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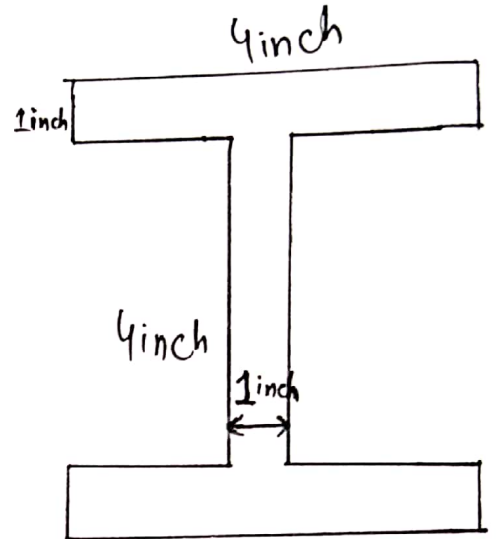
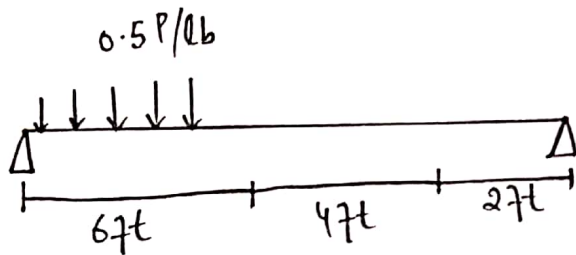
Section = A

Department = B.S civil
Engineering

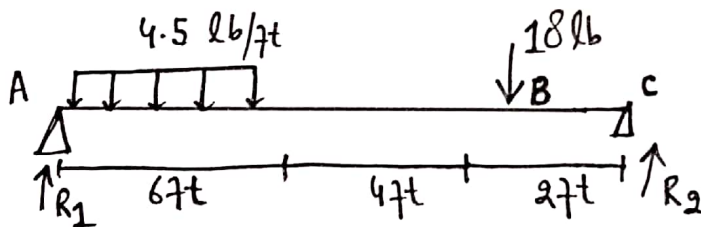
Paper = MOS 2

Question No 01Ans:-

Given Beam

Note:-Put the value of $P = 09$

So we have



First to find unknown reaction at the support apply equilibrium equation-

$$\sum \uparrow x = 0 \quad \text{i.e } R_3 = 0$$

$$\sum \uparrow y = 0 \quad \oplus \uparrow \ominus \downarrow$$

$$R_1 + R_2 = (4.5 \times 6) \text{ lb} + 18 \text{ lb}$$

$$\boxed{R_1 + R_2 = 45 \text{ lb}}$$

Next

$$\sum M_A = 0 \quad (\uparrow \oplus \quad \downarrow \ominus)$$

$$R_2 \times 12 - 10 \times 18 - (27) \times 3 = 0$$

$$12R_2 = 180 + 81$$

$$\frac{12R_2}{12} = \frac{261}{12}$$

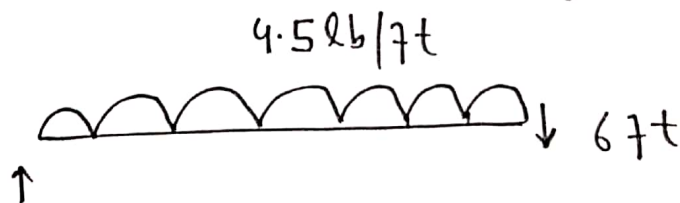
$$R_2 = 21.75$$

$$\textcircled{1} \quad R_1 + R_2 = 45$$

$$\Rightarrow R_1 = 45 - 21.75$$

$$\Rightarrow R_1 = 23.25$$

Now shear force at change point of
Beam



Shear at 6 ft from support

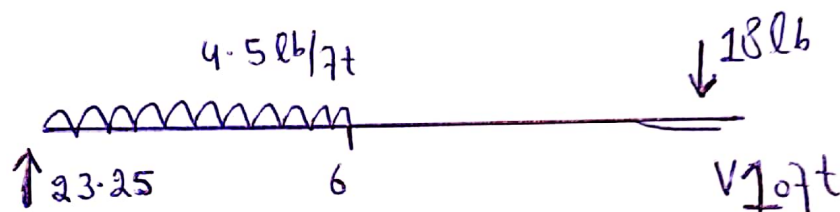
$$\sum F_y = 0 \quad \uparrow \oplus \quad \downarrow \ominus$$

