

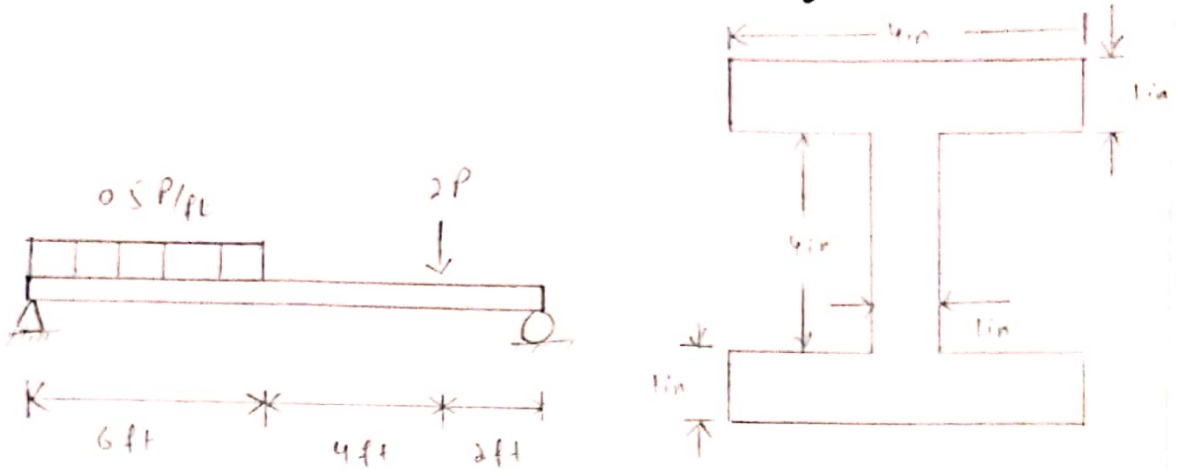
# MID TERM EXAM

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ID	7932
SECTION	B
DEPT	BE (C)
SUBJECT	MOS II
SEMESTER	4 <sup>th</sup>
SUBMITTED TO	Engr. Saqib Khan
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## QUESTION

Construct the Mohr's circle diagram

..... the stress transformation equation?



where as,

The value of  $P = 32$

SOL

PART A:

i) REACTIONS:-

$$\sum F_y = 0 \quad + \uparrow \text{ upward +ive}$$

$$\Rightarrow R_A + R_B - (0.5 \times 32 \times 6) - 2(32) = 0$$

$$\Rightarrow R_A + R_B - 96 - 64 = 0$$

$$\Rightarrow R_A + R_B = 96 + 64$$

$$R_A + R_B = 160 \quad \text{--- (i)}$$

$$\sum M_A = 0$$

$$(R_B \times 12) - (64 \times 10) - (96 \times 3) = 0$$

$$\Rightarrow 12R_B - 640 - 288 = 0$$

$$R_B = 928/12$$

$$R_B = 77.333 \text{ lb}$$

Putting the value of  $R_B$  in eq (i)

$$R_A + 77.333 = 160$$

$$R_A = 160 - 77.33$$

$$R_A = 82.67 \text{ lb}$$

ii) SHEAR FORCE :-

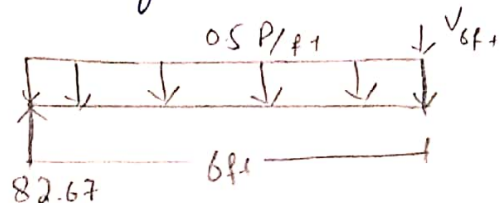
→ First of all we will find the shear force at 6ft.

