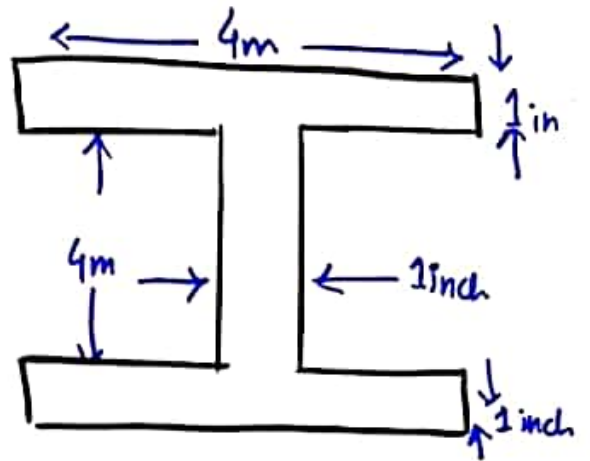
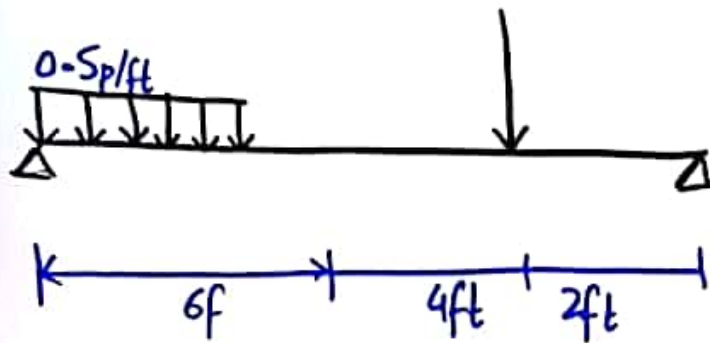


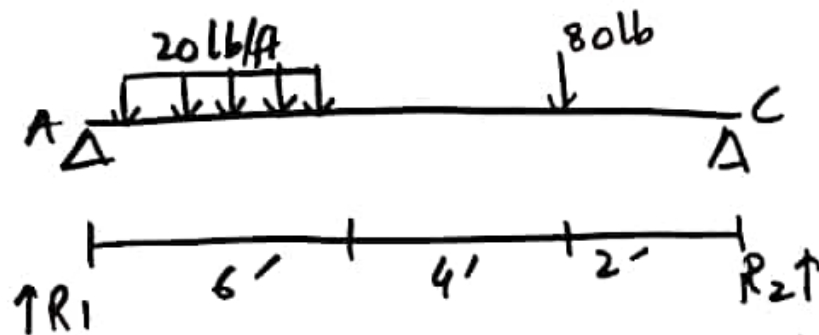
QUESTION

(1)

Given Beam:



Note: Put the value of $P=40$ so we have



First to find unknown reaction at the support apply equilibrium equation.

