

Quiz/Assignment

Note: Attempt all questions. Draw sketches where necessary. Assume missing data if any.

Q.NO (01)

(5+5)

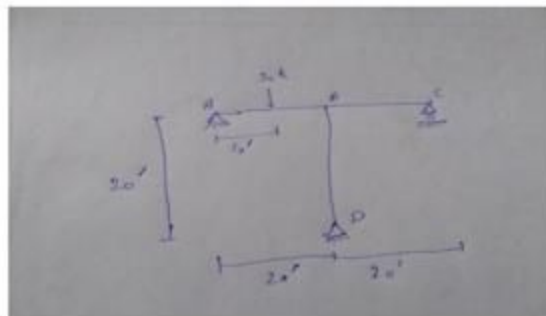
- Expalin description of flexibility method procedure.
- Differentiate between flexibility and stiffness method.

Q.NO (02)

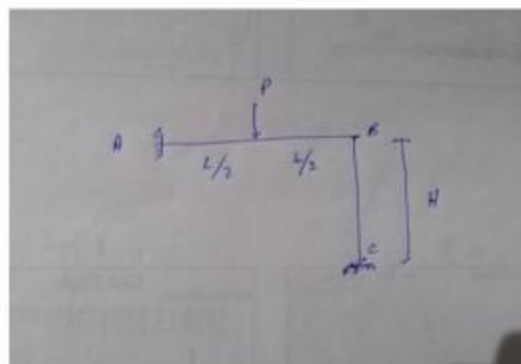
(10+10)

Analyse the frames as shown in figures by using flexibility method.

a)



b)



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Paper :- Advanced Structural
Analysis -

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Q#1 Part(A)

→ Explain description of flexibility method procedure -

Description of flexibility Method procedure -

① Determine the degree of structural indeterminacy (SI) - A number of releases equal to the degree of S.I. are applied to the structure - each released beam made by the removal of internal or external force -
 The ~~rest~~ release structure is referred to as a primary structure - The release structure must be chosen such that the remaining structure is geometrically stable and S-determinate
 In some case the no. of release can be less than the degree of indeterminacy.

② In all cases the redundant forces should be carefully chosen so that primary structure are carefully analyzed -

③ The release introduce displacement inconsistency into the structure and as a 2nd step these inconsistency are error in the primary structure are determine - In other

woods, we calculate the error in the displacement corresponding to the R-value. These displacements may be due to external applied loads.

④ The 3rd step consist of determination of displacement in primary structure due to unit value of redundants. These displacements are required at some location and in the some direction as the displacement error determine in step 2.

⑤ Value of the redundant force necessary to determine the error in the displacement. This require the writing of super position in which the effects of separate redundant are added to the displacement.

The super position at displacement result in the set of simultaneous linear eq - (No of release) that express the fact that there is zero relative displacement at each release.

These compatibility equations guarantee a final displaced shape.

Part # B

→ Differentiate between flexibility and Stiffness method -

Flexibility Method

Stiffness Method -

- | | |
|---|--|
| <p>① Determine the degree of static indeterminacy (degree of redundancy) n -</p> <p>② choose the redundants</p> <p>③ Assign co-ordinate 1, 2, ... n to the redundants -</p> <p>④ Remove all redundants to obtain the released structure</p> <p>⑤ Determine (A) the displacements at the coordinate due to applied load acting on the released structure -</p> <p>⑥ Determine (A_R) the displacements at the coordinate due to the redundants acting on the released structure -</p> | <p>① Determine the degree of kinematic indeterminacy (degree of freedom) -</p> <p>② identify the independent displacement component -</p> <p>③ Assign coordinate 1, 2, ... to the independent displacement component -</p> <p>④ Prevent all the independent displacement components to obtain the restrained structure -</p> <p>⑤ Determine (P_i) the force at the coordinate in the restrained structure due to the load other than those acting at the coordinate -</p> <p>⑥ Determine (P_A) the force required at the coordinate in the unrestrained structure to cause the independent displacement components - (A)</p> |
|---|--|

