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PAPER :- Steel structure

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Q No 1:-

Pg (1)

⇒ Given data.

$$DL = 60 \text{ kip}$$

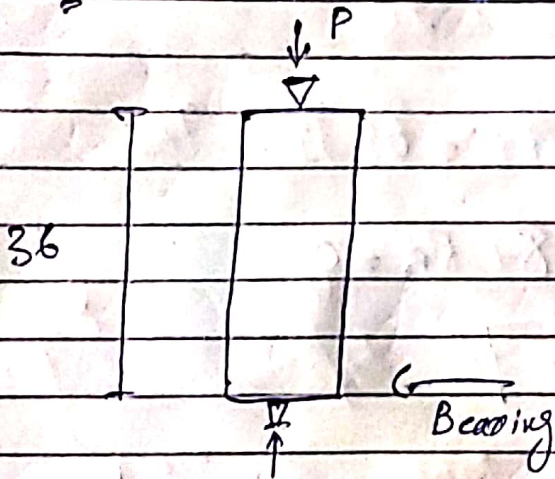
$$LL = 110 \text{ k}$$

$$L = 36 \text{ ft.}$$

$$K_x = 1$$

$$K_y = 1$$

Sol Use LRFD.



Assume hinged ~~at~~ Top and Bottom
of and y-axis bending.

Calculate the load (factor load).

$$P_u = 1.2 DL + 1.6 LL$$

$$\Rightarrow 1.2(60) + 1.6(110)$$

$$= P_u = 248 \text{ kips.}$$

To find out the support reaction.

$$\left(\begin{array}{c} \curvearrowright \\ \uparrow \end{array} \right)_{+ve} \sum M_B = 0$$

$$-15(R_A) + 1872(16) = 0$$

$$\boxed{R_A = 1996.8 \text{ lb}}$$

$$\uparrow + \sum F_y = 0$$

$$-1872 + R_A + R_B = 0$$

$$-1872 + 1996.8 + R_B = 0$$

$$\boxed{R_B = +124.8 \text{ lb}}$$

$$K_x = 1$$

$$K_y = 1$$

Since the AISC column load table are computed assuming $(K/L)_y$ controls enter these table with effective length $(K/L)_y$

Thus enter with $P_u = 248$

$$W_{12} \times 45 \quad \phi_c P_n = 257 \quad \gamma_{xy} = 2.65.$$

$$W_{8} \times 40 \quad \phi_c P_n = 238 \text{ kips} \quad \gamma_{xy} = 1.73.$$

$$W_{10} \times 45 \quad \phi_c P_n = 267 \text{ kips} \quad \gamma_{xy} = 2.65.$$

Since $(KL)_x = 2(KL)_y$.

$$\text{if } \gamma_{xy} = 2(KL)_y.$$

if $\gamma_{xy} \geq 2$ then weak axis control and tabular loads give the correct answer.

Since γ_{xy} for $W_{8} \times 40$ is less than 2.

Strong axis bending controls. The strength may be obtained from the table by entering with the equivalent $(KL)_y$ that corresponds to $(KL)_y = 36 \text{ ft}$.

$$\frac{(KL)_x}{r_x} = \frac{(KL)_y}{r_y}.$$

$$\text{Equivalent } (K L)_y = (K L)_x / r_x / r_y$$

$$= 36 / 1.73$$

$$\Rightarrow 20.80 \text{ ft}$$

with $P_u = 248 \text{ k}$.

$$W_{12} \times 36 \text{ } \phi P_u = 247 \text{ k}$$

$$\text{and } r_x / r_y = 1.75$$

for $(K L)_y = 20.80 \text{ ft}$ The

$W_{12} \times 36$ has a design strength

ϕP_n at only 248 kip therefore

it is not acceptable for $W_{10} \times 45$
with $A_g = 13.3 \text{ in}^2$.