$$\sum F_x = 0 \quad \text{i.e. } R_2 = 0$$

$$\sum F_y = 0 \quad \oplus \downarrow \ominus$$

$$R_1 + R_2 = (20 \times 6) \text{ lb} + 80 \text{ lb}$$

$$R_1 + R_2 = 120 \text{ lb} + 80 \text{ lb}$$

$$\boxed{R_1 + R_2 = 200 \text{ lb}} \quad \text{--- (1)}$$

Next

$$\sum M_A = 0 \quad (\oplus \ominus)$$

$$R_2 \times 12 - 10 \times 80 - (20 \times 6) \times 3 = 0 \quad (2)$$

$$12R_2 = 800 + 360$$

$$12R_2 = 1160$$

$$R_2 = 96.66$$

$$\Rightarrow R_1 + R_2 = 200$$

$$\Rightarrow R_1 = 200 - 96.66$$

$$R_1 = 103.34$$

Now shear force at change point of Beam



Shear force at 6ft from support

$$\sum f_y = 0 \quad \uparrow \ominus \downarrow$$

$$103.34 - 20 \times 6 - V_{6ft} = 0$$

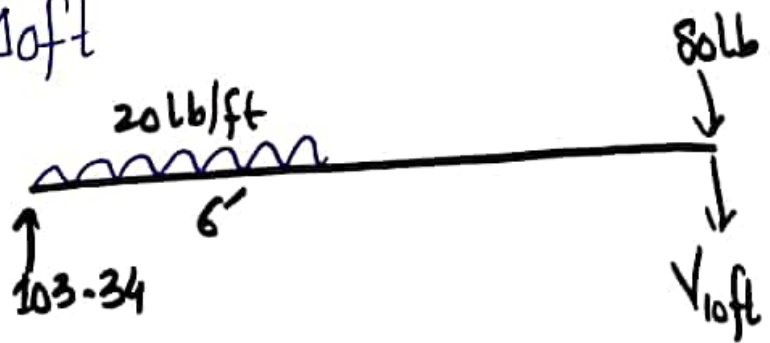
$$V_{6ft} = -16.66$$

Now shear force at 10ft

$$\sum f_y = 0 \quad \uparrow \ominus \downarrow$$

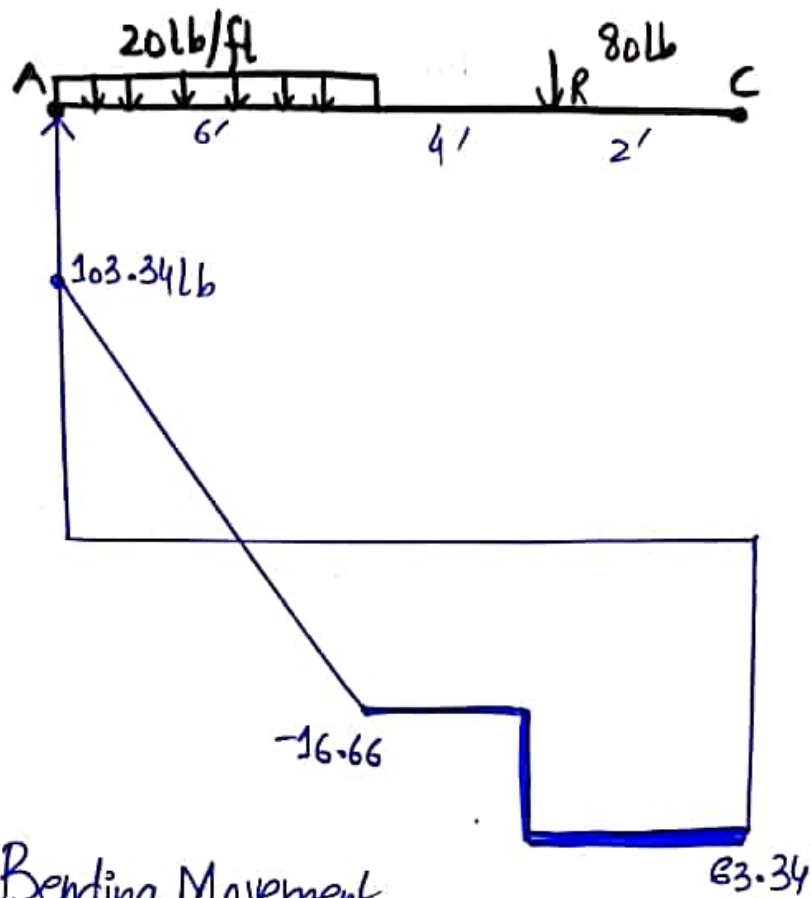
$$103.34 - 20 \times 6 - 80 - V_{10ft} = 0$$