⑦ compute the net displacement at the coordinate

$$\Delta = [\Delta_L] + [\Delta_R]$$

⑦ compute the force at the coordinate

$$[P] = [P'] + [P_\Delta]$$

⑧ Use the conditions of compatibility of displacement to compute the redundant force

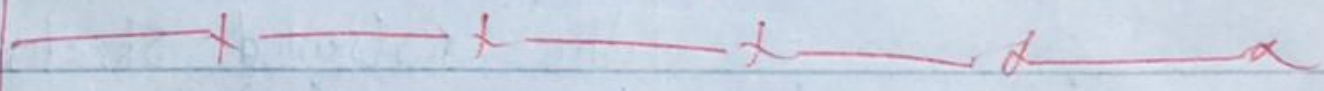
$$[P] = [A]^{-1} \{A - \Delta L\}$$

⑧ Use the condition of equilibrium of displacement to compute the displacement

$$[A] = [K]^{-1} \{P - P'\}$$

9 knowing the redundant compute the internal member force by using equation of statics.

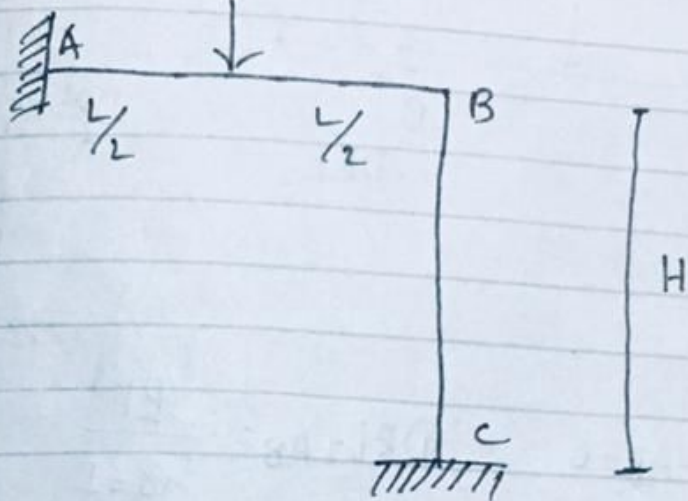
knowing the displacement compute the internal member force by using slope deflection equation -



Q#2)

(Part # B)

Analyse the frame by using flexibility method!



Sol:

- Degree of static indeterminacy:

$$D_s = (3M + R) - (3j + A)$$

$$D_s = (3(2) + 6) - (3(3) + 0) \quad M=2$$

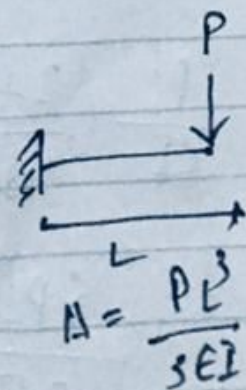
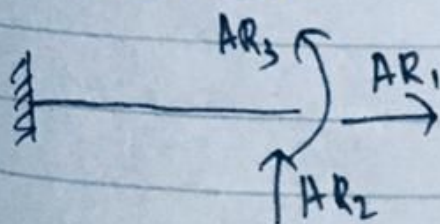
$$R=6$$