$$23.25 - 4.5 \times 6 - V_6 \text{ft} = 0$$

$$\Rightarrow \boxed{V_6 \text{ft} = -3.75}$$

Now shear force at 10ft

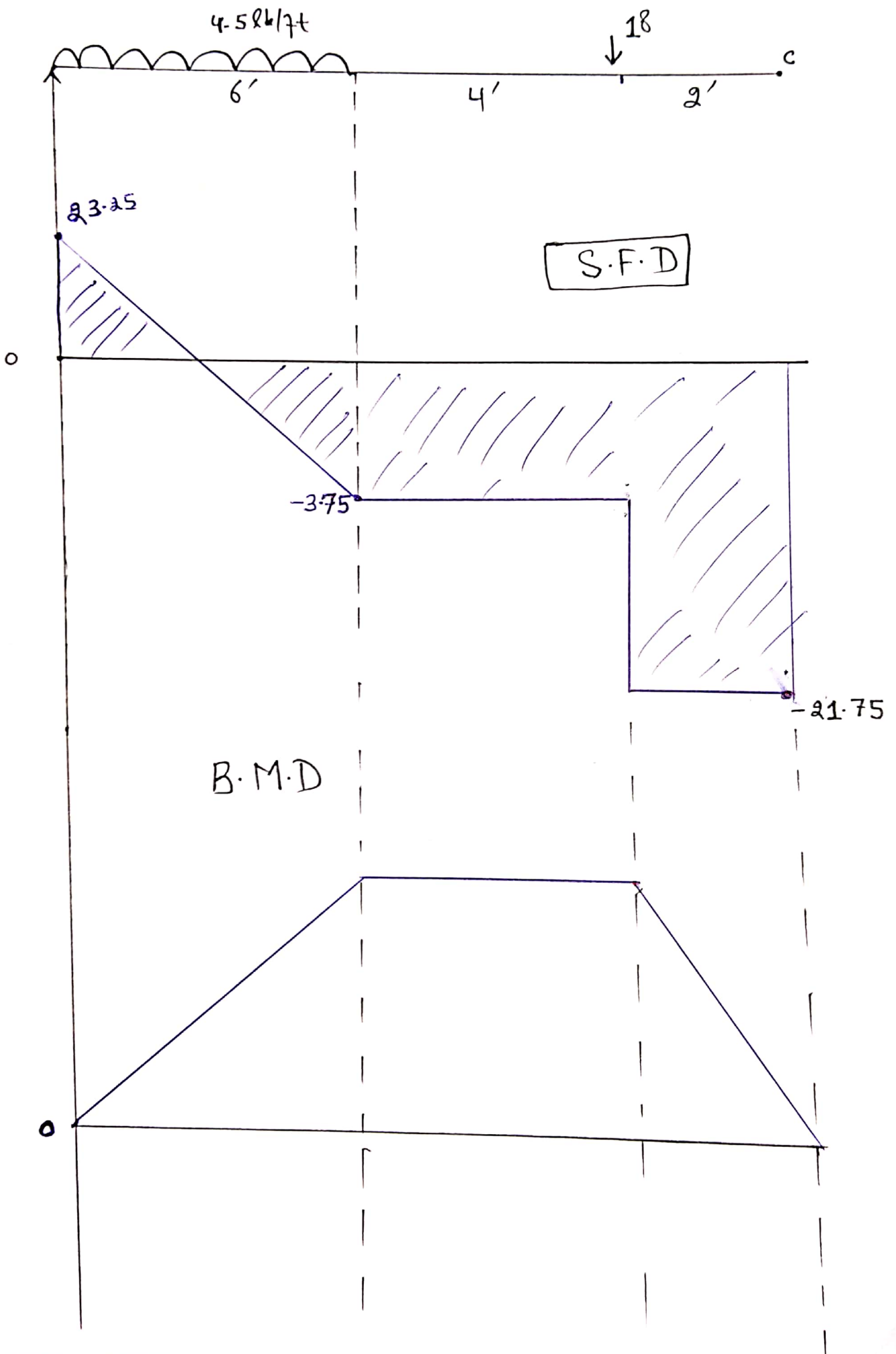
$$\sum F_y = \uparrow \oplus \downarrow \ominus$$



