}\\AR_{2}\end{bmatrix} = \begin{bmatrix}f_{11} & f_{12}\\f_{21} & f_{22}\end{bmatrix}^{-1} \begin{bmatrix}DRS_{1} - DRL_{1}\\DRS_{2} - DRL_{2}\end{bmatrix}$

$$\begin{bmatrix} AR_1 \\ AR_2 \end{bmatrix} = \begin{bmatrix} f_{11} & f_{12} \\ f_{21} & f_{22} \end{bmatrix}^{-1} \begin{bmatrix} DRS_1 - DRL_1 \\ DRS_2 - DRL_2 \end{bmatrix}$$

$$\begin{bmatrix} AR_1 \\ AR_2 \end{bmatrix} = EI \begin{bmatrix} 18000 & 6750 \\ 6750 & 5625 \end{bmatrix}^{-1} \begin{bmatrix} 0 - (-523125) \\ 0 - (-171562.5) \end{bmatrix} \times \frac{1}{EI}$$

$$\begin{bmatrix} AR_1 \\ AR_2 \end{bmatrix} = \begin{bmatrix} 32.05 \\ -7.95 \end{bmatrix}$$

-ive sign shows that our assumed redundant action direction is wrong



Final determinate structure

Using Equilibrium equations one can compute the remaining support reactions.



Step # 07: Compute the member end actions & draw SFD & BMD As we know that [AM] = [AML] + [AMR][AR]



Sign convention:

- Horizontal force in +ive X direction will be taken as positive.
- Upward vertical force will be taken as positive.
- Clockwise moment will be taken as positive.

a) Compute AML values.



b) Compute the AMR values: when unit is applied at redundant location 1.



b) Compute the AMR values: when unit is applied at redundant location 2.



$$[AMR]_{6*2} = \begin{bmatrix} AMR_{11} & AMR_{12} \\ AMR_{21} & AMR_{22} \\ AMR_{31} & AMR_{32} \\ AMR_{41} & AMR_{42} \\ AMR_{51} & AMR_{52} \\ AMR_{61} & AMR_{62} \end{bmatrix} = \begin{bmatrix} -1 & 0 \\ 0 & -1 \\ 30 & 0 \\ 1 & 0 \\ 0 & 1 \\ -30 & -15 \end{bmatrix}$$

Now member end actions will be computed as given below

$$\begin{bmatrix} AM_{1} \\ AM_{2} \\ AM_{3} \\ AM_{4} \\ AM_{5} \\ AM_{6} \end{bmatrix} = \begin{bmatrix} AML_{1} \\ AML_{2} \\ AML_{3} \\ AML_{4} \\ AML_{5} \\ AML_{6} \end{bmatrix} + \begin{bmatrix} AMR_{11} & AMR_{12} \\ AMR_{21} & AMR_{22} \\ AMR_{31} & AMR_{32} \\ AMR_{31} & AMR_{32} \\ AMR_{41} & AMR_{42} \\ AMR_{51} & AMR_{52} \\ AMR_{51} & AMR_{52} \\ AMR_{61} & AMR_{62} \end{bmatrix} \begin{bmatrix} AR_{1} \\ AR_{2} \end{bmatrix}$$
So member end actions will be

$$\begin{bmatrix} AM_1 \\ AM_2 \\ AM_2 \\ AM_3 \\ AM_4 \\ AM_5 \\ AM_6 \end{bmatrix} = \begin{bmatrix} 60 \\ -5 \\ -975 \\ -60 \\ 5 \\ 900 \end{bmatrix} + \begin{bmatrix} -1 & 0 \\ 0 & -1 \\ 30 & 0 \\ 1 & 0 \\ 0 & 1 \\ -30 & -15 \end{bmatrix} \begin{bmatrix} 32.05 \\ -7.95 \end{bmatrix} = \begin{bmatrix} 27.95 \\ 2.95 \\ -13.5 \\ -27.95 \\ -27.95 \\ 57.75 \end{bmatrix}$$