$$\sum F_y = 0$$

$$\Rightarrow -V_{6ft} + 82.67 - 96 = 0$$

$$-V_{6ft} - 13.33 = 0$$

$$V_{6ft} = -13.33 \text{ lb}$$



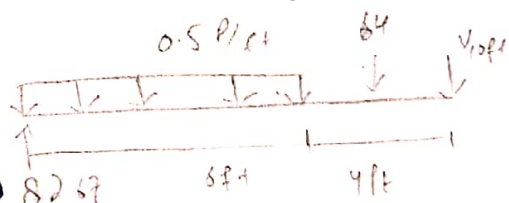
→ Secondly we will find at 10ft

$$\sum F_y = 0$$

$$\Rightarrow 82.67 - 96 - 64 - V_{10ft} = 0$$

$$\Rightarrow -V_{10ft} - 77.33$$

$$\Rightarrow V_{10ft} = -77.33 \text{ lb}$$



iii) BENDING MOMENT  $\approx$

$$\sum M_{6ft} = -(82.67 \times 6) + (16 \times 6) \left(\frac{6}{2}\right)$$

$$\sum M_{6ft} = -496.02 + 288$$

$$\boxed{\sum M_{6ft} = 208.02 \text{ lbft}}$$

$$\sum M_{3ft} = -(82.67 \times 3) + (16 \times 6 \times 3)$$

$$= -248.01 + 288$$

$$\boxed{\sum M_{3ft} = 39.99 \text{ lbft}}$$

Now we will find the moment of changing Point

$$\frac{82.62}{x} = \frac{13.33}{6-x}$$

$$82.62 \times (6-x) = 13.33x$$

$$495.72 - 82.62x = 13.33x$$

$$495.72 = 13.33x + 82.62x$$

$$495.72 = 95.95x$$

$$\frac{495.72}{95.95} = x$$

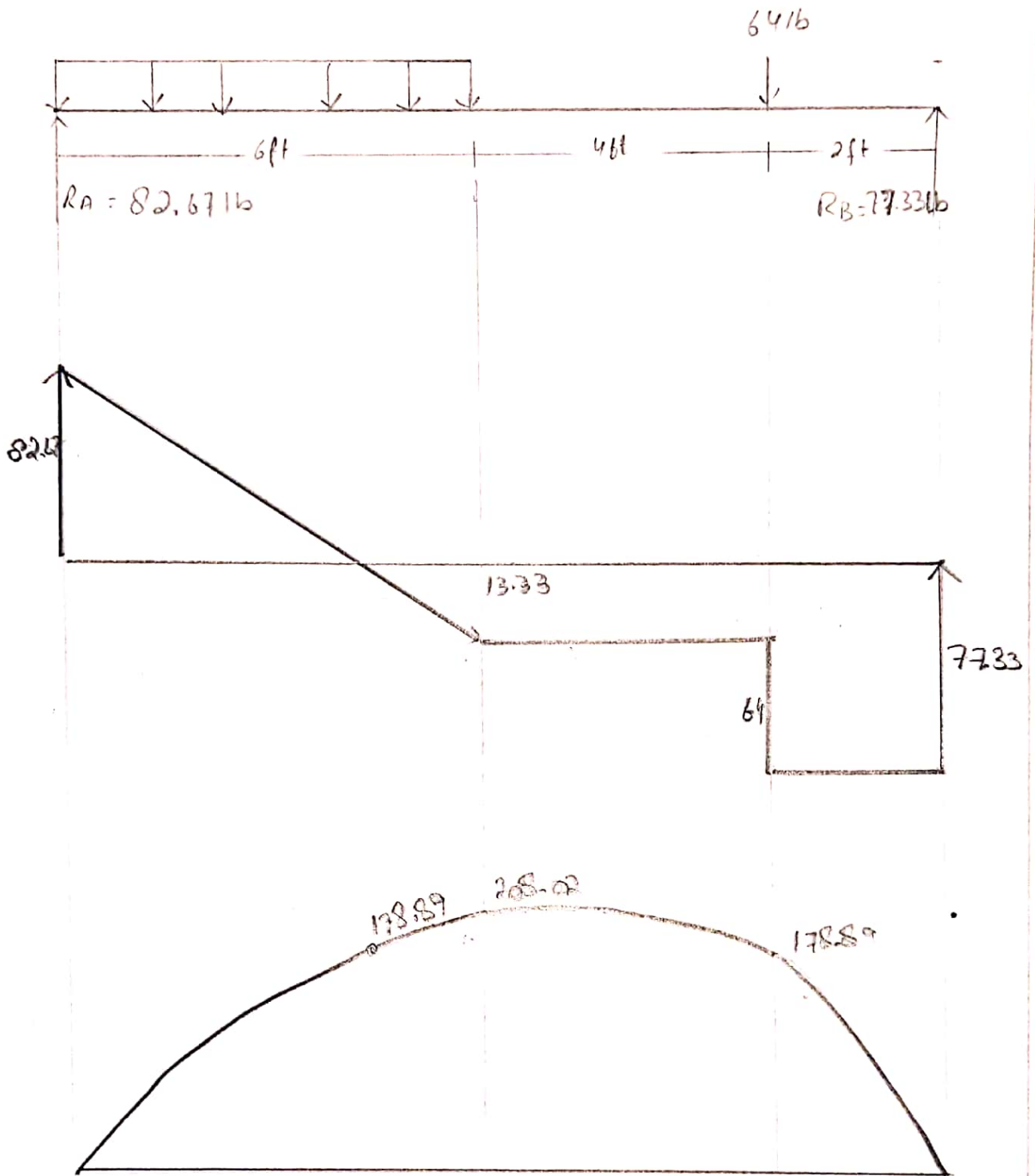
$$\boxed{x = 5.16 \text{ ft}}$$

$$\sum M_{5.16ft} = 0$$

$$\Rightarrow M_{5.16ft} - (82.67 \times 5.16) + (16 \times 6 \times \frac{5.16}{2}) = 0$$

$$\Rightarrow \boxed{M_{5.16ft} = 178.89 \text{ lbft}}$$

# SHEAR FORCE AND BENDING MOMENT DIAGRAM.





## PART B :

Now we have to find the moment of Inertia of the beam cross section

$$y_1 = 5.5$$

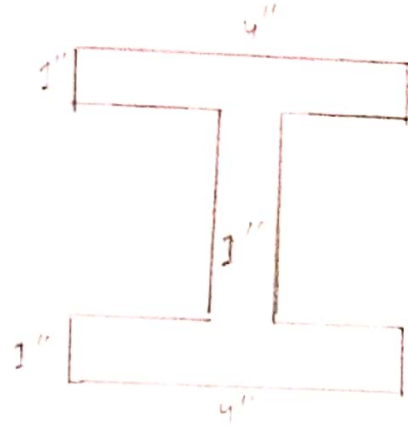
$$y_2 = 3$$

$$y_3 = 0.5$$

$$A_1 = 4 \text{ in}^2$$

$$A_2 = 4 \text{ in}^2$$

$$A_3 = 4 \text{ in}^2$$



$$\bar{y} = \frac{(A_1 \times y_1) + (A_2 \times y_2) + (A_3 \times y_3)}{A_1 + A_2 + A_3}$$