$$V_{10ft} = 63.34 \text{ lb}$$



Now draw shear force diagram

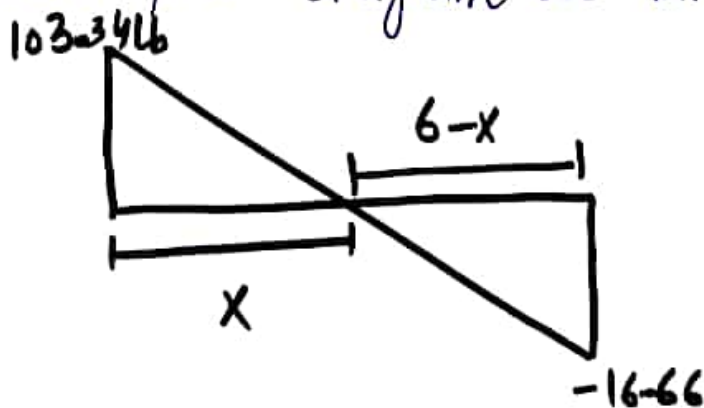
(3)



⇒ Point of Maximum Bending Movement

As we know that the point where shear force is ~~maximum~~ minimum the bending moment is maximum so from point of zero shear corresponding point will have maximum bending moment.

From shear force diagram we have



we know that

(4)

$$\frac{103.34 \text{ lb}}{x} = \frac{-16.66}{6-x}$$

$$\Rightarrow (6-x)(103.34) = (x)(-16.66)$$

$$\Rightarrow 620.04 - 103.34x = -16.66x$$

$$\Rightarrow 620.04 = -16.66x + 103.34x$$

$$\Rightarrow 620.04 = 86.68x$$

$$\Rightarrow x = 7.15 \text{ ft}$$

Now determine the value of moment at 7.15ft

$$M_{7.15 \text{ ft}} = 103.34 \times 7.15 + (20 \times 7.15) \times \left(\frac{7.15}{2} \right) \uparrow 103.34 \quad 7.15$$

$$M_{7.15} = 103.34 \times 7.15 + (20 \times 7.15) * \left(\frac{7.15}{2}\right) = 0$$

$$M_{7.15} - 738.8 + 143 * 3.57 = 0$$

$$M_{7.15} * \text{¢} - 738.8 + 510.51 = 0$$

$$M_{7.15} - 228.8 = 0$$

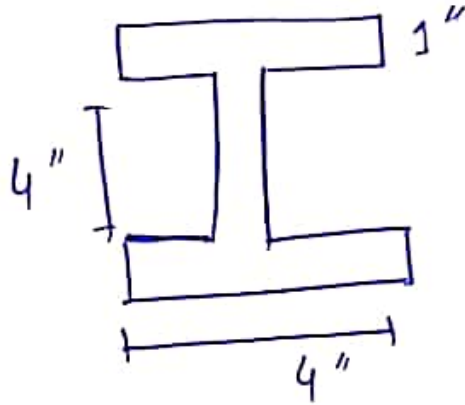
$$M_{7.15} = 228.8$$

For shear stress we have:

⑤

$$J = \frac{VQ}{Ib}$$

So first we determine moment of Inertia I of the given section of Beam



As the given figure is symmetrical along both the axis

$$\text{So } \bar{x} = 4/2 = 2 \text{ inch} \quad \bar{y} = 6/2 = 3 \text{ inch}$$

$$\text{i.e. } (\bar{x}, \bar{y}) = (2, 3)$$

[Centre of gravity]

extreme left and Bottom

$$\text{Area of point ①} = 4 \times 1 = 4 \text{ inch}^2$$

$$\text{Area of point ②} = 4 \times 1 = 4 \text{ inch}^2$$

$$\text{Area of point ③} = 4 \times 1 = 4 \text{ inch}^2$$

Moment of inertia about X-axis (centroid I) I_x

Determine distance b/w C.G of the whole section and corr. parts

Let G_1, G_2, G_3 be in the centre of gravity of point ① ② ③ and k_1, k_2 and k_3 be the distance b/w \bar{y} , and y_1, y_2, y_3 respectively.