So member end actions, Shear force & Bending moment diagrams of member AB



Member BC: member end actions for member BC are specified below



a) Compute AML values.



b) Compute the AMR values: when unit is applied at redundant location 1.



b) Compute the AMR values: when unit is applied at redundant location 2.



$$[AMR]_{6*2} = \begin{bmatrix} AMR_{11} & AMR_{12} \\ AMR_{21} & AMR_{22} \\ AMR_{31} & AMR_{32} \\ AMR_{41} & AMR_{42} \\ AMR_{51} & AMR_{52} \\ AMR_{61} & AMR_{62} \end{bmatrix} = \begin{bmatrix} -1 & 0 \\ 0 & -1 \\ 30 & 15 \\ 1 & 0 \\ 0 & 1 \\ 0 & -15 \end{bmatrix}$$

So member end actions will be

$$\begin{bmatrix} AM_1 \\ AM_2 \\ AM_3 \\ AM_4 \\ AM_5 \\ AM_6 \end{bmatrix} = \begin{bmatrix} 60 \\ 0 \\ -900 \\ 0 \\ 0 \end{bmatrix} + \begin{bmatrix} -1 & 0 \\ 0 & -1 \\ 30 & 15 \\ 1 & 0 \\ 0 & -15 \end{bmatrix} \begin{bmatrix} 32.05 \\ -7.95 \end{bmatrix} = \begin{bmatrix} 27.95 \\ 7.59 \\ -57.75 \\ 32.05 \\ -7.95 \\ 119.25 \end{bmatrix}$$

So member end actions of member BC





Member BC:



Member CD: member end actions for member CD are specified below



a) Compute AML values.





b) Compute the AMR values: when unit is applied at redundant location 2.



$$[AMR]_{6*2} = \begin{bmatrix} AMR_{11} & AMR_{12} \\ AMR_{21} & AMR_{22} \\ AMR_{31} & AMR_{32} \\ AMR_{41} & AMR_{42} \\ AMR_{51} & AMR_{52} \\ AMR_{61} & AMR_{62} \end{bmatrix} = \begin{bmatrix} 1 & 0 \\ 0 & 1 \\ 0 & 0 \\ -1 & 0 \\ 0 & -1 \\ 0 & 15 \end{bmatrix}$$

So member end actions will be

$$\begin{bmatrix} AM_{1} \\ AM_{2} \\ AM_{2} \\ AM_{3} \\ AM_{4} \\ AM_{5} \\ AM_{6} \end{bmatrix} = \begin{bmatrix} 0 \\ 0 \\ 0 \\ 0 \\ 0 \\ 0 \end{bmatrix} + \begin{bmatrix} 1 & 0 \\ 0 & 1 \\ 0 & 0 \\ -1 & 0 \\ 0 & -1 \\ 0 & 15 \end{bmatrix} \begin{bmatrix} 32.05 \\ -7.95 \\ 0 \\ -7.95 \end{bmatrix} = \begin{bmatrix} 32.05 \\ -7.95 \\ 0 \\ -32.05 \\ 7.95 \\ -119.25 \end{bmatrix}$$

So member end actions, Shear force & bending moment diagrams of member CD



Combined shear force & bending moment diagrams:



Problem 02: Analyze the given frame using flexibility method such that support D translate to right by 0.5in and also settle down by 0.75in.



Flexibility Method for Frames Analysis Step # 01: Identify the redundants and obtain BDS also compute • [DRS] values. 2k/ft5k **30ft** 2IVertical & horizontal $[AR] = \begin{bmatrix} AR_1 \\ AR_2 \end{bmatrix} = \begin{bmatrix} ? \\ ? \end{bmatrix}$ reaction at D are chosen as redundant 15ft I actions $\rightarrow AR_2$ AR_1 $[DRS] = \begin{bmatrix} DRS_1 \\ DRS_2 \end{bmatrix} = \begin{bmatrix} -0.75'' \\ 0.5'' \end{bmatrix} = \begin{bmatrix} -0.0625 \\ 0.0416 \end{bmatrix}$

- **Step # 02:** Analyze the BDS (primary structure) under the following loading conditions
- Analyze the BDS when acted upon by the external loads and write the equations of bending moment M for different segments.
- ii. Analyze the BDS when acted upon by the unit load at redundant location 1 & write the equations of BM m_1 for different segments.
- iii. Analyze the BDS when acted upon by the unit load at redundant location 2 & write the equations of BM m₂ for different segments.