$$\lambda_c = \frac{K L}{r \lambda} \sqrt{F_y / E}$$

$$= \frac{89.6}{\lambda} \sqrt{\frac{36}{29,000}}$$

$$\Rightarrow \lambda_c = 1.0049 < 1.75$$

So for

$$F_{cr} = \left[(0.658)^{k_c} \right] F_y$$

$$\Rightarrow F_{cr} = 23.59 \text{ ksi}$$

$$\phi_{F_{cr}} = 0.85 \times 23.59$$

$$\phi_{F_{cr}} = 20.1 \text{ ksi}$$

$$\begin{aligned} \text{Now } \phi_{P_n} &= \phi_c F_{cr} A_g \\ &= 20.1 (13.3) \end{aligned}$$

$$= 267 \text{ k} > 248 \text{ k}$$

$$\phi_c P_n = 267 > 248 \text{ k} \rightarrow \text{OK}$$

So Use W10 x 45

Q2: -

GIVEN DATA: -

$$D.L = 1.5 \text{ kips}$$

$$L.L = 4.5 \text{ kips}$$

$$L = 52 \text{ ft}$$

$$\Delta = \frac{1}{300}$$

$$F_y = 36 \text{ ksi}$$

Sol: -

The moment of inertia required to satisfy the deflection limit can be calculated. How the value of Δ by this equation should be multiplied by 0.95 in case of concentrated load Table (5-4)

$$\text{Concentrated load of} = 1.5 + 4.5 = 6 \text{ kips}$$

$$M = 1/2 \times 6 \times 30 - 6 \times 15$$

$$M = 234 \text{ ft-kips}$$

$$I = 0.95 \times \frac{5}{48} \frac{M^2}{E \Delta}$$

$$= 0.95 \times \frac{5}{48} \frac{234 \times 12 (52 \times 12)^2}{29,000 (60 \times 12 / 300)}$$

$$\Rightarrow I = \frac{5193497088}{3340800}$$

$$I = 1555 \text{ in}^4$$

From manual's moment of inertia selection table we find the lightest section to be the W24x62 for which $I = 1550 \text{ in}^4$

The factored design load at each of the load points is $1.2(1.4 + 0.062 \times 15) + 1.6 \times 5.4$
 $= 11.4 \text{ kips}$ So

$$M = 1/2 \times 11.4 \times 26 - 11.4 \times 13$$

$$M = 29.64 - \text{kips}$$

$$L_p = \frac{300 c_y}{f_y} = \frac{300 \times 1.38}{\sqrt{36}}$$

$$L_p = 69 \text{ in}$$

Since this is less than 15ft spacing at the lateral support Eqn (5-26) applies. The following properties of the W24x62 are from the manual load-factor design selection table.

$$L_p = 5.8 \text{ ft} \quad \lambda = 17.2 \text{ ft}$$

$$\phi M_p = 413 \text{ ft-kips} \quad \phi M_y = 225.5 \text{ ft-kips}$$

Since $\phi = 0.90$

$$M_p = \frac{413}{0.9}$$

$$M_p = 459 \text{ ft-kips}$$

$$M_x = \frac{225}{0.9}$$

$$M_x = 2.83 \text{ ft-kips}$$

($C_b = 1.11$) Then from eqn (5-26)

$$M_n = 1.11 \left(459 - 176 \frac{15 - 5.8}{17.2 - 5.8} \right)$$

$$M_n = 352 \text{ ft-kips} < M_p = 459$$

design strength is $\phi M_n = 0.9 \times 352 = 316 \text{ ft-kips}$

Since this is less than the required value of 296-kips-ft the w24.62 is adequate if the deflection check is needed

The manual contains design-moment charts based on $C_b = 1$, which may be used if C_b is greater than unity. The charts should be entered with a moment equal to required value divided by C_b .

$$\text{if } C_b = 1.11, \phi$$

$$\phi M_n = \frac{296.9}{1.11} = 267.03 \text{ ft-kips}$$

If an unbraced length = 13ft
we find the lightest section
to be the W24x62

used W24x62

with Braced length = 13ft



Q3

Pg 10

Given data:

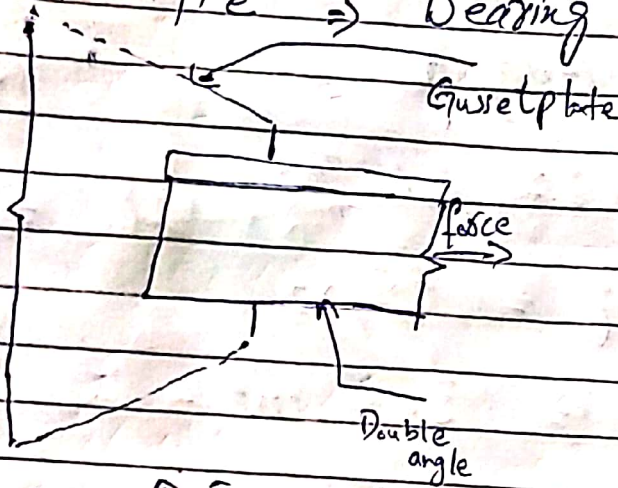
$$D.L = 50 \text{ k}$$

$$L.L = 150$$

$$B. dia = 3/4''$$

$$\text{length} = 18 \text{ Ft.}$$

Connection type \Rightarrow Bearing Type.



Solution \Rightarrow Total load = D.L + Live load

$$\Rightarrow 50 + 150$$

$$\Rightarrow 200 \text{ kips. } \text{or}$$

$$100 \text{ kip/angle}$$

For yielding at the gross area allowable stresses are $0.6 F_u = 0.6 \times 36$.

$$= 22 \text{ ksi}$$

For fracture at the net area allowable stresses are

$$0.5 F_u = 0.5 \times 58 \dots$$

$$\Rightarrow 29 \text{ ksi}$$

Since the connection is bolted so $A_g \neq A_n$

$$\text{Now } A_e = 0.85 A_n.$$

For Yielding

$$A_g \times 22 = 75$$

$$A_g = 75/22$$

$$= 3.4 \text{ in}^2$$

For Fracture

$$29 \times A_e = 75.$$

$$A_e = 2.59 \text{ in}^2$$

Assume 15% reduction in gross area for holes

$$\text{So } A_g = A_n / 0.85 = 3.58 \text{ in}^2.$$

For $5 \times \frac{3\frac{1}{2}}{2} \times \frac{7}{16}$

$$A = 3.53 \text{ in}^2 \approx 3.58 \text{ in}^2 \text{ OK}$$

$$r_x = 1.59 \text{ in} \quad \& \quad \text{with } \frac{3}{8} \text{ in G.P}$$

$$r_y = 1.47 \text{ in}$$

$$\frac{L}{r_{\min}} = \frac{18 \times 12}{1.47} = 146.93 \leq 300$$

OK

Design of Bolts:-

Using A325 bolts with threads included in shear plane.

$$A = 0.44 \text{ in}^2 \quad (\text{dia} = \frac{3}{4} \text{ in})$$

Allowable bolts are shear = 21 ksi
(Table 2.11 Gray Load)

Since the bolts are in double shear so

Allowable shear per bolts

$$= 2 \times 21 \times 0.44 = 18.5 \text{ kips.}$$

Allowable bearing on Two

$\frac{7}{16}$ " thick angle

$$\text{long angle leg} = 69.6 \times 2 \times \frac{7}{16} \times 0.75$$

$$= 45.68 \text{ kips} > 18.5 \text{ kips}$$

Allowable bolts-bearing stress

$$= 1.2 F_u = 1.2 \times 58 \Rightarrow 69.6 \text{ ksi}$$

Now No of bolts =

$$\frac{200}{18.5} = 10.81$$

10 Bolts

⇒ Design of Gusset plates

$$\text{Bearing stress} = 1.2 F_u = 69.6 \text{ ksi}$$

So

$$\text{Allowable bearing} = 69.6 \times 10 \times 0.75 \times t = 200$$

$$t = 0.38 \text{ in}$$

Use $\frac{3}{4}$ " G.P.