$$\bar{y} = \frac{(4 \times 5.5) + (4 \times 3) + (4 \times 0.5)}{4 + 4 + 4}$$

$$\bar{y} = 3''$$

Now 
$$\bar{I}_1 = \frac{bh^3}{12}$$
$$= \frac{4 \times 1^3}{12}$$

$$\bar{I}_1 = 0.33 \text{ inch}^4$$

$$\bar{I}_2 = \frac{bh^3}{12}$$

$$\bar{I}_2 = \frac{1 \times 4^3}{12}$$

$$\bar{I}_2 = 5.33 \text{ inch}^4$$

$$\bar{I}_3 = \frac{bh^3}{12}$$

$$= \frac{4 \times 1^3}{12}$$

$$\boxed{\bar{I}_3 = 0.33 \text{ inch}^4}$$

Now for d

$$d_1 = \bar{y} - y_1$$

$$\Rightarrow d_1 = 3 - 5.5$$

$$\boxed{d_1 = -2.5}$$

$$d_2 = \bar{y} - y_2$$

$$\Rightarrow d_2 = 3 - 3$$

$$\boxed{d_2 = 0}$$

$$d_3 = \bar{y} - y_3$$

$$\Rightarrow d_3 = 3 - 0.5$$

$$\boxed{d_3 = 2.5}$$

$$A d_1^2$$

$$A_1 d_1^2$$

$$\Rightarrow 4 \times (-2.5)^2$$

$$\Rightarrow 25 \text{ inch}^4$$

$$A_2 d_2^2$$

$$= 4 \times (0)^2$$

$$= 0$$

$$A_3 d_3^2$$

$$\Rightarrow 4 \times (2.5)^2$$

$$\Rightarrow 25 \text{ inch}^4$$

Now

$$\bar{I}_{1x} = \bar{I}_1 + A_1 d_1^2$$

$$= 0.33 + 25$$

$$\boxed{\bar{I}_{1x} = 25.33 \text{ inch}^4}$$

$$\bar{I}_{2x} = \bar{I}_2 + A_2 d_2^2$$

$$= 0 + 5.33$$

$$\boxed{\bar{I}_{2x} = 5.33 \text{ inch}^4}$$

$$\bar{I}_{3x} = \bar{I}_3 + A_3 d_3^2$$

$$= 0.33 + 25$$

$$\boxed{\bar{I}_{3x} = 25.33 \text{ inch}^4}$$

Now

$$\bar{I}_{xx} = \bar{I}_{1x} + \bar{I}_{2x} + \bar{I}_{3x}$$

$$= 25.33 + 5.33 + 25.33$$

$$\boxed{\bar{I}_{xx} = 55.99 \approx 56 \text{ inch}^4}$$

PART C :

SHEAR STRESS :-

CASE I :

We have to find the shear stress at top fiber.

As we know that

$$\tau = \frac{VQ}{It}$$

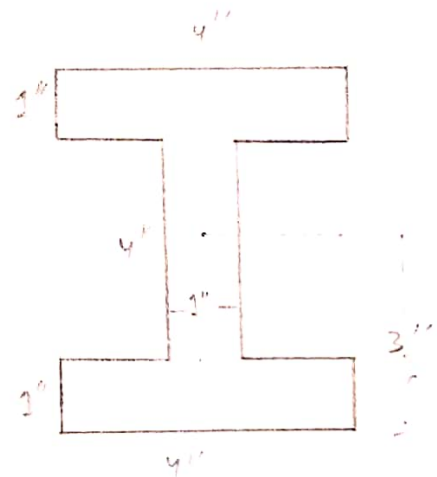
As top fiber  $A = 0$

$$Q = \bar{y}A \quad \therefore \bar{y} = 3''$$

$$Q = 3'' \times 0 = 0$$

$$\tau = \frac{77.33 \times 0}{56 \times 4}$$

$$\boxed{\tau = 0 \text{ Psi}}$$



$$V = 77.33, \bar{I} = 56, b = 4 \text{ in}$$



## CASE II

Now find the shear stress below the top fiber but here we have two cases.