Note: To write an equation take a section in each segment of loading & then write BM equation with proper sign convention.

i. BDS acted upon by the external loads



i. BDS acted upon by unit load at redundant location 1.



i. BDS acted upon by unit load at redundant location 2.



• **Step # 03:** Develop Bending moment equation table.

Segment	AB	BC	CD
Origin	А	В	С
Limits	0 to 15 ft	0 to 30 ft	0 to 15ft
Moment of inertia, I	Ι	21	Ι
M values	-975 + 5x	-x ²	0
m ₁ values	30	Х	0
m ₂ values	Х	15	Х

• **Step # 04:** Find BDS Displacements due to external loads or Compute the values of DRL matrix.

$$DRL_1 = \sum_{i=1}^n \int \frac{M m_1}{E_i I_i} dx$$

$$DRL_{1} = \frac{1}{EI} \left[\int_{0}^{15} (5x - 975)(30) dx + \int_{0}^{30} \frac{(-x^{2})(x)}{2} dx + 0 \right]$$
$$DRL_{1} = \frac{-523125}{EI}$$

$$DRL_{2} = \sum_{i=1}^{n} \int \frac{M m_{2}}{E_{i}I_{i}} dx$$
$$DRL_{2} = \frac{1}{EI} \left[\int_{0}^{15} (5x - 975)(x) dx + \int_{0}^{30} \frac{(-x^{2})(15)}{2} dx + 0 \right]$$
$$DRL_{2} = \frac{-171562.5}{EI}$$

$$[DRL] = \begin{bmatrix} DRL_1 \\ DRL_2 \end{bmatrix} = \frac{1}{EI} \begin{bmatrix} -523125 \\ -171562.5 \end{bmatrix} = \frac{1}{125000} \begin{bmatrix} -523125 \\ -171562.5 \end{bmatrix}$$

 $\begin{bmatrix} DRL \end{bmatrix} = \begin{bmatrix} DRL_1 \\ DRL_2 \end{bmatrix} = \begin{bmatrix} -4.185 \\ -1.372 \end{bmatrix}$

 Step # 05: Find BDS Displacements due to unit loads or Compute the values of flexibility coefficients & flexibility matrix.

$$f_{11} = \sum_{i=1}^{n} \int \frac{m_1 m_1}{E_i I_i} dx$$

$$f_{11} = \frac{1}{EI} \left[\int_0^{15} (30)(30)dx + \int_0^{30} \frac{(x)(x)}{2} dx + 0 \right]$$
$$f_{11} = \frac{18000}{EI}$$

$$f_{22} = \sum_{i=1}^{n} \int \frac{m_2 m_2}{E_i I_i} dx$$

$$f_{12} = f_{21} = \frac{1}{EI} \left[\int_0^{15} (30)(x) dx + \int_0^{30} \frac{(15)(x)}{2} dx + 0 \right]$$

$$f_{12} = f_{21} = \frac{0750}{EI}$$
$$f_{12} = f_{21} = \sum_{i=1}^{n} \int \frac{m_1 m_2}{E_i I_i} dx$$

$$f_{22} = \frac{1}{EI} \left[\int_0^{15} (x)(x) dx + \int_0^{30} \frac{(15)(15)}{2} dx + \int_0^{15} (x)(x) dx \right]$$
$$f_{22} = \frac{5625}{EI}$$

$$[f] = \begin{bmatrix} f_{11} & f_{12} \\ f_{21} & f_{22} \end{bmatrix}$$

 $[f] = \frac{1}{EI} \begin{bmatrix} 18000 & 6750 \\ 6750 & 5625 \end{bmatrix}$

 $[f] = \frac{1}{125000} \begin{bmatrix} 18000 & 6750 \\ 6750 & 5625 \end{bmatrix}$