$$D_s = (6+6) - 9$$

$$j=3$$

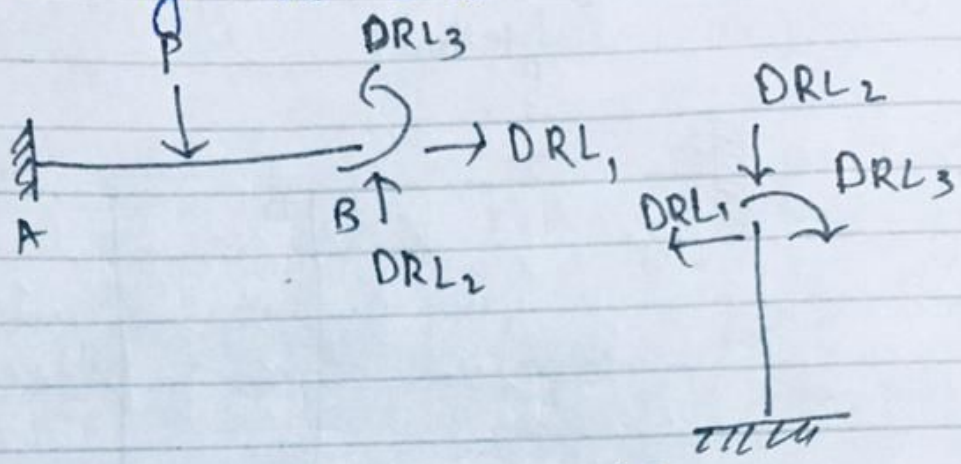
$$D_s = 12 - 9$$

$$D_s = 3$$

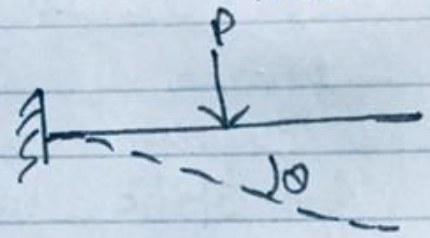


displacement.

Displacement in released structure caused by the load -



$$DRL_1 = \Delta_B = 0, \quad DRL_2 \Delta_B = \frac{5PL^3}{48EI}$$



$$\frac{P(L/3)^3}{3EI} \quad \Delta = \frac{P(L/2)^2}{2EI}$$

$$\Delta = \Delta_1 + \Delta_2$$

$$= \frac{PL^3}{24EI} + \frac{PL^3}{16EI} = \frac{PL^3}{8EI}$$

$$\Delta_2 = \frac{PL^2}{8EI} \times \frac{L}{2} = \frac{PL^3}{16EI}$$

$$= \frac{PL^3}{EI} \left(\frac{2+3}{48} \right)$$

$$= \frac{5PL^3}{48EI}$$

$$DRL_3(AB) = \frac{-PL^3}{8EI}$$

$DRL_1(BC) = 0$ (Because there is no Load.)

$$DRL_1 = 0 + 0 = 0$$

$$DRL_2 = \frac{-5PL^3}{48EI} + 0 = \frac{-5PL^3}{48EI}$$

$$DRL_3 = \frac{-PL^2}{8EI} + 0 = \frac{-PL^2}{8EI}$$

$$[DRL] = \begin{bmatrix} 0 \\ -5PL^3/48EI \\ -PL^2/8EI \end{bmatrix}$$

flexibility matrix:-

Apply $AR_1 = 1$ and obtain F_{11}, F_{21}, F_{31} the displacements at End B at member AB.

$$F_{11AB} = \frac{L}{EA}, \quad F_{21AB} = 0, \quad F_{31AB} = 0.$$

The displacement at End B at member BC

$$F_{11BC} = \frac{H^3}{3EI}, \quad F_{21(BC)} = 0$$

$$F_{31} = \frac{-H^3}{2EI}$$

The final value of flexibility co-eff are

$$F_{11} = \frac{L}{EA} + \frac{H^3}{3EI}$$

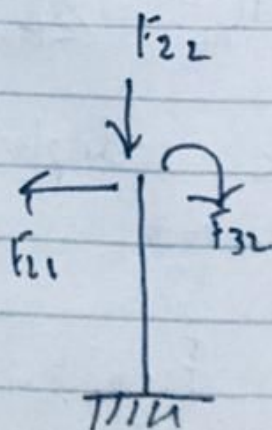
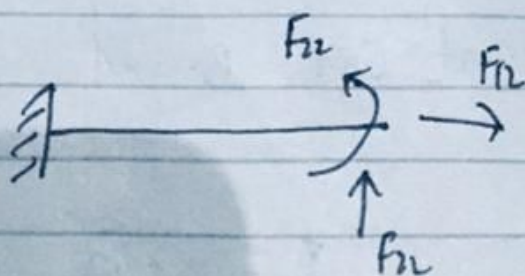
$$F_{21} = 0 + 0 = 0$$

$$F_{31} = \frac{0 - H^2}{2EI} = \boxed{\frac{-H^2}{2EI}}$$

$$[F] = \begin{bmatrix} \frac{L}{EA} + \frac{H^3}{3EI} & 0 & \frac{-H^2}{2EI} \\ 0 & \frac{L^3}{3EI} + \frac{H}{EA} & \frac{L^2}{2EI} \\ \frac{-H^2}{2EI} & \frac{L^2}{2EI} & \frac{L}{EI} + \frac{H}{EA} \end{bmatrix} \quad \text{--- (A)}$$

