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Subject # Mechanics of solid (II)

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Q#01 (a)

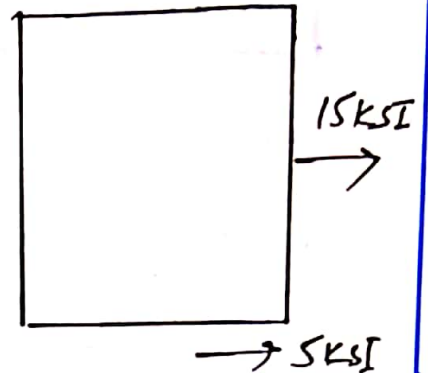
Ans

Given Data

$$\sigma_x = 15 \text{ ksi}$$

$$\sigma_y = 0$$

$$\tau_{xy} = -5 \text{ ksi}$$



Required Data:-

- (a) Principal stress
- (b) Max - plan shear stress
- (c) Average normal stress

Solution

(a) principal stress

$$\sigma_{1,2} = \frac{\sigma_x + \sigma_y}{2} \pm \sqrt{\left(\frac{\sigma_x - \sigma_y}{2}\right)^2 + \tau_{xy}^2}$$

$$\sigma_{1,2} = \frac{15 + 0}{2} \pm \sqrt{\left(\frac{15 - 0}{2}\right)^2 + (-5)^2}$$

$$\sigma_{1,2} = 7.5 \pm 9.01$$

$$\Rightarrow \boxed{\sigma_1 = 16.51 \text{ ksi}}$$

$$\sigma_2 = 7.5 - 9.01$$

$$\sigma_2 = -1.51 \text{ kSI}$$

Now we find orientation
:- we know that

$$2\theta_2 = \frac{\tau_{xy}}{(\sigma_x - \sigma_y)/2}$$

$$2\theta_2 = \frac{-5}{(15-0)/2}$$

$$\theta_2 = -0.33$$

→ now we check angles which
goes with which principle stress -
we know that

$$\begin{aligned}\sigma_{x_1} &= \frac{\sigma_x + \sigma_y}{2} + \frac{\sigma_x - \sigma_y}{2} \cos 2\theta + \tau_{xy} \sin 2\theta \\ &= \frac{15+0}{2} + \frac{15-0}{2} \cos 2(-0.33) + (-5) \sin 2(-0.33) \\ &= \frac{15}{2} + \frac{15}{2} (0.99) + (-5) (-0.72) \\ &= 14.925 + 0.6\end{aligned}$$

$$= 15.525$$

(3)

⑥ MAX - PLAN SHEAR STRESS:-

$$\tau_{max} = \sqrt{\left(\frac{\sigma_x - \sigma_y}{2}\right)^2 + \tau_{xy}^2}$$

$$\Rightarrow \tau_{max} = \sqrt{\left(\frac{15-0}{2}\right)^2 + (-5)^2}$$

$$\tau_{max} = 9.01 \text{ KSI}$$

⑦ Average Normal stresses

$$\sigma_x = 15 \text{ KSI}$$

$$\tau_{xy} = -5 \text{ KSI}$$

$$\sigma_y = 0$$

$$C = \frac{\sigma_x + \sigma_y}{2} = \frac{15+0}{2}$$

$$C = 7.5 \text{ ksi}$$

$$R = \sqrt{\left(\frac{\sigma_x - \sigma_y}{2}\right)^2 + \tau_{xy}^2}$$

$$R = \sqrt{\left(\frac{15-0}{2}\right)^2 + (-5)^2}$$

$$R = 9.01$$

$$\text{Scale} = 1 \text{ small box} = 0.5 \text{ KSI}$$

(4)

Now we find orientation
we know that

$$\begin{aligned}\tan 2\theta &= \frac{-(\sigma_x - \sigma_y)/2}{\tau_{xy}} \\ &= \frac{(15 - 0)/2}{-5}\end{aligned}$$

$$\tan 2\theta = +1.5$$

$$2\theta = \tan^{-1}(+1.5)$$

$$2\theta = 56$$

$$\boxed{\theta = 28}$$

we know that

$$\tau_{x'y'} = \frac{-\sigma_x - \sigma_y}{2} \sin 2\theta + \tau_{xy} \cos 2\theta$$

$$= \frac{-15 - 0}{2} \sin 2(28) + (-5) \cos 2(28)$$

$$= -7.5(0.82) - 2.8$$

$$\boxed{= -8.95}$$

Q#01 Part (b)

