

NAME

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SUBJECT

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Advance fluid  
Mechanics

SIR

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Eng. ABDUL WAHEED

Ques 1a:-

write down expression for velocity profile in laminar flow inside the pipe?

Answer

Velocity profile for laminar flow:-

As we have

$$hL = \frac{\tau \cdot 2L}{\epsilon r}$$

From viscosity  $\Rightarrow \tau = \mu \frac{du}{dy}$  (x)  
 where "u" is velocity at distance "y" from the boundary

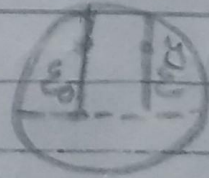
Thus

$$y = \epsilon_0 - \epsilon$$

$$dy = d\epsilon_0 - d\epsilon$$

Putting value in (x)

$$\tau = -\mu \frac{du}{d\epsilon}$$



$\therefore d\epsilon_0$   
constant value

$$\text{Now } hL = \frac{\tau \cdot 2L}{\epsilon r}$$

Integrating on b/s

Pg (2)

$$\int du = \int -\frac{hLy}{2\mu L} \cdot \xi \cdot dx$$

$$u = -\frac{hLy}{2\mu L} \frac{x^2}{2} + C$$

Now for  $\xi = 0$ ,  $u = u_{max}$

Putting value

$$u = -\frac{hLy}{2\mu L} \frac{x^2}{2} + C$$

$$u = u_{max}, \quad u_{max} = 0 + C$$

$$C = u_{max}$$

$$\text{Thus } u = u_{max} - \frac{hLy}{2\mu L} \frac{x^2}{2}$$

(velocity at any point)

$$\text{Assume } k = \frac{hLy}{4\mu L} \quad \therefore u = u_{max} - kx^2$$

$$\text{As for } \xi = x_0, \quad u = 0$$

$$0 = u_{max} - kx_0^2 \quad \left( \frac{hLy}{4\mu L} \cdot x_0^2 \right)$$

$$u_{max} = kx_0^2 = \frac{hLy}{4\mu L} \cdot x_0^2$$

(It is also known as critical velocity)

Now

$$V_{av} = \frac{V_{cr} + 0}{2} = 0.5V_{cr}$$

where  $v = \text{average velocity}$

Que 1 b:-

Define critical Reynold number. Write down its equation

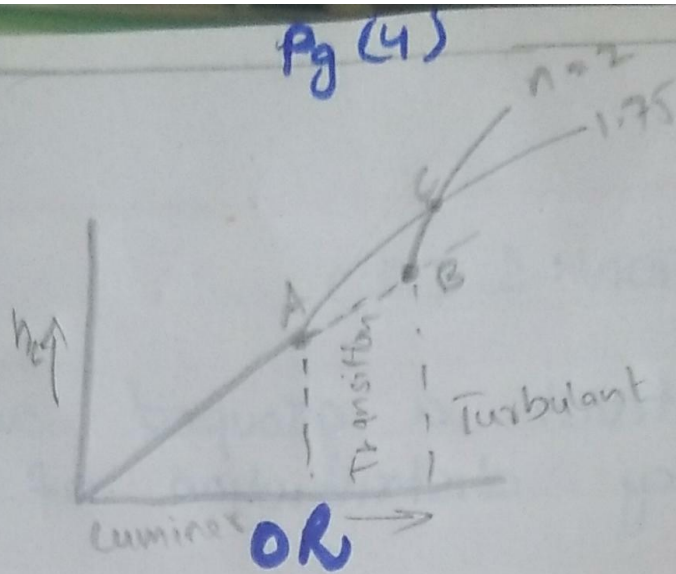
Answer:-

Critical Reynold number:-

If head loss in given length of uniform pipe is measured at different values of velocity. It will found as long as velocity is low enough to secure laminar flow, the head loss due to friction will be directly proportional to velocity but, increase in velocity change flow from laminar to turbulent cause change in head loss. Thus if value are plotted, line obtained with slope ranging about 1.75 to 2

Thus for laminar drop of energy varies as  $v$  and for turbulent friction varies as  $v^n$  where  $n$  is 1.75 to 2

⇒ The upper critical Reynolds number corresponding to point B is intermediate and depends upon care taken to prevent initial disturbance. Its value is 4000 but normally its impossible for flow to be in straight line after  $R$  is at 2000.



