

ID : 7966

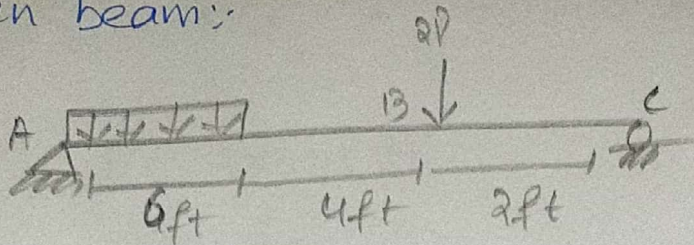
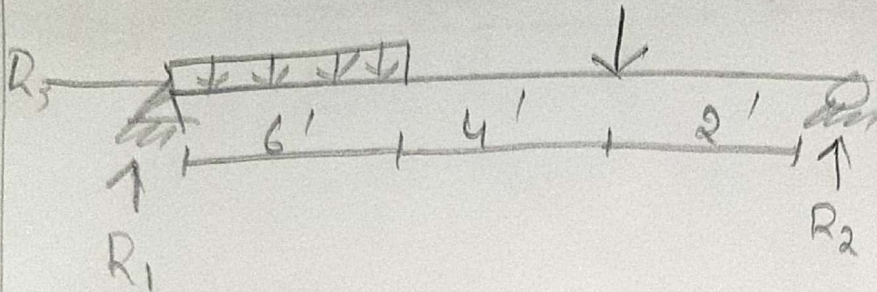
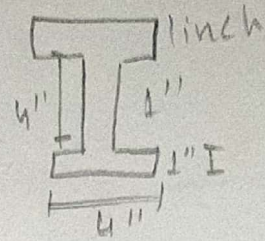
SECTION : B

SEMESTER : 4th

DEPT : CIVIL

①

Given beam:

Put the value of $P = 66$.

find the unknown reaction at the support
 apply equilibrium equation

$$\sum F_x = 0 \quad \text{i.e. } R_3 = 0$$

$$\sum F_y = 0 \quad \uparrow + \ominus$$

$$R_1 + R_2 = (10 \times 6) \text{ lb} + 66 \text{ lb}$$

$$R_1 + R_2 = 99 + 66$$

$$R_1 + R_2 = 165 \quad \text{--- (1)}$$

$$\text{Next } \sum M_A = 0 \quad \uparrow + \ominus$$

$$R_2 \times 12 - 10 \times \frac{6 \times 6}{2} - 66 \times 4 = 0$$

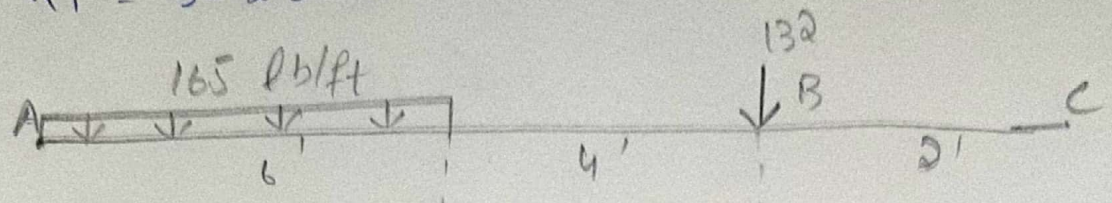
$$\Rightarrow 12R_2 = 180 + 264$$

$$\Rightarrow 12R_2 = 444 \text{ lb ft}$$

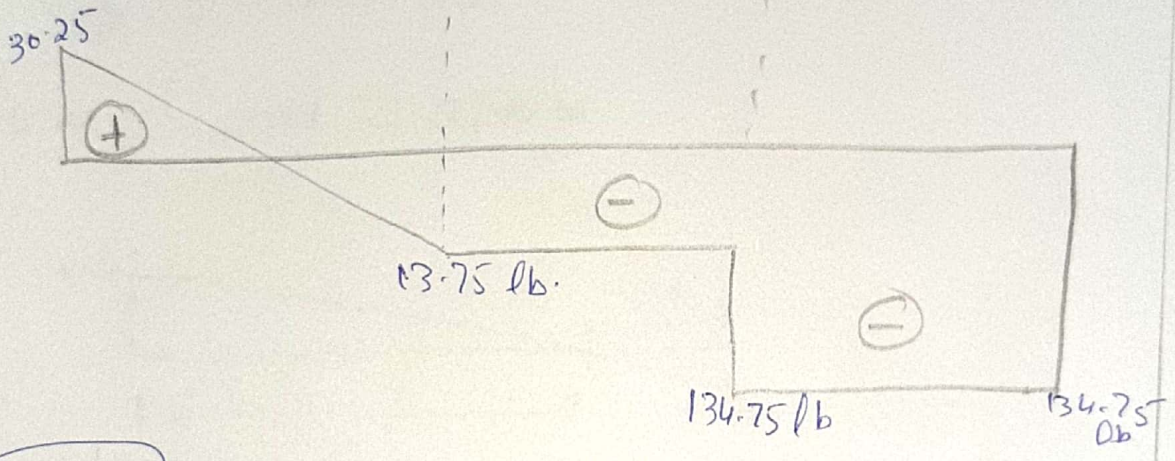
$$\Rightarrow 12R_2 = 444 \text{ lb ft}$$