Apply $HR_2 = 1$ and obtain

F_{12}, F_{22}, F_{32}



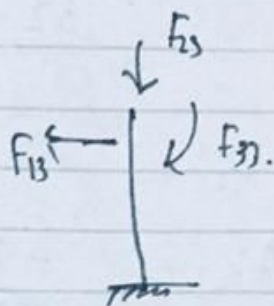
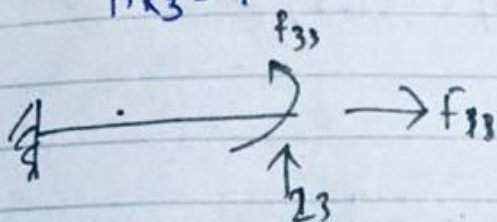
$$F_{12} = 0$$

$$F_{22} = \frac{L^3}{3EI} + \frac{H}{EA}$$

$$F_{32} = -\frac{L^2}{2EI}$$

Now

$$AR_3 = 1$$



$$F_{13} = \frac{-H^2}{2EI}$$

$$F_{23} = \frac{L^2}{2EI}$$

$$F_{33} = \frac{L}{EI} + \frac{H}{EA}$$

New final F Matrix - is (A)

$$[F]_2 = \begin{bmatrix} \frac{L}{EA} + \frac{H^3}{3EI} & 0 & \frac{-H^2}{2EI} \\ 0 & \frac{L^3}{3EI} + \frac{H}{EA} & \frac{L^2}{2EI} \\ \frac{-H^2}{2EI} & \frac{L^2}{2EI} & \frac{L}{EI} + \frac{H}{EA} \end{bmatrix}$$

$$AR_1 = \frac{-3P}{32} \left[\frac{1-12\gamma}{(1+3\gamma)(1+12\gamma)} \right]$$

where $\gamma = \frac{1}{AL_3}$

$$AR_2 = \frac{13P}{32} \left[\frac{1+84\gamma}{(1+3\gamma)(1+12\gamma)} \right]$$

$$AR_3 = \frac{-PL}{16} \left[\frac{1-12\gamma}{1+12\gamma} \right]$$

for typically plane frame Magnitude of γ is 10^{-3}

$$Q_1 = \frac{-3P}{32}$$

$$Q_2 = \frac{13P}{32}$$

$$Q_3 = \frac{-PL}{16}$$

$$\begin{bmatrix} 0.127 & 0 & -0.0019 \\ 0 & 0.0519 & 0.0048 \\ -0.0019 & 0.0048 & 0.0018 \end{bmatrix}$$