\* Critical reynolds number is a number which decides whether flow is laminar or turbulent.  
For example:

In external flow over a flat plate critical reynolds number is  $5 \times 10^5$  and reynolds number below it flow would be laminar and above the critical reynolds number flow will be turbulent.

In case of internal flow if reynolds number is below 2000 (critical reynolds number) flow would be laminar & if it is above 4000 flow would be treated as turbulent, region from reynolds num 2000-4000 is transition.

\* Equation

Ratio of inertial force to viscous force is called Reynold's number

$$R = \frac{F_I}{F_V} \Rightarrow F_I = ma = S l^3 \cdot \frac{L}{T^2}$$

$$\Rightarrow \int L^4 T^{-2} = \int \left(\frac{L}{T}\right) \left(\frac{L}{T}\right) \Rightarrow \int v^2 L^2$$

$$F_V = \mu \left(\frac{dv}{dy}\right) A = \mu \left(\frac{v}{L}\right) L^2 = \mu v L$$

$$R = \frac{L^2 v^2 \rho}{L \mu} = \frac{L v \rho}{\mu} = \frac{L v}{\nu} \Rightarrow \nu = \text{Kinematic velocity}$$

For circular pipe:

$$R = \frac{D v \rho}{\mu} = \frac{D v}{\nu}$$

Que 2:-

An oil of  $(S = 0.7)$  and kinematic viscosity of  $1.8 \times 10^{-5} \text{ m}^2/\text{s}$  is flowing in a pipe.

Ans :-

Given data:-

- Oil having  $S = 0.7$
- Kinematic viscosity  $= 1.8 \times 10^{-5} \text{ m}^2/\text{sec}$
- Dia of pipe  $= 150 \text{ mm} = 0.15 \text{ m}$
- Flow  $= 0.5 \text{ l}/\text{sec} = 0.0005 \text{ m}^3/\text{sec}$

Req:-

- \* Centerline velocity = ?
- \* Velocity at  $15 \text{ mm}$  from edge = ?
- \* Velocity at edge of pipe = ?
- \* Max shear stress at wall = ?

Solution:-

First we check flow is laminar or turbulent.

$$R = \frac{Dv}{\nu} \quad \text{--- (1)}$$

$$v = \frac{Q}{A} = \frac{Q}{\frac{\pi}{4}d^2} \Rightarrow \frac{0.0005}{\frac{\pi}{4}(0.15)^2}$$

$v = 0.028 \text{ m}/\text{sec}$

Pg (6)

$$R = \frac{(0.15)(0.028)}{1.8 \times 10^{-5}} \Rightarrow \boxed{233.37} < 2000 \text{ (laminar)}$$

$$R \approx 233.37 \quad v_{cr} = 2v \Rightarrow 2 \times 0.028 \Rightarrow \boxed{0.056 \text{ m/sec}}$$

As

$$U = U_{max} - Kr^2$$

at

$$r = r_0 = 0.075 \text{ m}, \quad u = 0$$

Thus

$$0 = U_{max} - Kr^2$$

$$U_{max} = Kr^2$$

$$K = \frac{U_{max}}{r^2} = \frac{0.056}{(0.075)^2}$$

$$\boxed{K = 9.96}$$

We get a equation

$$U = 0.056 - 9.96(r^2) \rightarrow (*)$$

at

velocity at 10mm from edge  
 $r = 0.065 \text{ m}$

$$v = 0.056 - 9.96(0.065)^2$$

$$v = 0.014 \text{ m/sec}$$

velocity at edge;

$$r = 0.075 \text{ m}$$

$$v = 0.056 - 9.96(0.075)^2$$

$$v = -0.00002 \text{ m/sec}$$

Seq

$$v = 0$$

Pg (7)

Similarly

$$f_2 = \frac{64}{R} = \frac{64}{233.33}$$

$$f = 0.27$$

Shear stress at wall

$$\tau = \frac{f}{4} \rho \frac{v^2}{2}$$

$$= \frac{0.27}{4} \times (0.7 \times 1000) + \frac{(0.056)^2}{2}$$

$$\boxed{\tau = 0.074 \text{ N/m}^2} \quad \text{ANS}$$