So $k_1 = \bar{y} - y_1 \Rightarrow 3 - 0.5 \Rightarrow 2.5 \text{ inch}$

$k_2 = \bar{y} - y_2 \Rightarrow 3 - 3 \Rightarrow 0 \text{ inch}$

$k_3 = \bar{y} - y_3 \Rightarrow 3 - 0.5 \Rightarrow 2.5 \text{ inch}$

So $I_{xx} = \frac{b_1 h_1^3}{12} + a_1 k_1^2 + \frac{b_2 h_2^3}{12} + a_2 k_2^2 + \frac{b_3 h_3^3}{12} + a_3 k_3^2$

$I_{xx} = \frac{(4)(1)^3}{12} + 4(2.5)^2 + \frac{(1)(4)^3}{12} + a_2(0) + \frac{4(1)^3}{12} + 4(1.5)^2$

$I_{xx} = \frac{4}{12} + 25 + \frac{64}{12} + \frac{4}{12} + 25$

$I_{xx} = \frac{4}{12} + 25 + \frac{64}{12} + \frac{4}{12} + 25$

$I_{xx} = \frac{4 + (25)(12) + 64 + 4 + (25)(12)}{12}$

$I_{xx} = 56 \text{ inch}^4$

Now

$I_{yy} = \frac{b_1^3 h_1}{12} + \frac{b_2^3 h_2}{12} + \frac{b_3^3 h_3}{12}$

$$I_{yy} = \frac{(4)^3(1)}{12} + \frac{(1)^3(4)}{12} + \frac{(4)^3(1)}{12}$$

$$I = I_{xx} + I_{yy}$$

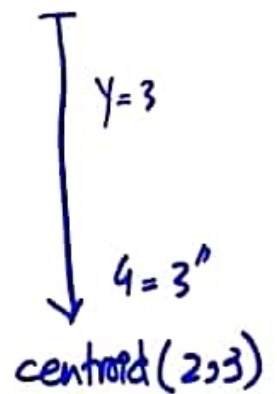
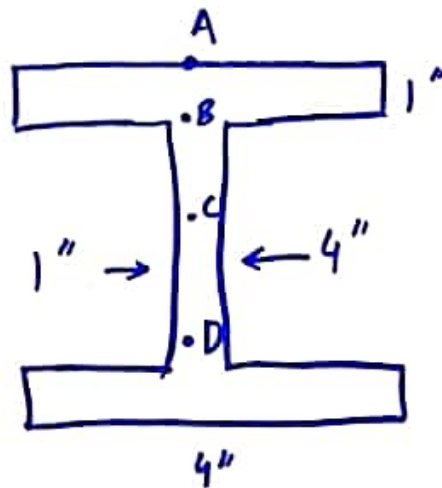
$$I = 67 \text{ inch}^4$$



$$I_{yy} = \frac{64+4+64}{12} = 11 \text{ inch}^4$$

Next find the shear stresses at various point we have

$$\tau = \frac{VQ}{Ib}$$



(i) Shear stress at point "A" i.e. At the top fiber

$$\tau' = \frac{VQ}{Ib}$$

$$V = \text{max} = 96.66 \text{ lb}$$

$$I = 67 \text{ inch}^4$$

So,

$$\tau = \frac{96.66(0)}{67(4)}$$

Here $A=0$ No area of the section exist above point A i.e. $Q = A\bar{y} \rightarrow 0$.