$$\Rightarrow R_2 = 37 \text{ lb}$$

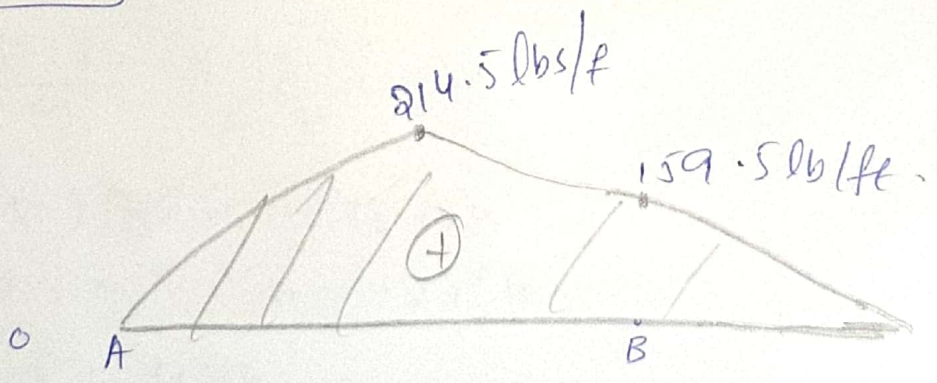
① $R_1 + R_2 = 165$
 $R_1 = 165 - R_2$
 $R_1 = 165 - 134.75 \text{ lb}$
 $R_1 = 30.25 \text{ lb}$



S.F.D



B.M.D



- Now shear force at change point of beam
Shear force at 6ft from left support

$$\sum F_y = 0 \uparrow + \downarrow -$$

$$30.25 - 16.5 \times 6 - V_{6ft} = 0$$

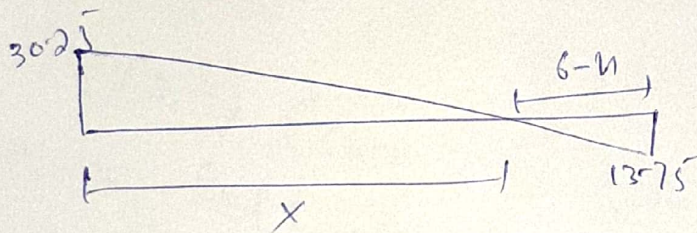
$$\Rightarrow V_{6ft} = 129.25 \text{ lb}$$

At 10ft:

$$30.25 - 16.5 \times 6 - 132 - V_{10ft} = 0$$

$$\Rightarrow V_{10ft} = -200.75$$

from Shear diagram



We know that

$$\frac{30.25}{n} = \frac{13.75}{6-n}$$

$$(6-n)(30.25) = 13.75n$$

$$181.5 - 30.25n = 13.75n$$

$$181.5 = 44n$$

$$n = 4.125 \text{ ft}$$

4

Now determine the value of moment at 4.125 ft

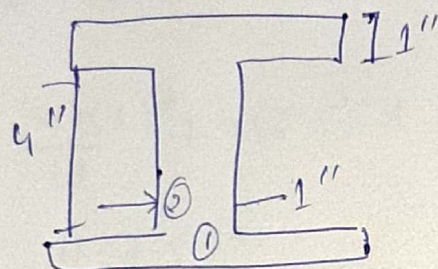
$$M_{4.125} = 30.25 \times 4.125 + (16.5 \times 4.125) \times \left(\frac{4.125}{2}\right)$$

$$\Rightarrow M_{4.125} = 124.7812 + 68.06 \cdot 140.37$$

$$\Rightarrow M_{4.125} = 15.597 \text{ lb ft}$$

for shear stress we have $\tau = \frac{VQ}{Ib}$.