$$23.25 - 4.5 \times 6 - 18 - V_{10\text{ft}} = 0$$

$$\Rightarrow \boxed{V_{10\text{ft}} = -21.75}$$

Now draw shear force and
Bending moment diagram
we have

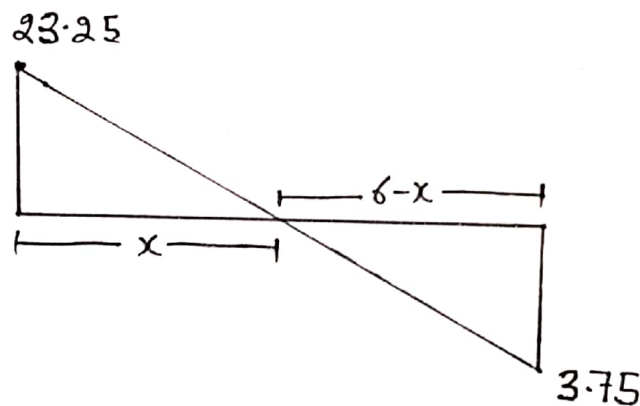


⇒ Point of Maximum Bending Moment

As we know that the point where shear force is maximum the bending moment is maximum so from point of zero shear corresponding point will have maximum Bending Moment.

From shear force diagram

We have



We know that

$$\frac{23.25}{x} = \frac{3.75}{6-x}$$

$$\Rightarrow (6-x)(23.25) = 3.75x$$

$$\Rightarrow 139.5 - 23.25x = 3.75x$$

$$\Rightarrow 139.5 = 3.75x + 23.25x$$

$$\Rightarrow \frac{27x}{27} = \frac{139.5}{27} \Rightarrow \boxed{5.166}$$

Now determine the value of moment at
5.157 ft



$$M_{5.157} - 23.25 \times 5.157 + (4.5 \times 5.157) \left(\frac{5.157}{2} \right) = 0$$

$$M_{5.157} - 119.90 + (23.2065)(2.5785) = 0$$

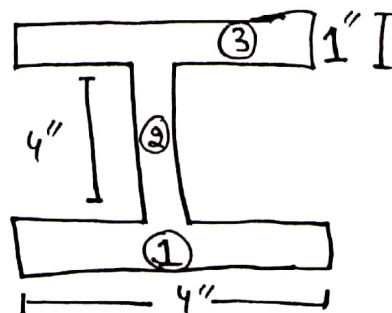
$$M_{5.157} = 119.90 - 59.837$$

$$M_{5.157} = 60.063 \text{ lbft}$$

For shear stress we have:-

$$f = \frac{VQ}{Ib}$$

So first we determine moment of inertia I for the given section of beam



As the given figure is symmetrical along
Both the axis

$$\text{So } \bar{x} = 4/2 = 2 \text{ inch}$$

$$\bar{y} = 6/2 = 3 \text{ inch}$$

$$\text{i.e. } (\bar{x}, \bar{y}) = (2, 3)$$

(center of Gravity)

extreme left and Bottom

$$\text{Area of Point ①} = 4 \times 1 = 4 \text{ inch}^2$$

$$\text{Area of Point ②} = 4 \times 1 = 4 \text{ inch}^2$$

$$\text{Area of Point ③} = 4 \times 1 = 4 \text{ inch}^2$$

Moment of ~~inertia~~ inertia about
x-axis (centroid I) I_{xx}

Determine Distance between C.G of the
whole section and corresponding Parts -

Let G_1, G_2, G_3 be in the Centre of Gravity
of Point ① ② and ③ and k_1, k_2, k_3 be the
distance b/w \bar{y} and y_1, y_2, y_3 Respectively -

