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Radar and Satellite Communication

Q1

a) Find and explain Track and Search Radar Range equation.

Ans

Radar Range Equation

• For a directional radar, the power density per unit area is given by

$$P_{\text{directional}} = \frac{P_t G_t}{4\pi R^2}$$

where $P_{\text{directional}}$ = radiation power density of the isotropic radiation $[W/m^2]$

G_t = directional gain of the antenna measured in the target direction

- A target at distance R intercepts the transmitted energy, Part of the energy will be reflected by the target

- The re-radiated power due to the reflection

$$P_{T \text{ target}} = P_{\text{direction}}$$

$$\Phi = \frac{P_T \Gamma_T \sigma}{4\pi R^2}$$

where σ = target radar cross section, the target EM size viewed by the radar.

- The reflected power from the target received by the radar,

$$P_{\text{reflected}} = \frac{P_T \Gamma_T \sigma}{4\pi R^2}$$

$$= \frac{P_T \Gamma_T \sigma}{(4\pi R^2)^2}$$

$$P_r = P_{\text{reflected}} A_e = \frac{P_T G_T A_e}{(4\pi R^2)^2} \sigma$$

$$A_e = \frac{G_T \lambda^2}{4\pi}$$

$$P_r = \frac{P_T G_T G_T \lambda^2}{(4\pi)^3 R^4} \sigma$$

$$P_r = \frac{P_T G^2 \lambda^2}{(4\pi)^3 R^4} \sigma$$

• The signal to noise ratio at output,

$$SNR_o = \frac{S_o}{N_o} = \frac{AP_r/L}{AF_n KTB} = \frac{P_r}{F_n KTB L}$$

$$P_r = \frac{P_T G^2 \lambda^2}{(4\pi)^3 R^4} \sigma$$

the complete radar equation

$$SNR_o = \frac{P_T G^2 \lambda^2 \sigma}{(4\pi)^3 R^4 F_n KTB L}$$

Maximum Detection Range

- The maximum detection range of a target with a specified radar cross section, σ , is

$$R = \left[\frac{P_T G^2 \lambda^2 \sigma}{(4\pi)^3 F_n k T B L (SNR_c)} \right]$$

- The maximum range is obtained when the signal-to-noise ratio of a target is minimum,

$$R_{max} = \left[\frac{P_T G^2 \lambda^2 \sigma}{(4\pi)^3 F_n k T B L (SNR_{o, min})} \right]$$

Development of Search Radar Equation:-

Track Radar Equation

$$\frac{S}{N} = \frac{P_T G^2 A^2 \sigma}{(4\pi)^3 R^4 k T_s B_n L}$$

Search Radar Equation

$$\frac{S}{N} = \frac{P_{av} A_e t_s \sigma}{4\pi \Omega R^4 T_s L}$$

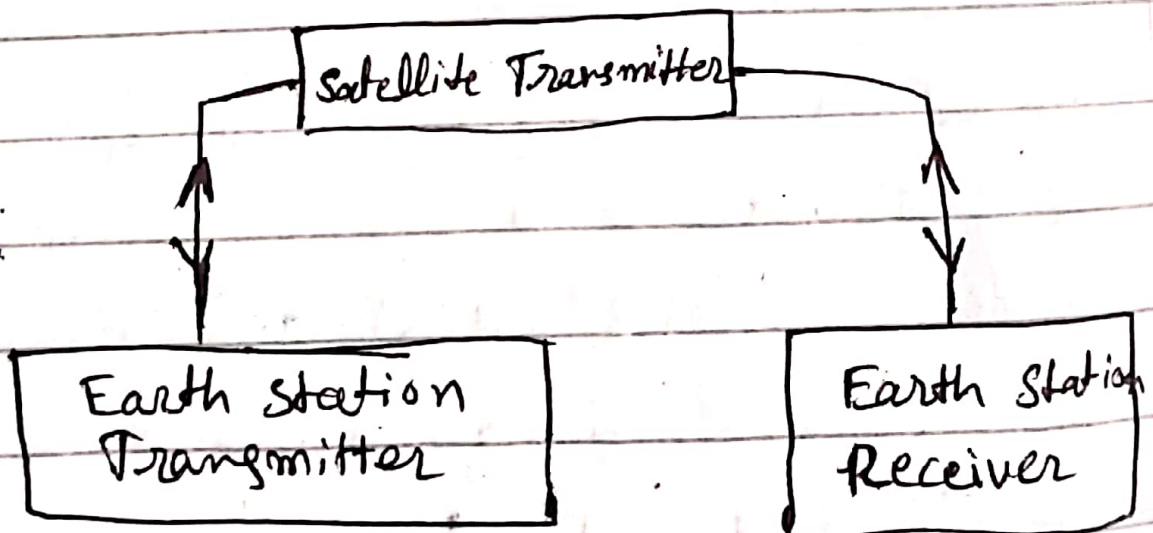
Q.1

Part B Explain working principle of satellite communications system (GPS), and difference between triangulation and Trilateration?