$$\bar{y} = 2 + \frac{1}{2} = 2.5$$

$$A = 1.4 \times 4$$

$$Q_A = \bar{y} A$$

$$Q_A = 2.5 \times 4 = 10$$

$$b_A = 4 \text{ inch}$$

So

$$\tilde{\tau}_A = \frac{V Q_A}{I b_A}$$

$$= \frac{(77.33)(10)}{56 \times 4}$$

$$\tilde{\tau}_A = 3.452 \text{ Psi}$$

$$\tilde{\tau}_B = \frac{V Q_B}{I b_B}$$

$$\therefore Q = 2.5 \times 4 = 10$$

$$b_B = 1$$

$$= \frac{(7.33)(10)}{56 \times 1}$$

$$= 13.808$$

$$\tilde{\tau}_B = 13.808 \text{ Psi}$$

### CASE III $\approx$

Find stress at centroidal axis

$$Q = Q_1 + Q_2 \quad \text{--- (i)}$$

$$Q_1 = \bar{y}, A_1 = \frac{1}{2}(1 \times 2) = 2$$

$$Q_2 = 2.5 \times 4 = 10$$

So,  $Q = 2 + 10 \quad \therefore b = 1 \text{ inch}$

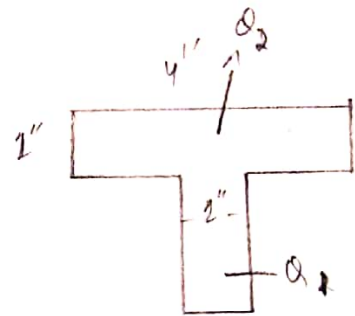
$$Q = 12$$

So,

$$\tilde{L}_{max} = \frac{VQ}{Ib}$$

$$= \frac{77.33 \times 12}{56 \times 1}$$

$$\tilde{L}_{max} = 16.570 \text{ Psi}$$



### CASE IV $\approx$

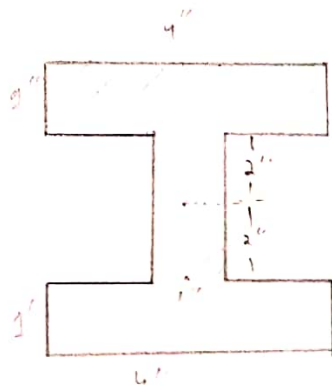
Find shear stress from  $I$  in above the fiber

$$\tilde{L}_A = \frac{77.33 \times (2.5 \times 4)}{56 \times 1}$$

$$\tilde{L}_A = 13.808 \text{ Psi}$$

$$\tilde{L}_B = \frac{77.33 \times (2.5 \times 4)}{56 \times 4}$$

$$\tilde{L}_B = 3.452 \text{ Psi}$$



CASE V : ~

Find shear stress at bottom fiber

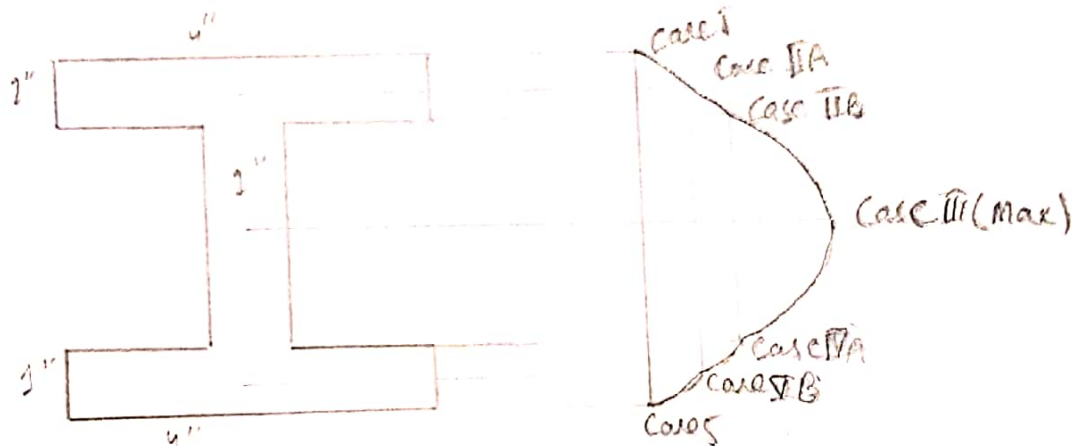
$$\text{Hence } \bar{y} = 0 \quad \& \quad Q = 4 \times 0 = 0$$

$$\tau = \frac{VQ}{Ib}$$

$$= \frac{77.33 \times 0}{56 \times 1}$$

$$\tau = 0 \text{ Psi}$$

SHEAR STRESS VARIATION DIAGRAM : ~



CASE VI : ~

We have to find the maximum shear stress at a distance of 6ft from left support of beam along its length

As we know that shear force at 6ft is

$$V = 13.33 \text{ lb}$$

$$\tau_{\text{max}} = \frac{VQ}{Ib}$$

$$Q = 12 \text{ (From case VI)}$$

$$\tau_{\text{max}} = \frac{13.33 \times 12}{56 \times 1}$$

$$\tau_{\text{max}} = 2.856 \text{ Psi}$$

## CASE VII :-

Now we have to find the shear stress at a distance of 6ft from left support and 1 inch below the top fiber.