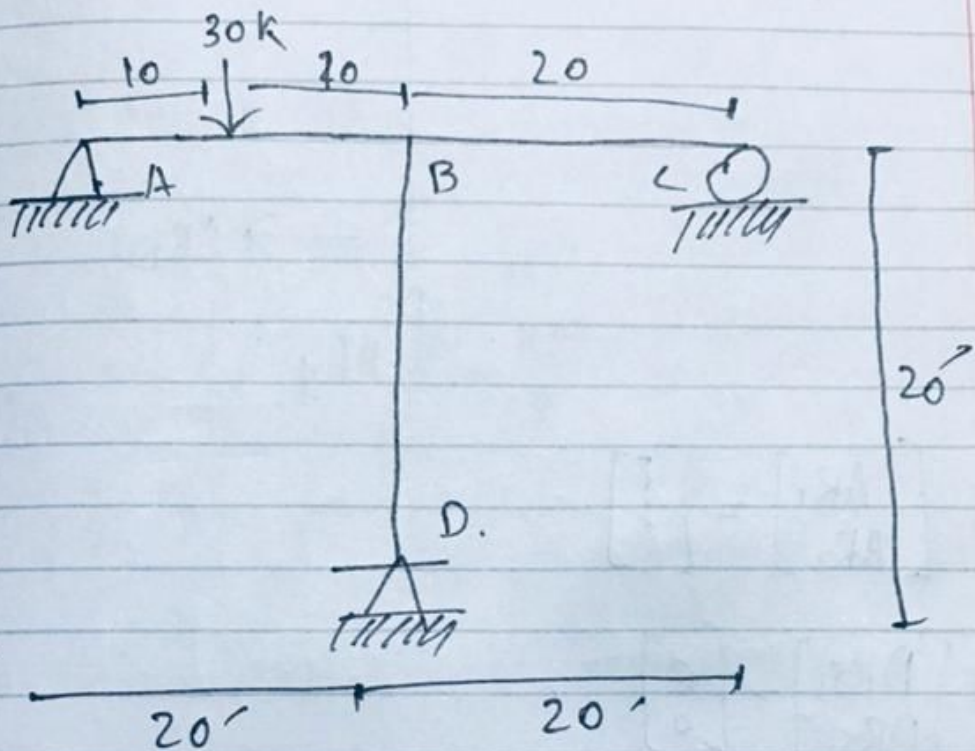
$$AR_1 = -2.50$$

$$AR_2 = 6.25$$

$$AR_3 = -16.76$$

Q# 2 Part # (A)