 $[f] = \begin{bmatrix} 0.144 & 0.054 \\ 0.054 & 0.045 \end{bmatrix}$

Flexibility Method for Frames Analysis Step # 06: Apply compatibility and principle of superposition at the location of the removed redundants and solve the matrices to compute the values of redundant actions AR. As we know

 $[DRS] = [DRL] + [f] \cdot [AR]$ $[AR] = [f]^{-1} \cdot [DRS - DRL]$ $\begin{bmatrix}AR_{1}\\AR_{2}\end{bmatrix} = \begin{bmatrix}f_{11} & f_{12}\\f_{21} & f_{22}\end{bmatrix}^{-1}\begin{bmatrix}DRS_{1} - DRL_{1}\\DRS_{2} - DRL_{2}\end{bmatrix}$

From this

$$\begin{bmatrix} AR_1 \\ AR_2 \end{bmatrix} = \begin{bmatrix} f_{11} & f_{12} \\ f_{21} & f_{22} \end{bmatrix}^{-1} \begin{bmatrix} DRS_1 - DRL_1 \\ DRS_2 - DRL_2 \end{bmatrix}$$

$$\begin{bmatrix} AR_1 \\ AR_2 \end{bmatrix} = EI \begin{bmatrix} 0.144 & 0.054 \\ 0.054 & 0.045 \end{bmatrix}^{-1} \begin{bmatrix} -0.0625 - (-4.185) \\ 0.04167 - (-1.3725) \end{bmatrix}$$

$$\begin{bmatrix} AR_1 \\ AR_2 \end{bmatrix} = \begin{bmatrix} 30.62 \\ -5.32 \end{bmatrix}$$
-ive sign shared undant a

-ive sign shows that our assumed redundant action direction is wrong



Final determinate structure

Step # 07: Compute the member end actions also draw the SFD & BMD As we know that

[AM] = [AML] + [AMR][AR]

Member AB:



a) Compute AML values.



b) Compute the AMR values: when unit is applied at redundant location 1.



b) Compute the AMR values: when unit is applied at redundant location 2.



$$[AMR]_{6*2} = \begin{bmatrix} AMR_{11} & AMR_{12} \\ AMR_{21} & AMR_{22} \\ AMR_{31} & AMR_{32} \\ AMR_{41} & AMR_{42} \\ AMR_{51} & AMR_{52} \\ AMR_{61} & AMR_{62} \end{bmatrix} = \begin{bmatrix} -1 & 0 \\ 0 & -1 \\ 30 & 0 \\ 1 & 0 \\ 0 & 1 \\ -30 & -15 \end{bmatrix}$$

$$\begin{bmatrix} AM_{1} \\ AM_{2} \\ AM_{2} \\ AM_{3} \\ AM_{4} \\ AM_{5} \\ AM_{6} \end{bmatrix} = \begin{bmatrix} 60 \\ -5 \\ -975 \\ -60 \\ 5 \\ 900 \end{bmatrix} + \begin{bmatrix} -1 & 0 \\ 0 & -1 \\ 30 & 0 \\ 1 & 0 \\ 0 & 1 \\ -30 & -15 \end{bmatrix} \begin{bmatrix} 30.62 \\ -5.32 \end{bmatrix} = \begin{bmatrix} 29.38 \\ 0.32 \\ -56.4 \\ -29.38 \\ -0.32 \\ 61.20 \end{bmatrix}$$
So member end actions, Shear force & Bending moment diagrams of member AB



Member BC: member end actions for member BC are specified below



a) Compute AML values.



b) Compute the AMR values: when unit is applied at redundant location 1.



b) Compute the AMR values: when unit is applied at redundant location 2.