$$\Rightarrow \tilde{\tau} = \frac{VQ}{Ib}$$

$$Q = 10 \quad (\text{from Case II A})$$

$$b = 4 \text{ inch}$$

$$\tilde{\tau} = \frac{13.33 \times 10}{56 \times 4}$$

$$\tilde{\tau} = 0.595 \text{ Psi}$$

## FLEXURE STRESS ANALYSIS:-

As we know that

$$b = \frac{My}{I}$$

The maximum bending moment is 256.08 lbf ft

## CASE I :-

Find stress at top fiber

$$b_{\text{top}} = \frac{(208.02)(3)}{56} \quad \therefore y = 3$$

$$b_{\text{top}} = 11.14 \text{ Psi}$$

## CASE II :-

Find stress at 1 inch below Top fiber

$$b_{1\text{in}} = \frac{(208.02)(2)}{56}$$

$$b_{1\text{in}} = 7.42 \text{ Psi}$$

CASE III : ~

Find the flexural stress at Centroidal axis

$$b_{\text{center}} = \frac{(208.02)(0)}{56}$$

$$b_{\text{center}} = 0 \text{ Psi}$$

CASE IV : ~

Find flexure stress at 1 inch above the bottom fiber.

$$b = \frac{My}{I}$$

$$b = \frac{(208.02)(2)}{56} \quad \because y = 2$$

$$b = 7.429 \text{ Psi}$$

CASE V : ~

Find flexural stress at the bottom fiber

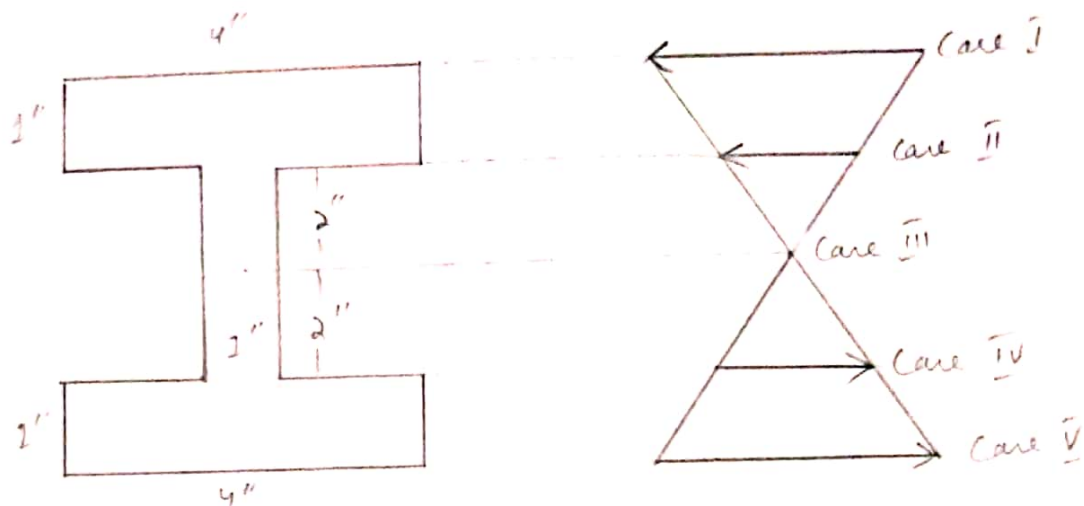
$$b_{\text{Bottom}} = \frac{My}{I}$$

$$\Rightarrow b_{\text{Bottom}} = \frac{208.02 \times 3}{56} \quad \because y = 3$$

$$b_{\text{Bottom}} = 11.14 \text{ Psi}$$



## FLEXURE STRESS VARIATION DIAGRAM :-



## SHEAR STATE OF A POINT ELEMENT :-

Now we have to find stress state of a point element located at a distance of 3 ft from left support and 1 inch below the top fiber.

As to find the condition of stressed element at point C in this given I section. It requires to find all the stresses at this point.

As in the given problem the stresses acting on point C is flexural and shear stresses. There is no torsional stresses forces acting on this beam due to load symmetry along the beam axis (longitudinal axis).

Flexure stress at point C

$$\sigma = 7.41 \text{ Psi} \quad (\text{from flexural stress Case II})$$

# Shear stress at Point "C"

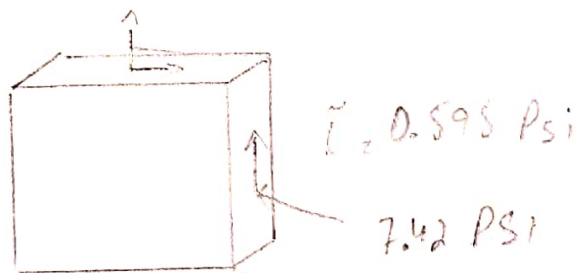
$$\bar{\tau} = 0.595 \text{ Psi} \text{ from Case III of shear stress}$$

Consider the Point "C" is a plane element

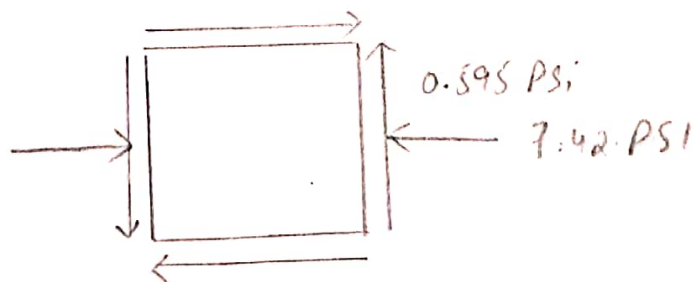


9.14 Psi is compressive because point C lies in compression zone of beam cross section.

If the point C lies below the centroidal then shear would be tensile.