Analyse the frame by flexibility Method :-



Sol:-

Degree of static indeterminacy -

$$D_s = (3M + R) - (3J + A)$$

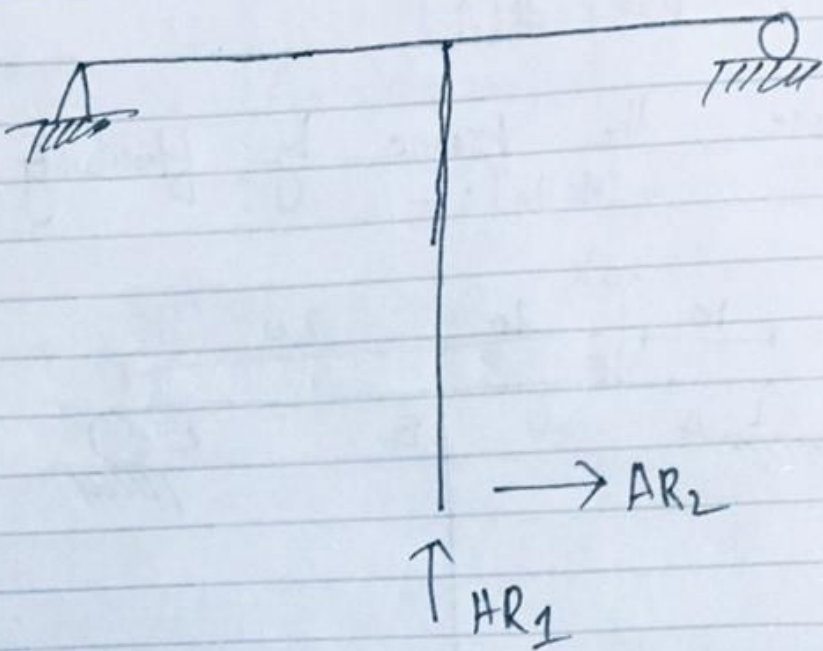
$$D_s = (3(3) + 5) - (3(4) + 0)$$

$$\begin{aligned} M &= 3 \\ R &= 5 \\ J &= 4 \end{aligned}$$

$$D_s = (9 + 5) - (12)$$

$$D_s = 14 - 12$$

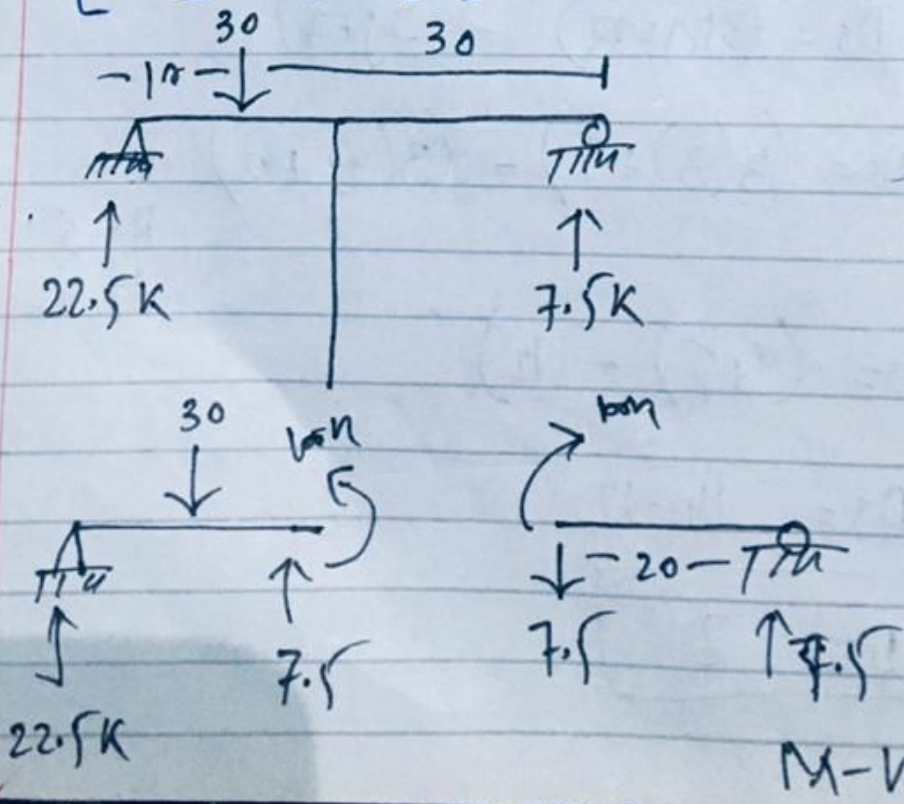
$$D_s = 2$$



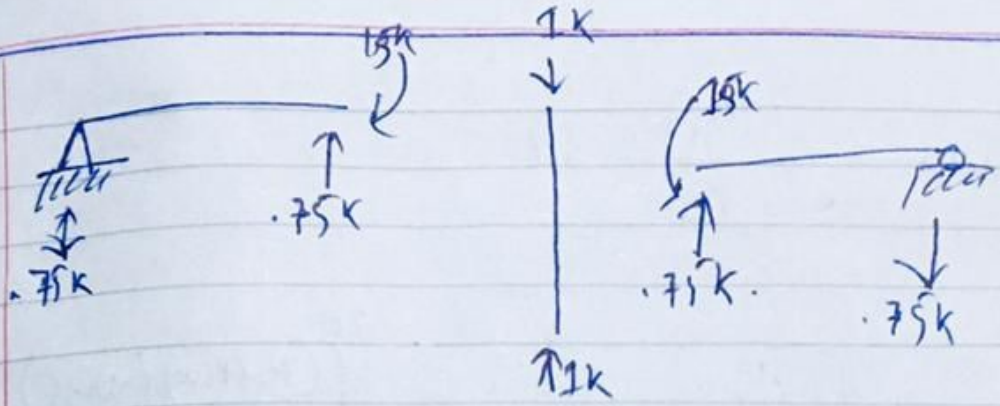
$$\begin{bmatrix} AR_1 \\ AR_2 \end{bmatrix} = \begin{bmatrix} ? \\ ? \end{bmatrix}$$

$$\begin{bmatrix} DR_{S_1} \\ DR_{S_2} \end{bmatrix} = \begin{bmatrix} 0 \\ 0 \end{bmatrix}$$