Average Normal Stresses

$$\sigma_x = 15 \text{ KSI}$$

$$\tau_{xy} = -5 \text{ KSI}$$

$$\sigma_y = 0$$

$$C = \frac{\sigma_x + \sigma_y}{2} = \frac{15 + 0}{2}$$

$$C = 7.5 \text{ KSI}$$

$$R = \sqrt{\left(\frac{\sigma_x - \sigma_y}{2}\right)^2 + \tau_{xy}^2}$$

$$R = \sqrt{\left(\frac{15 - 0}{2}\right)^2 + (-5)^2}$$

$$R = 9.01$$

Scale = 1 small box = 0.5 KSI

⑥

Now we find orientation

we know that

$$\tan 2\theta = \frac{-(\sigma_x - \sigma_y)/2}{\tau_{xy}}$$

$$\tan 2\theta = \frac{(15-0)/2}{-5}$$

$$\tan 2\theta = \textcircled{0} + 1.5$$

$$2\theta = \tan^{-1}(+1.5)$$

$$2\theta = 56$$

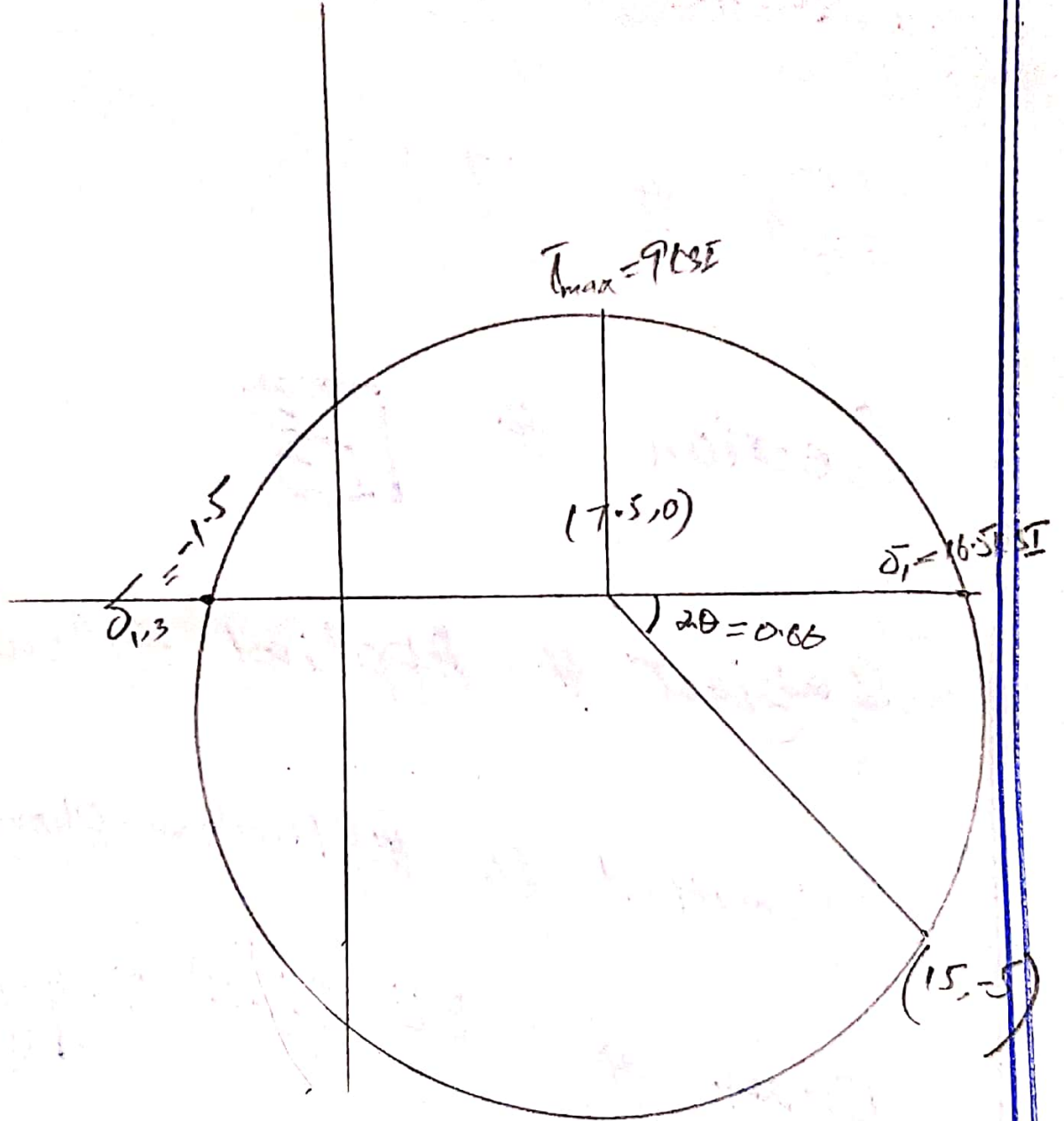
$$\boxed{\theta = 28}$$

we know that

$$\begin{aligned}\tau_{x'y'} &= \frac{-\sigma_x - \sigma_y}{2} \sin 2\theta + \tau_{xy} \cos 2\theta \\ &= \frac{-15-0}{2} \sin 2(28) + (-5) \cos 2(28) \\ &= -7.5 (0.82) - 2.8\end{aligned}$$

$$\boxed{= -8.95}$$

7



8

Q#02

Ans

$$\tau_{\max \text{ abs}} = \frac{\sigma_1}{2} = \frac{32}{2} = 16 \text{ MPa}$$

$$\sigma_{\text{avg}} = \frac{32 + 0}{2} = 16 \text{ MPa}$$

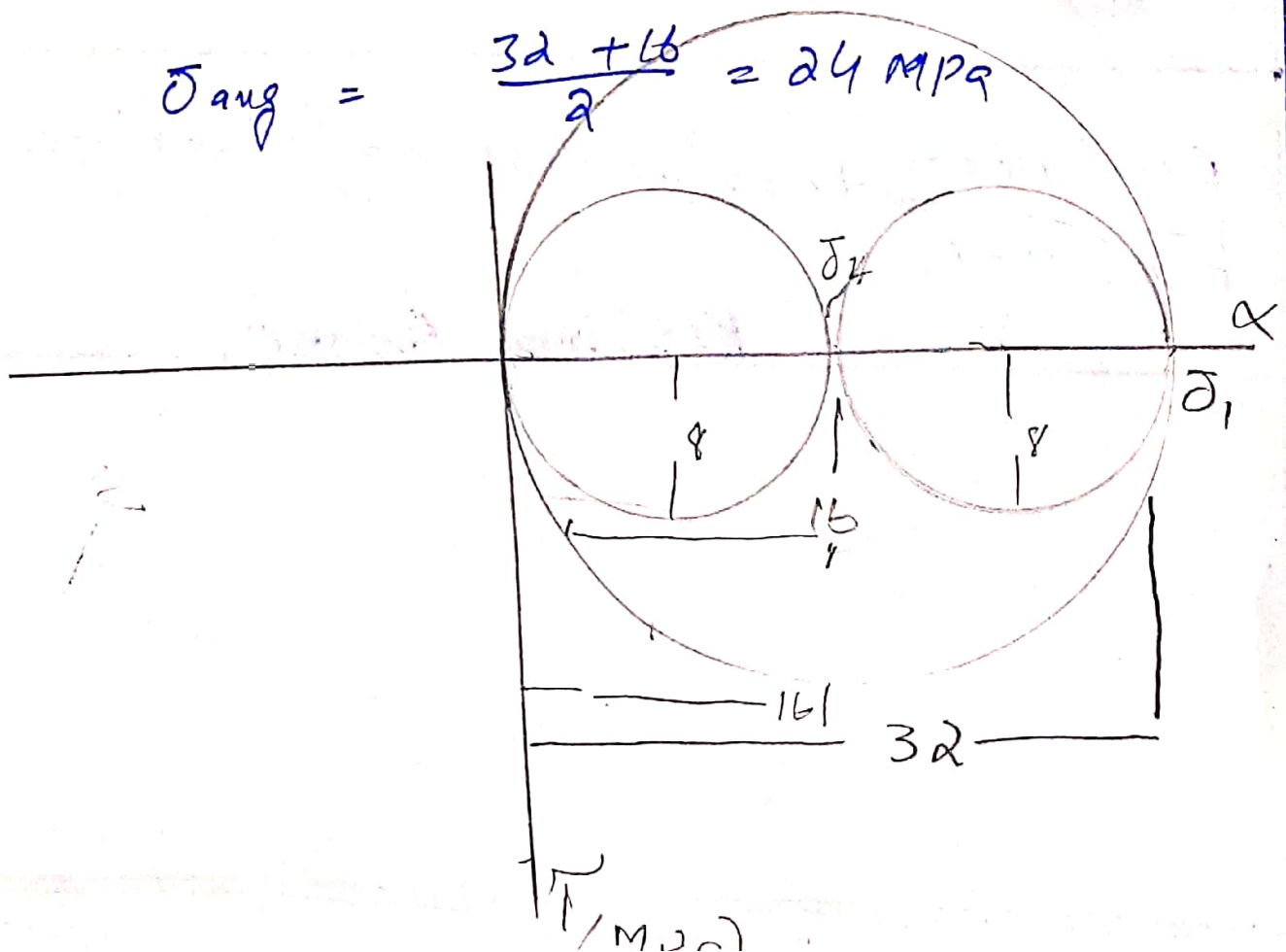
By Compression the maximum in plane shear stress can be determined from the Mohr's circle drawn b/w