Combining stresses on 2d element



PART D :-

PRINCIPLE STRESS :-

In this we have to find  $\theta_p$

$$\tan 2\theta_p = \frac{\tilde{\tau}}{(b_x - b_y)/2}$$

$$\tan 2\theta_p = \frac{0.595}{-7.42 - 0/2}$$

$$2\theta_p = \tan^{-1}(-0.16)$$

$$\theta_p = \frac{9.111}{2}$$

$$\boxed{\theta_p = -4.55}$$

Now for  $b_{x'}$

$$b_{x'} = \frac{b_x + b_y}{2} + \frac{b_x - b_y}{2} \cos 2\theta + \tilde{\tau}_{xy} \sin 2\theta$$

$$\Rightarrow b_{x'} = \frac{-7.42 + 0}{2} + \left(\frac{-7.42 - 0}{2}\right) \cos 2(-4.55) + 0.595 \sin 2(-4.55)$$

$$\boxed{b_{x'} = -7.467 \text{ Psi}} \text{ (Compression)}$$

$\Rightarrow$  Now for  $b_{y'}$

$$b_{y'} = \frac{b_x + b_y}{2} - \frac{b_x - b_y}{2} \cos 2\theta - \tilde{\tau}_{xy} \sin 2\theta$$

$$\Rightarrow b_{y'} = \frac{7.42 + 0}{2} - \left(\frac{-7.42 - 0}{2}\right) \cos 2(-4.55) + 0.595 \sin 2(4.55)$$

$$\boxed{b_{y'} = -0.140 \text{ Psi}} \text{ Compression}$$

Now

Shear Plane

$$\tan 2\theta_p = \frac{(-b_x - b_y)/2}{\tilde{\tau}_{xy}}$$
$$= \frac{(-7.42 - 0)/2}{0.596}$$

$$2\theta_p = \tan^{-1}(-2.80716)$$

$$\theta_p = \frac{-70.39}{2}$$

$$\boxed{\theta_p = 35.196}$$

$$\tilde{\tau}_{x'y'} = \left(\frac{-b_x - b_y}{2}\right) \sin 2\theta + \tilde{\tau}_{xy} \cos 2\theta$$

$$= \left(\frac{-7.42 - 0}{2}\right) \sin 2(35.196) + 0.595 \cos 2(35.196)$$

$$\Rightarrow \boxed{\tilde{\tau}_{x'y'} = 3.694}$$

STRESS TRANSFORMATIONS:-

Now we have to find the stress state condition of point C at a clockwise orientation where  $\theta = -25^\circ$  (assumed)

$$b_{x'} = \frac{b_x + b_y}{2} + \frac{b_x - b_y}{2} \cos 2\theta + \tilde{\tau}_{xy} \sin 2\theta$$

$$\Rightarrow b_{x'} = \left(\frac{7.42 + 0}{2}\right) + \left(\frac{-7.42 - 0}{2}\right) \cos 2(-25) + 0.598 \sin 2(-25)$$

$$\Rightarrow \boxed{b_{x'} = -6.550} \quad \text{Compression}$$



$$\sigma_{y'} = \frac{\sigma_x - \sigma_y}{2} - \frac{\sigma_x - \sigma_y}{2} \cos 2\theta - \tau_{xy} \sin 2\theta$$

$$\Rightarrow \sigma_{y'} = \frac{-7.42 + 0}{2} - \frac{-7.42 - 0}{2} \cos 2(25) - 0.595 \sin 2(-25)$$

$$\boxed{\sigma_{y'} = -0.869 \text{ Psi}} \quad \text{Compression}$$

Now

$$\tau_{x'y'} = \frac{\sigma_x - \sigma_y}{2} \sin 2\theta + \tau_{xy} \cos 2\theta$$