$$\tau = 0$$

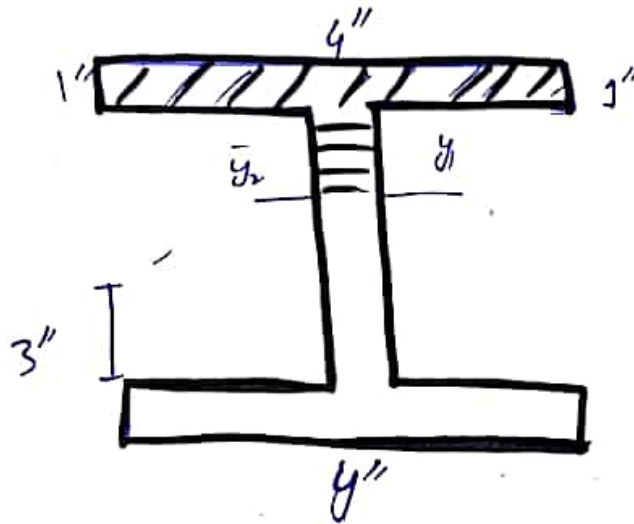
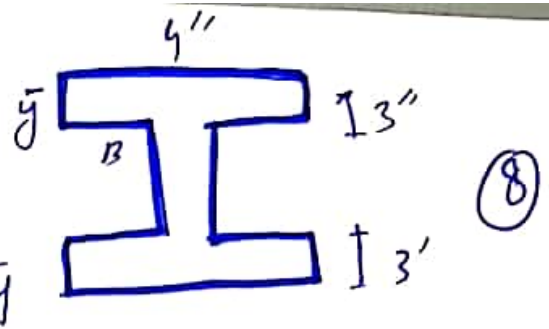
(ii) Shear stress at point "B"

$$\tau = \frac{VQ}{Ib} = \frac{96.66 \times (4 \times 1) \times (3 - 0.5)}{67 \times 4} = \frac{386.64 \times 2.5}{67 \times 4}$$

$$\Rightarrow \frac{966.6}{67 \times 4}$$

$$\Rightarrow 3.606 \text{ lb/in}^2$$

\Rightarrow Shear stress at point "C" i.e. at NA



$$\tau = \frac{VQ}{It}$$

$$\tau = \frac{96.66 \times [4 \times 1 \times (3 - 0.5) + (1 \times 2)(2 - 1)]}{67 \times 1}$$

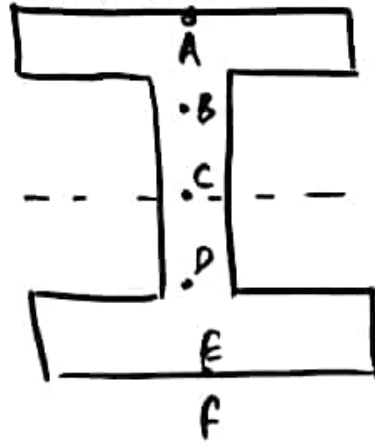
$$\tau = 15.86$$

iv. Shear stress at point D and E will be the same because of the symmetry.

Note :

The maximum shear stress value occurs at the Neutral axis and minimum at the top of the section.

Shear stress



M_{CB}

Max

N.A

M_{CX}

(9)

flexural stress determination

$$S = \frac{my}{I}$$

(i) Flexural stress at the top fibre point A

$$S = \frac{My}{I}$$

$$S = \frac{228.8 \times (3) / 0.5}{67}$$

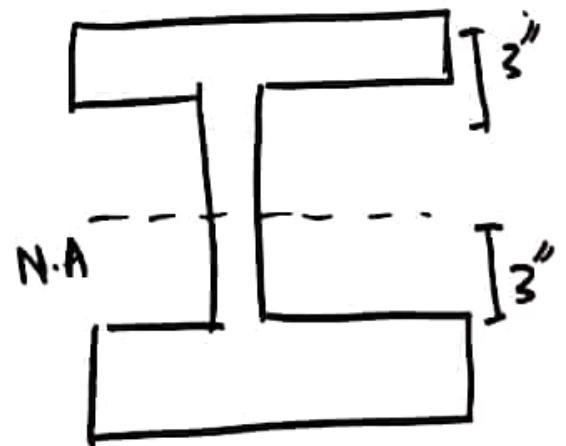
$$S = 10.24 \text{ lb/in}^2$$

(ii) Flexural stress at point "B"