So

$$k_1 = \bar{y} - y_1 \Rightarrow 3 - 0.5 = 2.5 \text{ inch}$$

$$k_2 = \bar{y} - y_2 \Rightarrow 3 - 3 = 0 \text{ inch}$$

$$k_3 = \bar{y} - y_3 \Rightarrow 3 - 0.5 = 2.5 \text{ inch}$$

So

$$I_{xx} = \frac{b_1 h_1^3}{12} + a_1 k_1^2 + \frac{b_2 h_2^3}{12} + a_2 k_2^2$$

$$+ \frac{b_3 h_3^3}{12} + a_3 k_3^2$$

$$I_{xx} = \frac{(4)(1)^3}{12} + 4(2.5)^2 + \frac{(1)(4)^3}{12} + a_2(0)$$

$$+ \frac{4(1)^3}{12} + 4(2.5)$$

$$I_{xx} = \frac{4}{12} + 25 + \frac{64}{12} + \frac{4}{12} + 25$$

$$I_{xx} = \frac{4 + (25) + 64 + 4 + (25)(12)}{12}$$

$$I_{xx} = 56 \text{ inch}^2$$

Now

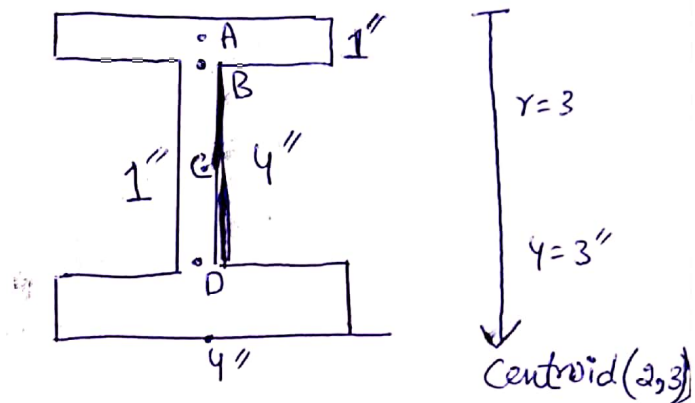
$$I_{yy} = \frac{b_1^3 h_1}{12} + \frac{b_2^3 h_2}{12} + \frac{b_3^3 h_3}{12}$$

$$I_{yy} = \frac{(4)^3(1)}{12} + \frac{(1)^3(4)}{12} + \frac{(4)^3(1)}{12}$$

$$I_{yy} = \frac{64 + 4 + 64}{12} = 11 \text{ inch}^2$$

Next find the shear stresses at various Point
we have

$$\tau = \frac{VQ}{Ib}$$



(i) shear stress at Point "A" i.e. At the
top fiber

$$\tau = \frac{VQ}{Ib}$$

$$V_{max} = 21.75$$

$$I = 67 \text{ in}^2$$

$$\therefore Q = A\bar{y}$$

$$\text{So } \tau = \frac{21.75(0)}{67(4)}$$

$$\tau = 0$$

Here $A=0$ Beam
no Area of the
Section exist above
Point A

$$\text{i.e. } Q = A\bar{y} = 0(\bar{y}) = 0$$

(ii) Shear stress at Point "B"

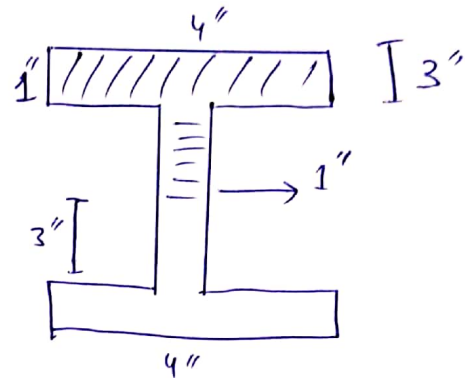
$$\tau = \frac{VQ}{Ib}$$

$$\tau = \frac{(21.75)(4 \times 1)(3 - 0.5)}{67 \times 4}$$

$$\tau = \frac{217.5}{268.0}$$

$$\tau = 0.8112 \text{ in}^2$$

shear stress at Point "c" i.e at
N.A



$$\tau = \frac{VQ}{Ib}$$

$$\tau = \frac{[(21.75) \times (4 \times 1 \times (3 - 0.5)) + (1 \times 2) (2 - 1)]}{67 \times 1}$$