$$\sigma_1 = 32 \text{ MPa and } \sigma_2 = 16 \text{ MPa}$$

This gives a value of

$$\tau_{\text{max plane}} = \frac{32 - 16}{2} = 8 \text{ MPa}$$

$$\sigma_{\text{avg}} = \frac{32 + 16}{2} = 24 \text{ MPa}$$



9

Q#3 ⇒ What are the main stresses

responsible for failure of ductile & brittle materials? Name of 2 failure theories for ductile materials and 2 for brittle materials -

Ans ⇒ Stresses are responsible for Failure of Ductile and Brittle Materials :-

⇒ Ductile materials are limited by their shear strength - Ductile materials usually fail because the shear stresses exceed the strength of ductile material -

⇒ Brittle materials are limited by their tensile strength - Brittle materials fail when the tensile stresses exceed the strength of material -

TWO Failure Theories for Ductile Materials

① :- Maximum Shear Stress Theory

⇒ According to this theory failure in ductile materials occurs when the maximum shear stress in the part exceeds the shear stress in a tensile test specimen (of the same material) at yield. The maximum shear stress can be determined by observing Mohr's circle for the element. The result indicates that

$$\tau_{max} = \sigma_y / 2$$

Maximum - Distortion - Energy Theory

⇒ It was stated in Sec 3.5 that an external loading will deform a material, causing it to store energy internally throughout its volume. The energy per unit volume of material is called the strain energy density.

and if the material is subjected to a uniaxial stress the strain energy density becomes

$$U = \frac{1}{2n} \sigma \epsilon$$

⇒ Two Failure Theory for Brittle Materials

① Maximum Normal Stress Theory

⇒ According to this theory "A brittle material will fail when maximum tensile stresses in the material reaches a value that is equal to ultimate normal stress the material can sustain when it is subjected to simple tension -

Mohr's Failure Criterion

:- In some brittle materials tension and compression properties are different - this occurs a criterion based on the use of Mohr's circle may be used to predict failure -

(12)

Failure Criterion

:- To apply it, one first performs three test on the material -

A uniaxial tensile test and uniaxial compressive test are used to determine the ultimate tensile stress (σ_{ult}) and compressive stresses, and (σ_{ult}^c) respectively.

Also a torsion test is performed to determine the material's ultimate shear stress (τ_{ult}).

Mohr's circle for each of these stress conditions is plotted as shown in figure - These three circles are contained in a

"failure envelop" indicated by extrapolated colored curve that is drawn tangent to all circles.