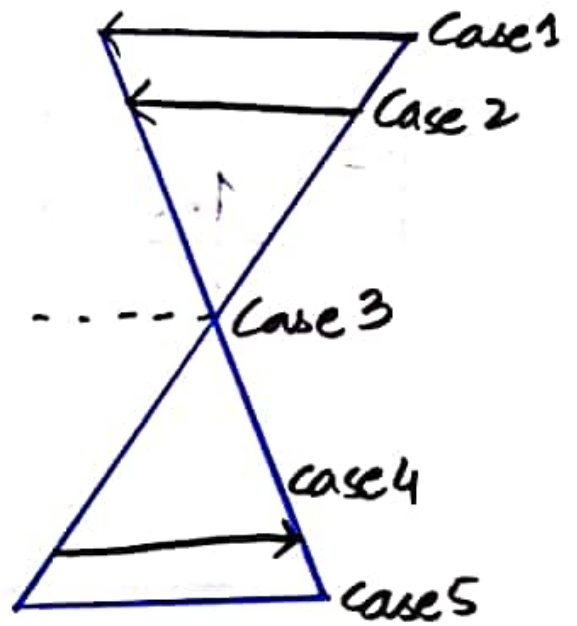
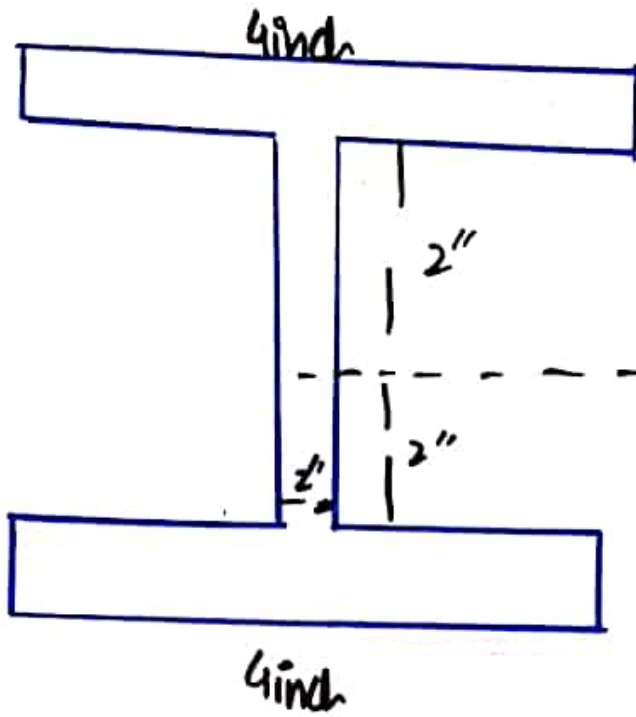
$$S = \frac{My}{I}$$

$$S = \frac{228.8 \times (3 - 0.5)}{67}$$

$$S = 8.53 \text{ lb/in}^2$$



Bending Diagram

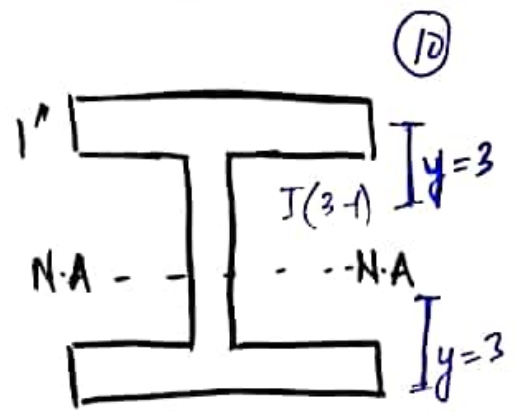


(iii) Flexural stress at point "C"