$$[AMR]_{6*2} = \begin{bmatrix} AMR_{11} & AMR_{12} \\ AMR_{21} & AMR_{22} \\ AMR_{31} & AMR_{32} \\ AMR_{41} & AMR_{42} \\ AMR_{51} & AMR_{52} \\ AMR_{61} & AMR_{62} \end{bmatrix} = \begin{bmatrix} -1 & 0 \\ 0 & -1 \\ 30 & 15 \\ 1 & 0 \\ 0 & 1 \\ 0 & -15 \end{bmatrix}$$

So member end actions will be

$$\begin{bmatrix} AM_1 \\ AM_2 \\ AM_3 \\ AM_4 \\ AM_5 \\ AM_6 \end{bmatrix} = \begin{bmatrix} 60 \\ 0 \\ -900 \\ 0 \\ 0 \\ 0 \end{bmatrix} + \begin{bmatrix} -1 & 0 \\ 0 & -1 \\ 30 & 15 \\ 1 & 0 \\ 0 & -15 \end{bmatrix} \begin{bmatrix} 30.62 \\ -5.32 \end{bmatrix} = \begin{bmatrix} 29.38 \\ 5.32 \\ -61.20 \\ 30.62 \\ -5.32 \\ 79.8 \end{bmatrix}$$

So member end actions of member BC





Member CD: member end actions for member CD are specified below



a) Compute AML values.





b) Compute the AMR values: when unit is applied at redundant location 2.



$$[AMR]_{6*2} = \begin{bmatrix} AMR_{11} & AMR_{12} \\ AMR_{21} & AMR_{22} \\ AMR_{31} & AMR_{32} \\ AMR_{41} & AMR_{42} \\ AMR_{51} & AMR_{52} \\ AMR_{61} & AMR_{62} \end{bmatrix} = \begin{bmatrix} 1 & 0 \\ 0 & 1 \\ 0 & 0 \\ -1 & 0 \\ 0 & -1 \\ 0 & 15 \end{bmatrix}$$

So member end actions will be

$$\begin{bmatrix} AM_{1} \\ AM_{2} \\ AM_{2} \\ AM_{3} \\ AM_{4} \\ AM_{5} \\ AM_{6} \end{bmatrix} = \begin{bmatrix} 0 \\ 0 \\ 0 \\ 0 \\ 0 \\ 0 \end{bmatrix} + \begin{bmatrix} 1 & 0 \\ 0 & 1 \\ 0 & 0 \\ -1 & 0 \\ 0 & -1 \\ 0 & 15 \end{bmatrix} \begin{bmatrix} 30.62 \\ -5.32 \\ -5.32 \end{bmatrix} = \begin{bmatrix} 30.62 \\ -5.32 \\ 0 \\ -30.62 \\ 5.32 \\ -79.8 \end{bmatrix}$$

So member end actions, Shear force & bending moment diagrams of member CD



Combined shear force & bending moment diagrams:



Problem 03: Analyze the given frame using flexibility method.



• **Step # 01:** Identify the redundants and obtain BDS also compute [DRS] values.

Vertical & horizontal reaction at C are chosen as redundant actions





Basic determinate structure (BDS) or Primary structure or Released structure

- **Step # 02:** Analyze the BDS (primary structure) under the following loading conditions
- Analyze the BDS when acted upon by the external loads and write the equations of bending moment M for different segments.
- ii. Analyze the BDS when acted upon by the unit load at redundant location 1 & write the equations of BM m_1 for different segments.
- iii. Analyze the BDS when acted upon by the unit load at redundant location 2 & write the equations of BM m₂ for different segments.

Note: To write an equation take a section in each segment of loading & then write BM equation with proper sign convention.

i. BDS acted upon by the external loads



i. BDS acted upon by unit load at redundant location 1.



i. BDS acted upon by unit load at redundant location 2.



• **Step # 03:** Develop Bending moment equation table.