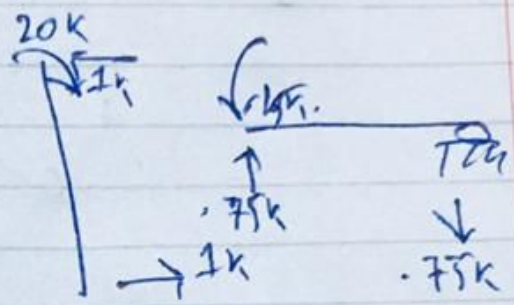
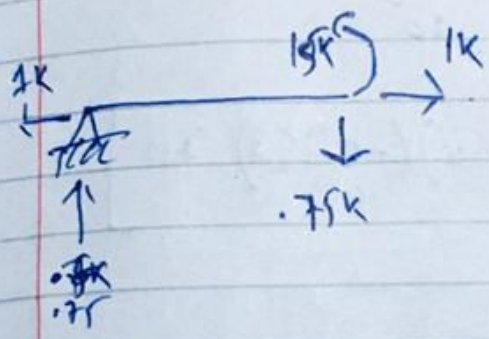
$[DRL]$ and $[F]$.



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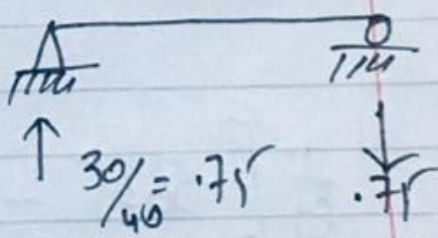
m_1 Value -



m_2 = Value -

Extra

Now



| Member | AB | Bc | BD |
|--------|------------------------------|----------|------|
| origin | A A | C | D |
| Limit | 0-10 10-20 | 0-20 | 0-20 |
| I | I I | I | I |
| M | $22.5x$ $22.5x - 20(x-10)$ | $7.5x$ | 0 |
| m_1 | $-0.75x$ $-0.75x$ | $-0.75x$ | 0 |
| m_2 | $0.75x$ 0.75 | $-0.75x$ | x |

$$DRL_1 = \int_0^L \frac{M_1 m_1}{EI} dx$$

$$= \frac{1}{EI} \left[\int_0^{10} (22.5x)(-0.75x) + \int_{10}^{20} \frac{(2.5x+200)(-0.75)}{\cancel{5x+700}} + \int_0^{20} (7.5x)(-0.75x) + 0 \right]$$

$$= \frac{1}{EI} \left[-\overset{5625}{\cancel{16875}} + (-26875) + (-15000) \right]$$

$$\frac{1}{EI} [-5625 - 26875 - 15000]$$

$$-47500/EI$$

New

$$DRL_2 = \int_0^L \frac{M_1 m_2}{EI} dx$$

$$DRL_2 = \frac{1}{EI} \left[\int_0^{10} (22.5x)(0.75x) + \int_{10}^{20} (2.5x+200)(0.75x) \right. \\ \left. + \int_0^{20} (7.5x)(0.75x) \right]$$

$$DRL_2 = \frac{1}{EI} [5625 + 26875 - 15000]$$

$$DRL_2 = \frac{1}{EI} [17500]$$

$$= \frac{17500}{EI}$$

Now

flexibility Matrix.

$$f_{11} = \int_0^L \frac{m_1^2}{EI}$$

$$f_{11} = \int_0^{10} (-0.75x)^2 = 187.5 / EI$$

$$f_{22} = \int \frac{(m_2)^2}{EI} =$$

16

$$= \int_0^L \frac{(m_2)^2}{EI}$$

$$= \int_0^{10} \frac{(0.75x)^2}{EI} = 550/EI.$$

$$F_{12} = F_{21} = 0.$$

Now

$$\begin{bmatrix} AR_1 \\ AR_2 \end{bmatrix} = \begin{bmatrix} 187.5 & 0 \\ 0 & 550.5 \end{bmatrix}^{-1} \begin{bmatrix} -47500 \\ 17500 \end{bmatrix}$$

$$\begin{bmatrix} AR_1 \\ AR_2 \end{bmatrix} = \begin{bmatrix} 21.15 \\ +1.35 \end{bmatrix}$$

— x — x — x — x — x —