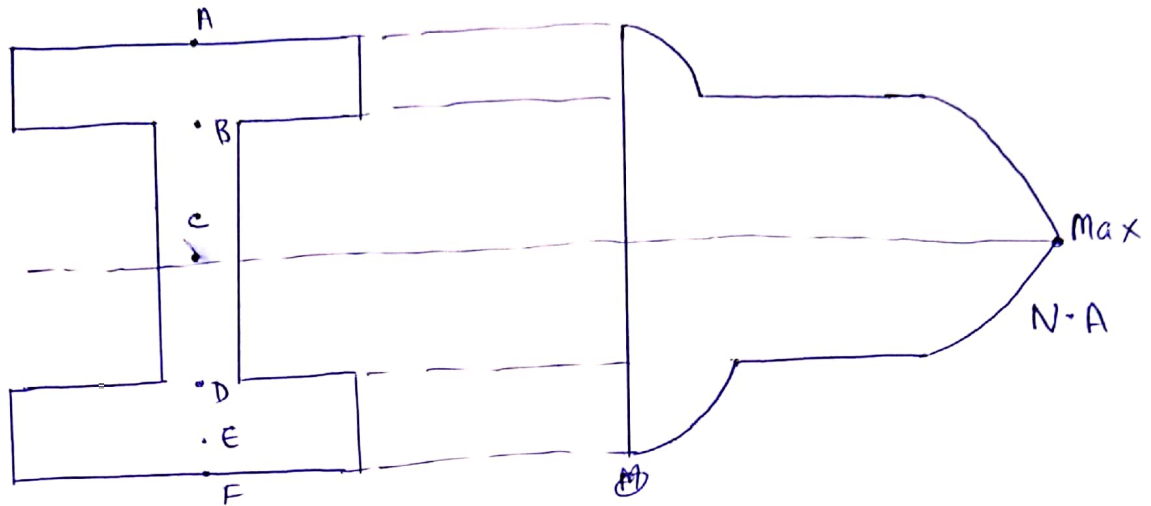
$$\tau = 3.276 \text{ lb/in}^2$$

(iv) shear stress at point D and E will be the same because of the symmetry.

Note:- The maximum shear stress value ~~will be~~ occur at the Neutral axis and minimum value at the top of the section -

Shear stress diagram

Pg#11



Flexural stress Determination

$$\sigma = \frac{My}{I}$$

① Flexural stress at the top Fiber Point A

$$\sigma = \frac{My}{I}$$

$$\sigma = \frac{60.063 \times 3}{67}$$

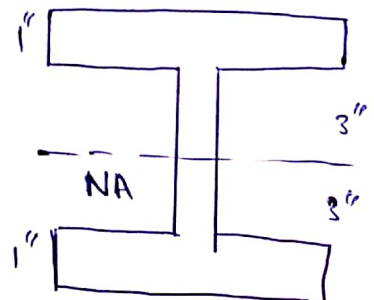
$$\sigma = 2.689 \text{ lb/in}^2$$

② Flexural stress at Point "B"

$$\sigma = \frac{My}{I}$$

$$\sigma = \frac{60.063 \times (3 - 0.5)}{67}$$

$$\sigma = 9.949 \text{ lb/in}^2$$



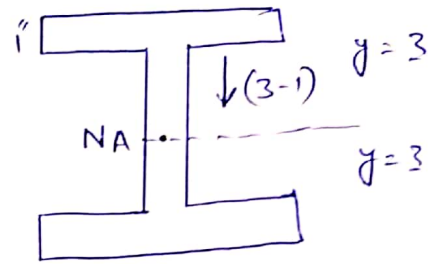
③ Flexural stress at Point "C"

Pg# 12

$$\sigma = \frac{My}{I}$$

$$\sigma = \frac{60.063 \times (3-1)}{67}$$

$$\sigma = 1.792 \text{ lb/in}^2$$

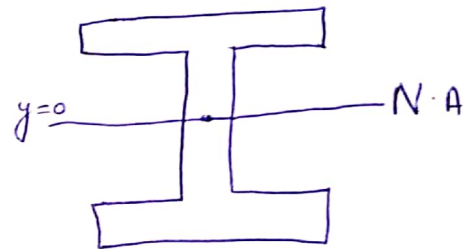


④ Flexural stress at Neutral axis (N.A)

$$\sigma = \frac{My}{I}$$

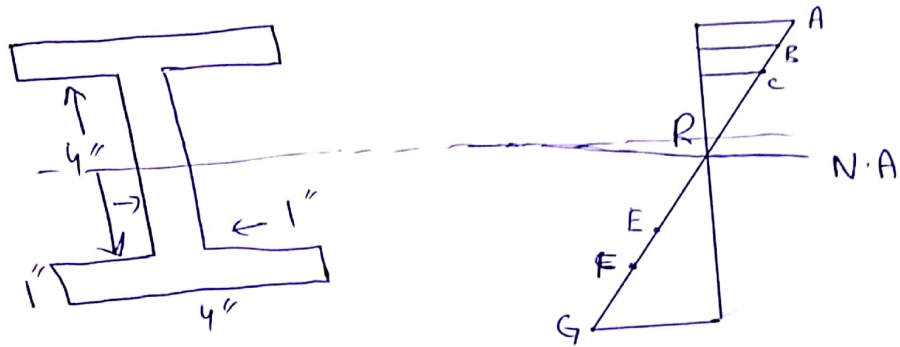
$$\sigma = \frac{60.063 \times 0}{67}$$

$$\sigma = 0 \text{ lb/in}^2$$



Flexural stress value at Point ~~C~~
E, F and G remain the same because of symmetry - The upper portion above the N.A shows tension and below the N.A show compression

Flexural Stress diagram



Stress State:-

Find stress state of a

Point element located 3ft from left support and 1 inch below from top fiber

Flexural stress at Point "C"

$$\sigma = 1.792 \text{ Psi}$$

Shear stress at Point "C"

$$\tau = 3.276 \text{ lb/in}^2$$

Consider Point "C" is a planar element



As the flexural stress is perpendicular to the cross section can be represented normal stress.