$$\tau_{x'y'} = \frac{7.42 - 0}{2} \sin 2(25) + 0.595 \cos 2(-25)$$

$$\boxed{\tau_{x'y'} = -2.45 \text{ Psi}}$$

MOHR'S CIRCLE :-

Coordinates :-

$$(h, k) = \left( -\frac{7.42}{2}, 0 \right)$$

$$(h, k) = (-3.71, 0)$$

RADIUS :-

$$r = \sqrt{\left( \frac{\sigma_x - \sigma_y}{2} \right)^2 + \tau_{xy}^2}$$

$$r = \sqrt{\left( \frac{-7.42 - 0}{2} \right)^2 + (0.595)^2}$$

$$\boxed{r = 3.64}$$



$$V = 7.42$$

Scale

$$1 \text{ psi} = 1 \text{ cm}$$

$$\bar{I}_{x'y'} = 0.595$$

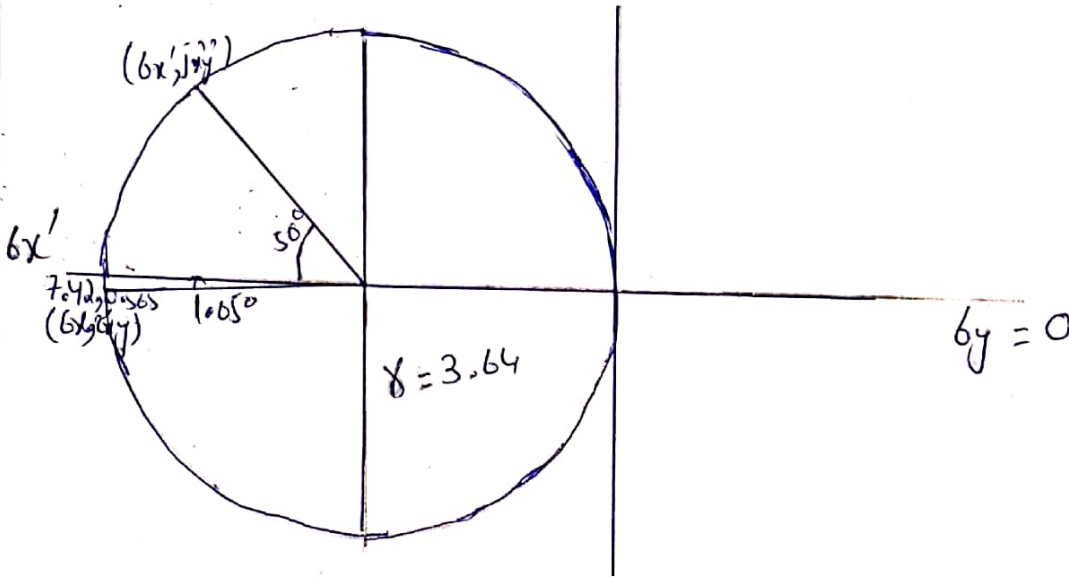
MOHR'S CIRCLE :-

$$b_x = -7.42, \quad b_y = 0, \quad \bar{I}_{x'y'} = 0.595$$

Now

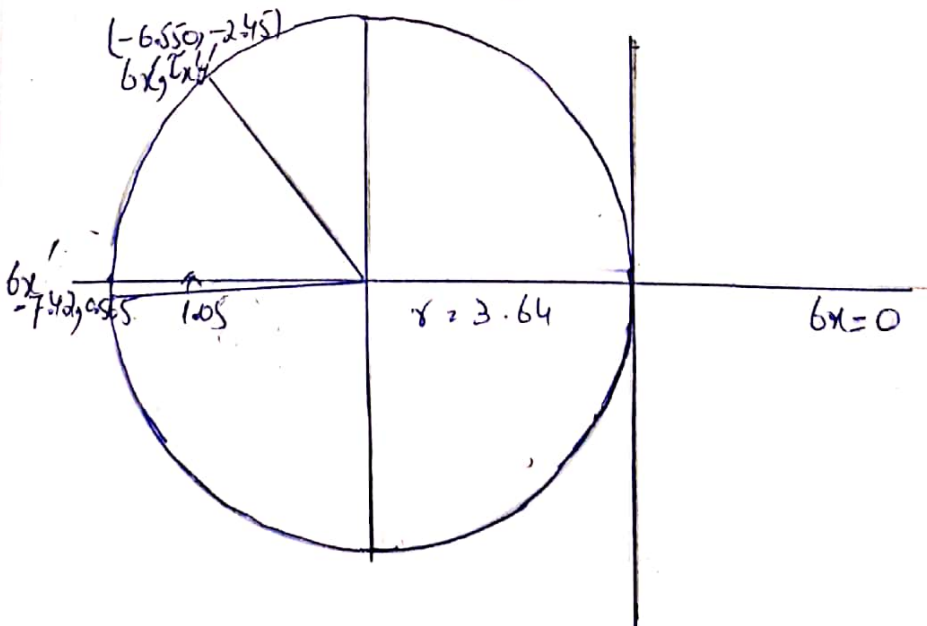
$$\theta = \tan^{-1} \left( \frac{0.596}{7.42 + 25} \right)$$

$$\Rightarrow \theta = 1.05^\circ$$



FOR NEW ORIENTATION ( $-25^\circ$  Clockwise)