→ Checking various Limit
(states)

$$\text{Yielding} = 0.6 F_y A_g$$

$$= 0.6 \times 36 \times (10 \times 0.75)$$

$$\Rightarrow 162 \text{ kip} > 150 \text{ kips}$$

$$\text{Fracture} = 0.5 \times F_u \times A_e$$

$$\Rightarrow 0.5 \times 58 \times 0.85 [10 - (3/4) \times 2] \times 3/4$$

$$\Rightarrow 157 \text{ kips} > 150 \text{ k}$$

Check for tearing failure

$$L_c = \frac{2P}{F_{ut}} \Rightarrow$$

$$1.25 = \frac{2P}{58 \times 0.38}$$

$$P = 16.53 \text{ kips}$$

$$L = \frac{2P}{F_{ut}} + \frac{d_n}{2}$$

$$2 = \frac{2P}{58 \times 0.38} + \frac{3/4}{2}$$

$$2 = \frac{2P}{22.04} + \frac{3/4}{2}$$

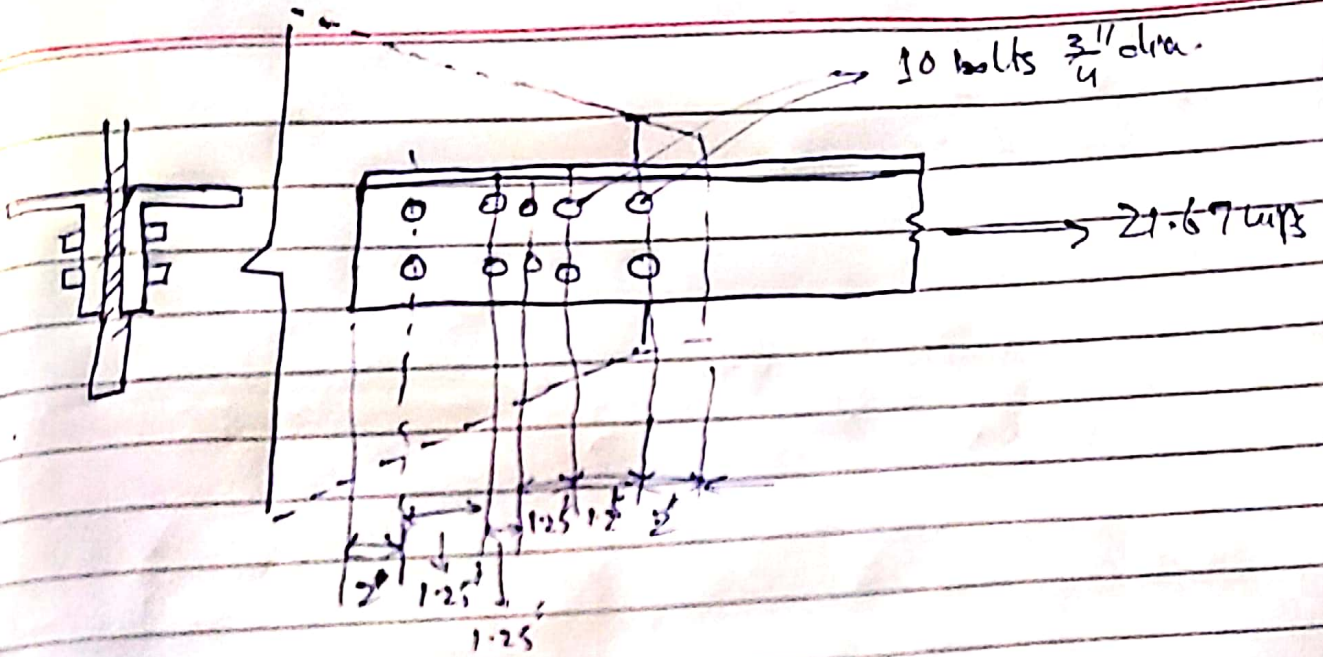
$$2 = \frac{2P}{22.04} + 0.37$$

$$2 \times 22.04 = 2P + 0.37$$

$$\frac{44.08}{2} = \frac{2P + 0.37}{2}$$

$$22.04 = P + 0.37$$

$$P = 21.67 \text{ KIP}$$



$t = 0.38$
use two $\frac{3}{4}$ C.P.