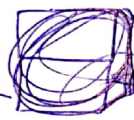
$\tau = 3.276 \text{ Psi}$ is compressive
Because Point "C" lies in compression
Zone of Beam Cross section



If Point c lies below the centroid
the stress should be tensile.



$$\tau = 3.276 \text{ Psi}$$



$$\tau = 3.276 \text{ Psi}$$

$$\sigma_x = 1.792 \text{ Psi}$$

Combine stress ~~on~~ on 2d element.

Find its Principle Stress:-

we have also find

$$\sigma_x = 1.792$$

$$\sigma_y = 0$$

$$\tau_{xy} = 3.276 \text{ Psi}$$

Principle stress equation

$$\sigma_{x,y} = \sigma_x - \sigma_y \pm \sqrt{\frac{(\sigma_x - \sigma_y)^2}{2} + \tau_{xy}^2}$$

$$\sigma_{x,y} = \frac{-1.792 + 0}{2} \pm \sqrt{\frac{(-1.792 - 0)^2}{2} + (3.276)^2}$$

$$\sigma_{x,y} = -0.896 \pm \sqrt{1.60 + 10.732}$$

$$= -0.896 \pm \sqrt{12.332}$$

$$\sigma_{x,y} = -0.896 \pm (3.51)$$

$$\sigma_x = -0.896 - 3.51 = \boxed{4.407}$$

$$\sigma_y = -0.896 + 3.51 = \boxed{2.614}$$

or First Find $QP = ?$

$$\tan 2QP = \frac{\tau_{xy}}{(\sigma_x - \sigma_y)/2}$$

$$\tan 2QP = \frac{3.276}{\frac{(-1.792 - 0)}{2}}$$

$$\tan 2QP = \frac{3.276}{-0.896}$$

$$\tan 2QP = -3.65$$

$$2QP = \tan^{-1}(-3.65)$$

$$2QP = -74.67$$

$$\boxed{QP = -37.33}$$

Put in General Equation

$$\sigma_{\max} = \frac{-1.792 + 0}{2} + \frac{-1.792 + 0}{2}$$

$$\cos 2(-37.33) + 3.276 \sin 2(-37.33)$$

$$\sigma_{p \max} = -43.362$$

Max in Plane shear stress in this case

$$\tan 2\theta_s = \frac{-(\sigma_x - \sigma_y)/2}{\sigma_{xy}}$$

$$\tan 2\theta_s = \frac{-(-1.792 - 0)/2}{3.276}$$

$$\tan 2\theta_s = \frac{0.896}{3.276}$$

$$2\theta_s = \tan^{-1}(0.273)$$

$$\frac{2\theta_s}{2} = \frac{15.26}{2}$$

$$\theta_s = 7.63$$

Put in these General equation

$$\tau_{x'y'} = -\left[\frac{\sigma_x - \sigma_y}{2}\right] \sin 2\theta + \sigma_{xy} \cos 2\theta$$

$$= -\left[\frac{-1.792 - 0}{2}\right] \sin 2(7.63) + 3.276 \cos 2(7.63)$$

$$= 0.238 + 24.98$$

$$\tau_{x'y'} = 25.218$$

To Draw Mohr's Circle

Centre Co-ordinate

$$(h, k) = \left(\frac{\sigma_x + \sigma_y}{2}, 0 \right)$$

$$\Rightarrow \left(\frac{-1.792 + 0}{2}, 0 \right)$$

$$\Rightarrow (-0.896, 0)$$

Radius of Mohr's Circle

$$r = \sqrt{\left(\frac{\sigma_x - \sigma_y}{2} \right)^2 + \tau_{xy}^2} = \sqrt{\left(\frac{-1.792 - 0}{2} \right)^2 + (3.270)^2}$$

$$= \sqrt{1.605 + 10.732}$$

$$= \sqrt{12.337}$$

$$r = 3.51240$$