$$S = \frac{My}{I}$$

$$S = \frac{228.8 \times (3-1)}{67}$$

$$S = 6.82 \text{ lb/in}$$

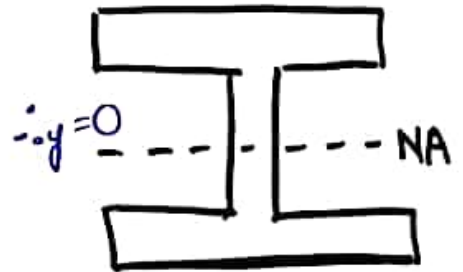


iv. Flexural stress at neutral axis (N.A)

$$S = \frac{My}{I}$$

$$S = \frac{634.02 \times 0}{67}$$

$$S = 0 \text{ lb/in}$$



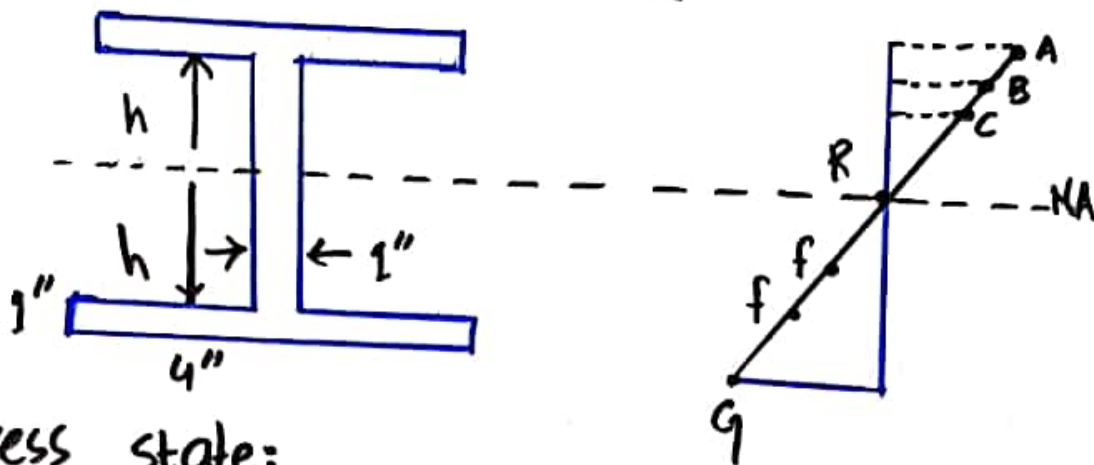
Flexural stress value at point E, F and G remain the same because of symmetry. The upper portion above the N.A show tension and below the N.A show compression.

Note:

The flexural stress value its maximum at extreme top ~~value~~ and bottom fibre at zero at N.A.

Flexural Stress diagram

11



Stress state:

Find stress state of a point element located 3ft from left support and 1 inch below from top fibre.

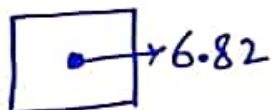
Flexural stress at point "C"

$$\sigma = 6.82$$

Shear stress at Point "C"

$$\tau = 15.86$$

Consider point "C" is a planar element



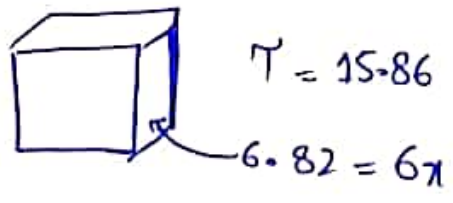
As the flexural stress is perpendicular to the cross section can be represented normal stress.

$\tau = 15.86$ psi is compressive because point "C" lies in compression zone of Beam cross section.



If point C lies below the centroid then stress would be tensile.

$$\tau = 15.86 \text{ psi}$$



Combine stress on 2d element.