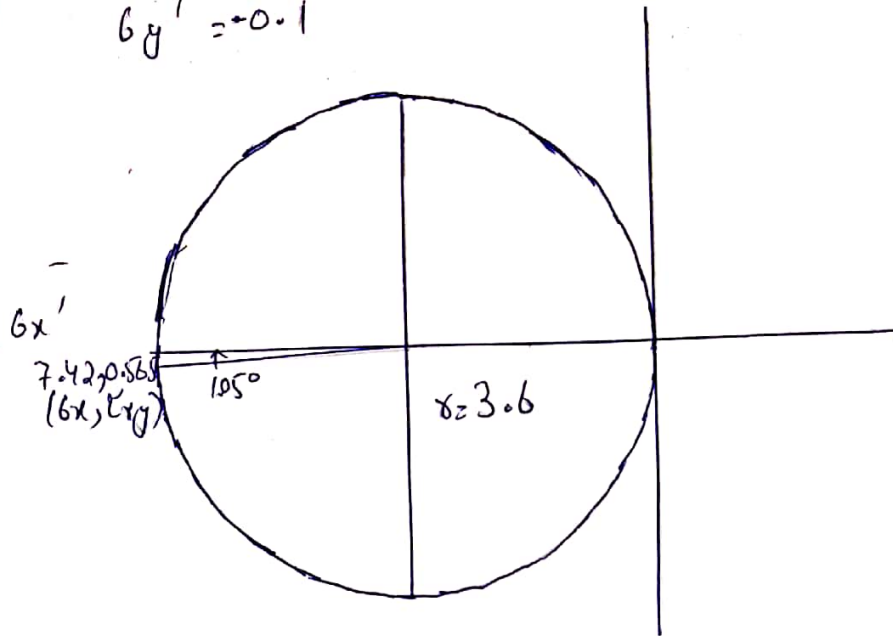
$$b_x' = -6.550 \text{ Psi}, \quad \bar{I}_{x'y'} = -2.45 \text{ Psi}$$



FOR PRINCIPLE STRESSES :-

$$\sigma_{x'} = -7.467$$

$$\sigma_{y'} = +0.1$$



Scale 1psi = 1cm

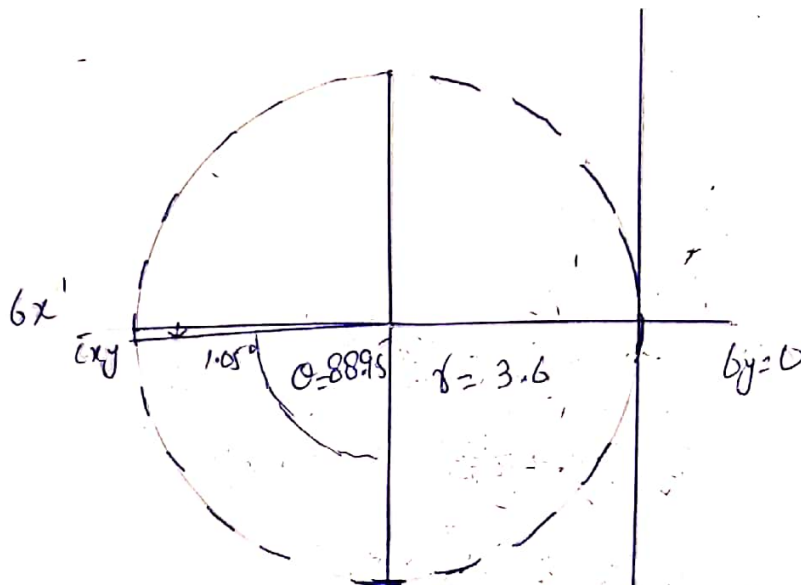
$$r = 3.6$$

$$\theta_p = \frac{1.05}{2}$$

$$\theta_p = 0.525$$

FOR Shear Stresses :-

$$\tau_{x'y'} = 3.69$$



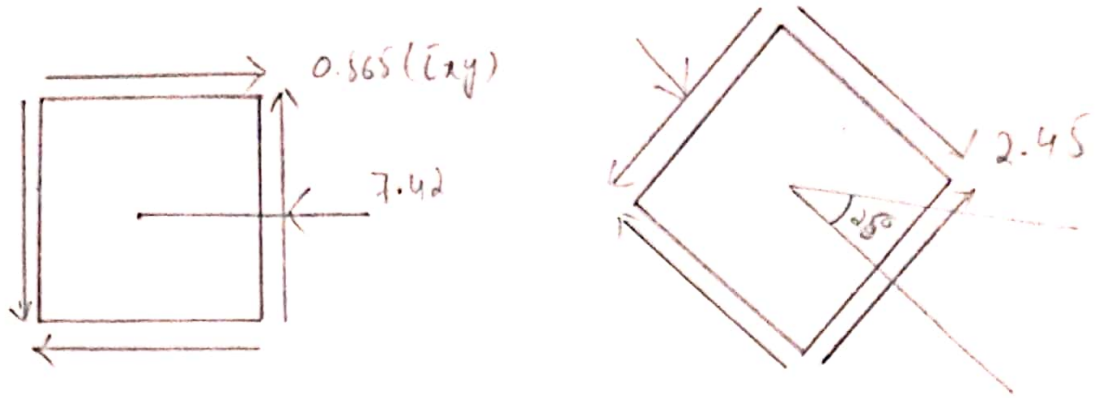
$$\theta_1 = \frac{88.95}{2}$$

$$\theta_3 = 44.475^\circ$$

$$\tau_{x'y'} = 3.69 \text{ psi}$$

$\tau_{\max (+)}$  in plane

# COMPARISON OF MOHR'S CIRCLE WITH SHEAR TRANSFORMATION $\theta = 45^\circ$



## PRINCIPLE AND PLAN SHEAR STRESS $\theta = 45^\circ$

Principle