Inertia = ?



$$\text{so } \bar{x} = 4/2 = 2 \text{ inch}, \quad \bar{y} = 4/2 = 2 \text{ inch}$$

i.e. $(\bar{x}, \bar{y}) = (2, 2)$ (centre of gravity),
from extreme left and bottom.

$$(1) = 4 \times 1 = 4 \text{ inch}^2$$

$$(2) = 4 \times 1 = 4 \text{ in}^2$$

$$(3) = 4 \times 1 = 4 \text{ in}^2$$

5

let G_1, G_2, G_3 be the centre of gravity of point ①, ②, ③ and k_1, k_2, k_3 the distances b/w \bar{y} and y_1, y_2, y_3 .

So

$$k_1 = \bar{y} - y_1 = 3 - 0.5 = 2.5 \text{ in}$$

$$k_2 = \bar{y} - y_2 = 3 - 3 = 0 \text{ in}$$

$$k_3 = \bar{y} - y_3 = 3 - 0.5 = 2.5 \text{ in}$$

So

$$\begin{aligned} I_{xx} &= \frac{b_1 h_1^3}{12} + a_1 k_1^2 + \frac{b_2 h_2^3}{12} + a_2 k_2^2 + \frac{b_3 h^3}{12} \\ &+ a_3 k_3^2 \end{aligned}$$

$$\begin{aligned} I_{xx} &= \frac{(4)(1)^3}{12} + 4(2.5)^2 + \frac{(1)(4)^3}{12} + a_2(0) + \frac{4(1)^3}{12} \\ &+ 4(2.5)^2 \end{aligned}$$

$$I_{xx} = \frac{4 + 10(25) + 64 + 4 + 12(25)}{12}$$

$$\Rightarrow I_{xx} = 56 \text{ in}^3$$

Now

$$I_{yy} = \frac{b_1^3 h_1}{12} + \frac{b_2^3 h_2}{12} + \frac{b_3^3 h}{12}$$

$$I_{yy} = \frac{64}{12} + \frac{4}{12} + \frac{64}{12}$$

$$I_{yy} = 11 \text{ in}^4$$

⑥ • At various point:- $\bar{I} = \frac{VQ}{Ib}$

i) At point A.

Centroid (2, 3)

$$V_{max} = 134.75$$

$$\bar{I} = \frac{VQ}{Ib}$$

$$\therefore Q = A\bar{Y}$$

$$A = 0$$

$$0(5) = 0.$$

$$\bar{I} = \frac{134.75(0)}{67(4)}$$

$$\bar{I} = 0.$$

ii) At point B

$$\bar{I} = \frac{VQ}{Ib}$$

$$\bar{I} = \frac{134.75 \times (4 \times 1) (3 - 0.5)}{67 \times 4}$$

$$\bar{I} = 5.027 \text{ lb/in}^2$$

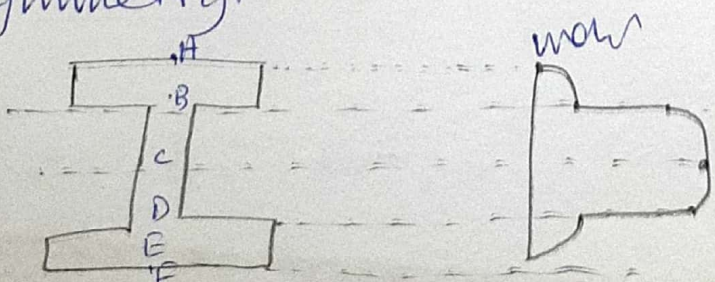
iii) At point C

$$\bar{I} = \frac{VQ}{Ib}$$

$$= \frac{134.75 \times [4 \times 1 \times (3 - 0.5) + (1 \times 2) (2 - 1)]}{67 \times 1}$$

$$\bar{I} = 20.14 \text{ lb/in}^2$$

iv) At point D:- At this point, D and E will be the same because of the symmetry.



Max = N.A.

⑧ Flexural stress determination.

$$\sigma = \frac{Mx}{I}$$

→ at the top fibre of point (A)

$$\sigma = \frac{my}{I}$$

$$\sigma = \frac{15.597 \times 3}{67}$$

$$M_{\max} = 15.597 \text{ lb-ft}$$

$$\sigma = 0.698 \text{ lb/in}^2$$

→ At (N.A):

$$\sigma = \frac{my}{I} = \frac{15.597 \times 0}{67}$$

$$\sigma = 0 \text{ lb/in}^2$$

Flexural stress value at point E, F, and G remain the same because of symmetry. Above the N.A is tension and below is compression.

→ At (B)

$$\sigma = \frac{15.597 \times (3 - 0.5)}{67} = 0.581 \text{ lb/in}^2$$

→ At (C)

$$\sigma = \frac{15.597 (3 - 1)}{67}$$

$$\Rightarrow 0.465 \text{ lb/in}^2$$

9) flexural stress diagram.