Ans Principle of communication

→ In satellite communication, signal transferring between the sender and receiver is done with the help of satellite.

→ In this ~~process~~ process, the signal which is basically a beam of modulated microwaves is sent towards the satellite. Then the satellite amplifies the signal and sent it back to the receiver's antenna present on the earth's surface.



GPS Basics:

• Triangulation

- Multiple bearings on an unknown target
- Readings taken from known locations
- Tends to be planar

• Trilateration

- Multiple distance determinations to an unknown target
- Readings taken from known locations
- 3D positioning

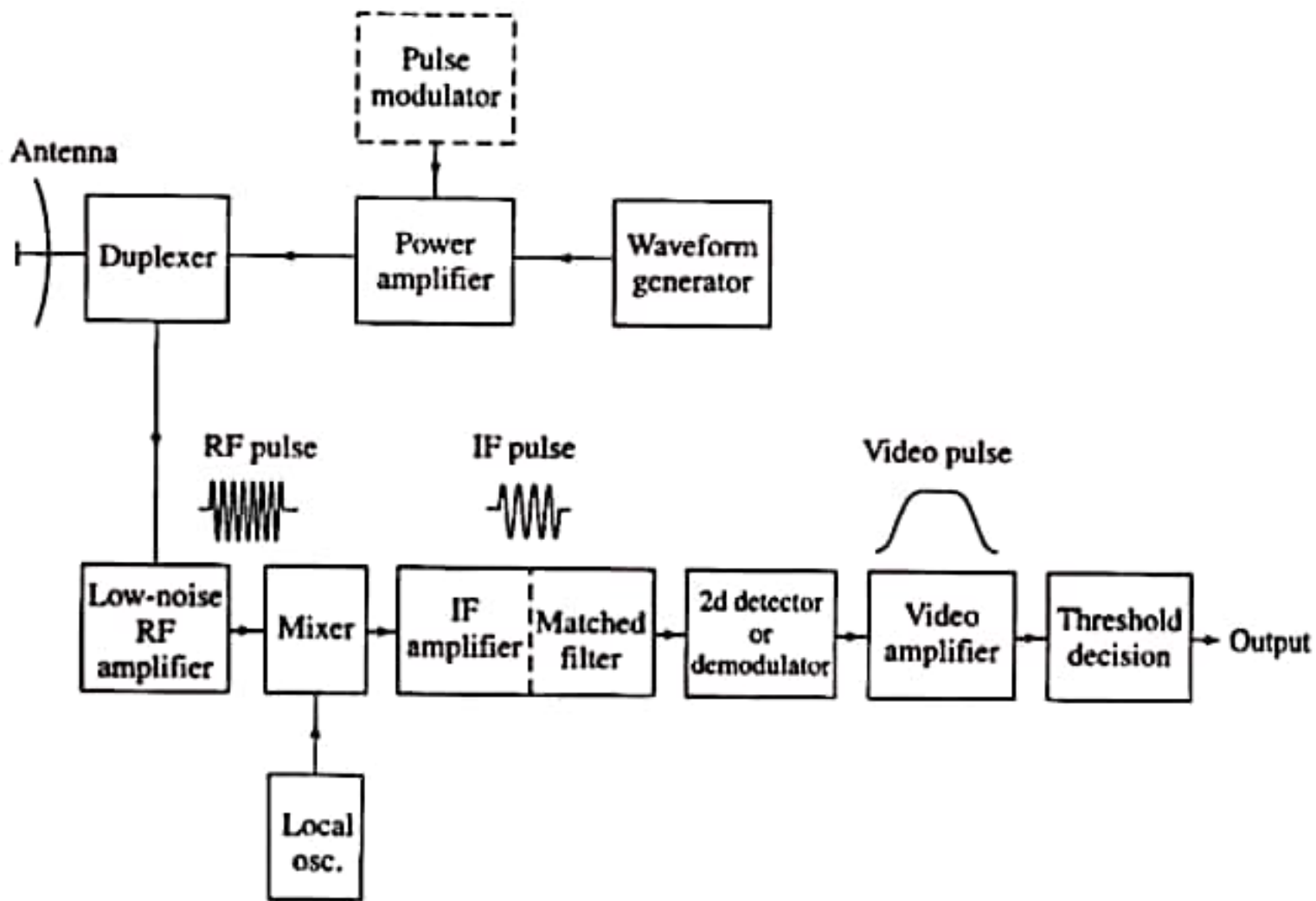
Q.2 Draw and Explain

a) Radar Block Diagram.