Find its principle stress

We have also find,

$$\sigma_x = 6.82$$

$$\sigma_y = 0$$

$$\tau_{x,y} = 15.86$$

Principle stress equation

$$\sigma_{x,y} = \frac{\sigma_x + \sigma_y}{2} \pm \sqrt{\left(\frac{\sigma_x - \sigma_y}{2}\right)^2 + \tau_{x,y}^2}$$

$$= \frac{-6.82 + 0}{2} \pm \sqrt{\left(\frac{-6.82 - 0}{2}\right)^2 + (15.86)^2}$$

$$= -3.41 \pm \sqrt{-23.25 + 251.53}$$

$$= -3.41 \pm \sqrt{228.28}$$

$$= -3.41 \pm 15.10$$

$$\sigma_{x,y} \Rightarrow -3.41 + 15.10 = 11.69$$

$$\sigma_{n,y} \Rightarrow -3.41 - 15.10 = -18.51$$

Find $\phi P = ?$

$$\tan 2\phi P = \frac{\tau_{xy}}{\left(\frac{\sigma_x - \sigma_y}{2}\right)} \Rightarrow \frac{15.86}{\left(\frac{-6.82 - 0}{2}\right)}$$

$$\Rightarrow \frac{15.86}{-3.41}$$

$$\Rightarrow -4.65$$

$$\tan 2\phi_P = -4.65$$

(13)

$$2\phi_P = \tan^{-1}(-4.65)$$

$$\phi_P = -38.93$$

Put in general equation

$$\sigma_{\max} = \frac{-6.82 + 0}{2} + \frac{-6.82 + 0}{2}$$

$$\cos 2(-38.93) + 15.86 \quad \sin 2(-38.93)$$

$$\sigma_{p_{\max}} = -3.41 - 3.41$$

Max in plane shear stress in this case

$$\tan 2\phi_s = \frac{-(\sigma_x - \sigma_y)/2}{\sigma_{xy}}$$

$$= \frac{-(-6.82 - 0)/2}{15.86}$$

$$\tan 2\phi_s = +0.21$$

$$2\phi_s = \tan^{-1} 0.21$$

(14)

$$2\phi_s = +11.85$$

$$\phi_s = \frac{11.85}{2} \Rightarrow 5.92$$

Put in these general equation

$$T_{xy}' = - \left[\frac{6x - 6y}{2} \right] \sin 2\phi + 6xy \cos 2\phi$$

$$= - \left(\frac{-6.82 - 0}{2} \right) \sin 2(5.92) + 15.86 \cos 2(5.92)$$

$$= 3.41 \times 5.38 + 15.86 \times -2.46$$

$$= 18.34 - 39.01$$

$$= 20.67$$

To draw Mohr's circle centre coordinate (15)

$$(h, k) = \left(\frac{\sigma_x + \sigma_y}{2}, 0 \right)$$

$$= \left(\frac{-6.82 + 0}{2}, 0 \right)$$

$$\Rightarrow -3.41, 0$$

Radius of Mohr's circle

$$r = \sqrt{\left(\frac{\sigma_x - \sigma_y}{2} \right)^2 + \tau_{xy}^2} \Rightarrow \sqrt{\left(\frac{-6.82 - 0}{2} \right)^2 + (15.86)^2}$$

$$\Rightarrow 263.15$$

$$1 \text{ psi} = 1 \text{ cm}$$

$$G_x = -6.82, \quad G_y = 0$$

$$T_{xy} = 20.61 \text{ psi}$$

(16)

Scale :

$$1 \text{ psi} = 2 \text{ cm}$$

Then

$$\text{Radius, } r_1/2 = \frac{263.15}{2} = 131.57 \text{ cm}$$

$$\theta = \tan^{-1} \left(\frac{20.61}{-6.82 + 15} \right)$$

$$\theta = \tan^{-1}(2.51) \Rightarrow 68.27^\circ$$

$$\theta = 68.27^\circ$$

For $G_x' = -18.51 \text{ psi}$

$$G_y' = 0$$

Scale $1 \text{ psi} = 2 \text{ cm}$

$$\text{Radius} = \frac{263.15}{2} = 131.575$$

$$\phi_p = \frac{68.27}{2}$$

$$\phi_p = 34.135^\circ$$