Segment	AB	BC
Origin	А	В
Limits	0 to 8 ft	0 to 16 ft
Moment of inertia, I	Ι	21
M values	-40 + 5x	0
m ₁ values	-16	-X
m ₂ values	8 - x	0

• **Step # 04:** Find BDS Displacements due to external loads or Compute the values of DRL matrix.

$$DRL_1 = \sum_{i=1}^n \int \frac{M m_1}{E_i I_i} dx$$

$$DRL_1 = \frac{1}{EI} \left[\int_0^8 (5x - 40)(-16) dx + 0 \right]$$

$$DRL_1 = \frac{2560}{EI}$$

$$DRL_{2} = \sum_{i=1}^{n} \int \frac{M m_{2}}{E_{i}I_{i}} dx$$
$$DRL_{2} = \frac{1}{EI} \left[\int_{0}^{8} (5x - 40)(8 - x) dx + 0 \right]$$

$$DRL_2 = \frac{-853.33}{EI}$$

$$DRL_1 = \frac{2560}{EI}$$
 $DRL_2 = \frac{-853.33}{EI}$

$$[DRL] = \begin{bmatrix} DRL_1 \\ DRL_2 \end{bmatrix} = \frac{1}{EI} \begin{bmatrix} 2560 \\ -853.33 \end{bmatrix}$$

 Step # 05: Find BDS Displacements due to unit loads or Compute the values of flexibility coefficients & flexibility matrix.

$$f_{11} = \sum_{i=1}^{n} \int \frac{m_1 m_1}{E_i I_i} dx$$

$$f_{11} = \frac{1}{EI} \left[\int_0^8 (-16)(-16)dx + \int_0^{16} \frac{(-x)(-x)}{2} dx \right]$$

$$f_{11} = \frac{2730.67}{EI}$$

$$f_{22} = \sum_{i=1}^{n} \int \frac{m_2 m_2}{E_i I_i} dx$$

$$f_{12} = f_{21} = \frac{1}{EI} \left[\int_0^8 (-16)(8-x) dx + 0 \right]$$

$$f_{12} = f_{21} = \frac{-512}{EI}$$

$$f_{12} = f_{21} = \sum_{i=1}^n \int \frac{m_1 m_2}{E_i I_i} dx$$

$$f_{22} = \frac{1}{EI} \left[\int_0^8 (8-x)(8-x) dx + 0 \right]$$

$$170.67$$

$$f_{22} = \frac{17000}{EI}$$

$$f_{11} = \frac{2730.67}{EI}$$
 $f_{22} = \frac{170.67}{EI}$

$$f_{12} = f_{21} = \frac{-512}{EI}$$

$$[f] = \begin{bmatrix} f_{11} & f_{12} \\ f_{21} & f_{22} \end{bmatrix}$$

$$[f] = \frac{1}{EI} \begin{bmatrix} 2730.67 & -512 \\ -512 & 170.67 \end{bmatrix}$$

Flexibility Method for Frames Analysis Step # 06: Apply compatibility and principle of superposition at the location of the removed redundants and solve the matrices to compute the values of redundant actions AR. As we know

 $[DRS] = [DRL] + [f] \bullet [AR]$ $[AR] = [f]^{-1} \bullet [DRS - DRL]$

From this

 $\begin{bmatrix} AR_1 \\ AR_2 \end{bmatrix} = \begin{bmatrix} f_{11} & f_{12} \\ f_{21} & f_{22} \end{bmatrix}^{-1} \begin{bmatrix} DRS_1 - DRL_1 \\ DRS_2 - DRL_2 \end{bmatrix}$

$$\begin{bmatrix} AR_1 \\ AR_2 \end{bmatrix} = \begin{bmatrix} f_{11} & f_{12} \\ f_{21} & f_{22} \end{bmatrix}^{-1} \begin{bmatrix} DRS_1 - DRL_1 \\ DRS_2 - DRL_2 \end{bmatrix}$$

 $\begin{bmatrix} AR1\\ AR2 \end{bmatrix} = EI \begin{bmatrix} 2730.67 & -512\\ -512 & 170.67 \end{bmatrix}^{-1} \begin{bmatrix} 0 - (2560)\\ 0 - (-853.33) \end{bmatrix} \times \frac{1}{EI}$