Ans Explanation:

The block diagram of a simple radar system is shown in figure.

This block diagram indicates that the radar



system consists of both the transmitting and the receiving system. This connects the antenna to the transmitter during transmission and to the receiver during reception.

b) Find the escape velocity of satellite?

Ans Escape velocity

- The minimum velocity needed for a celestial body to escape the gravitational pull of another, larger body and not fall back to that body's surface.

In our case, satellites escape earth's gravity

Expression for orbital velocity

$$\frac{1}{2} m v_R^2 = G M_E m / R \text{ or } v_e = \sqrt{(2 G M_E / R)} = \sqrt{R g}$$

Relation between orbital velocity and Escape velocity

$$\text{orbital velocity } v_o = \sqrt{R g}$$

$$\text{Escape velocity } v_e = \sqrt{2 R g}$$

$$v_e = \sqrt{2} v_o$$

Q.3 Find out the orbital speed
a) and Time period of satellite?

Ans Time Period of a satellite.

Distance covered by the satellite in 1 revolution
= circumference of the circle

Time taken to cover this distance is the time period

critical speed $v_c = \frac{\text{circumference}}{\text{period/time}}$

$$v_c = \frac{2\pi r}{T}$$

But $v_c = \sqrt{\frac{GM}{r}}$

$$\sqrt{\frac{GM}{r}} = \frac{2\pi r}{T}$$

$$\frac{GM}{r} = \frac{4\pi^2 r^2}{T^2}$$

$$T^2 = \frac{4\pi^2 r^3}{GM}$$

As $\frac{4\pi^2}{GM}$ is a constant,

so we get

$$T^2 \propto r^3$$

$$T = 2\pi \sqrt{\frac{r^3}{GM}}$$

orbital speed

- orbital speed of a satellite is the minimum speed required to put the satellite into a given orbit around earth.

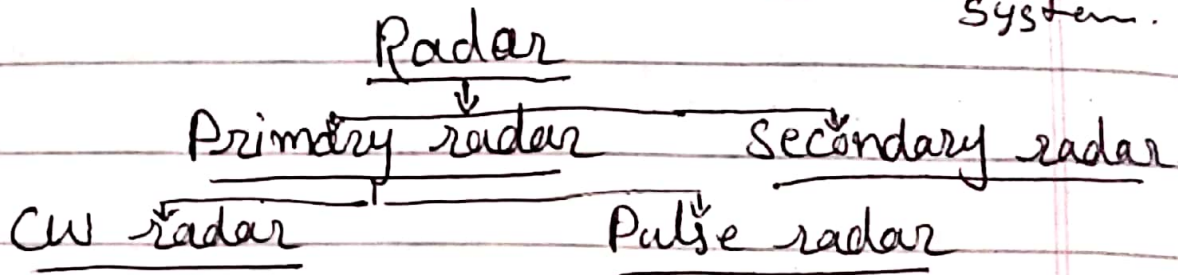
- Expression - $v_0 = \sqrt{gR}$ (Re-g)

Where, $g = 9.8 \text{ m/s}^2$ and $R = \text{radius of earth}$. The value for orbital velocity was found to 7.9 km/s .

$$u = \sqrt{\frac{GM}{R}} = \sqrt{\frac{gR^2}{R}}$$

- It is independent of mass of the satellite.
- Decrease with an increase in the radius in the radius in the height of satellite.

Q.3 Explain different types of radar system.
Part B



Types of Radar

There are basically two different types of radar, Primary and secondary

Primary Radar:

relies solely on the energy that it has generated and radiated, being reflected back from the target.

Secondary Radar:

has some co-operation from the target the target generates its own comm radiation.

• Frequency Modulated CW Radar

- Use for radar altimeters and missile guidance.

• Pulse Doppler

- carrier wave frequency within pulse is compared with a reference signal to detect moving targets.