 $\begin{bmatrix} AR_1 \\ AR_2 \end{bmatrix} = \begin{bmatrix} 0 \\ 5 \end{bmatrix}$



Final determinate structure

Step # 07: Compute the member end actions. Also Draw shear force and bending moment diagrams As we know that

[AM] = [AML] + [AMR][AR]

Member AB:



a) Compute AML values.



b) Compute the AMR values: when unit is applied at redundant location 1.



b) Compute the AMR values: when unit is applied at redundant location 2.



$$[AMR]_{6*2} = \begin{bmatrix} AMR_{11} & AMR_{12} \\ AMR_{21} & AMR_{22} \\ AMR_{31} & AMR_{32} \\ AMR_{41} & AMR_{42} \\ AMR_{51} & AMR_{52} \\ AMR_{61} & AMR_{62} \end{bmatrix} = \begin{bmatrix} 1 & 0 \\ 0 & 1 \\ -16 & 8 \\ -1 & 0 \\ 0 & -1 \\ 16 & 0 \end{bmatrix}$$

So member end actions will be

$$\begin{bmatrix} AM_1 \\ AM_2 \\ AM_3 \\ AM_4 \\ AM_5 \\ AM_6 \end{bmatrix} = \begin{bmatrix} 10 \\ -5 \\ -40 \\ -10 \\ 5 \\ 0 \end{bmatrix} + \begin{bmatrix} 1 & 0 \\ 0 & 1 \\ -16 & 8 \\ -1 & 0 \\ 0 & -1 \\ 16 & 0 \end{bmatrix} \begin{bmatrix} 0 \\ 5 \end{bmatrix} = \begin{bmatrix} 10 \\ 0 \\ 0 \\ -10 \\ 0 \\ 0 \end{bmatrix}$$
So member end actions, Shear force & bending moment diagrams of member AB



Member BC: member end actions for member BC are specified below



a) Compute AML values.



b) Compute the AMR values: when unit is applied at redundant location 1.



b) Compute the AMR values: when unit is applied at redundant location 2.



$$[AMR]_{6*2} = \begin{bmatrix} AMR_{11} & AMR_{12} \\ AMR_{21} & AMR_{22} \\ AMR_{31} & AMR_{32} \\ AMR_{41} & AMR_{42} \\ AMR_{51} & AMR_{52} \\ AMR_{61} & AMR_{62} \end{bmatrix} = \begin{bmatrix} 1 & 0 \\ 0 & 1 \\ -16 & 0 \\ -1 & 0 \\ 0 & -1 \\ 0 & 0 \end{bmatrix}$$

So member end actions will be

$$\begin{bmatrix} AM_{1} \\ AM_{2} \\ AM_{3} \\ AM_{4} \\ AM_{5} \\ AM_{6} \end{bmatrix} = \begin{bmatrix} 0 \\ 0 \\ 0 \\ 0 \\ 0 \\ 0 \end{bmatrix} + \begin{bmatrix} 1 & 0 \\ 0 & 1 \\ -16 & 0 \\ -1 & 0 \\ 0 & -1 \\ 0 & 0 \end{bmatrix} \begin{bmatrix} 0 \\ 5 \end{bmatrix} = \begin{bmatrix} 0 \\ 5 \\ 0 \\ 0 \\ -5 \\ 0 \end{bmatrix}$$

So member end actions of member BC



Shear force & bending moment diagrams of member BC





Complete analyzed structure

Combined shear force & bending moment diagram:



Assignment 02: Analyze the given frames using flexibility method Note:

- Section B have to solve problem 4 & 5.
- Section C have to solve problem 6 & 7.



Problem 05:



Problem 06:



Problem 07:



References

- Structural Analysis by R. C. Hibbeler
- Matrix structural analysis by William Mc Guire
- Matrix analysis of frame structures by William Weaver
- Online Civil Engineering blogs