

Name

Aamir Ghazoor

ID

7728

Section

A

Subject

Advance Fluid Mechanics

Teacher

Engr Abdul Waheed

Q No 1 Part (a)

Expression For Velocity Profile in laminar flow

$$\Delta s = h_c = \frac{\tau \Delta L}{\rho g}$$

From viscosity $\therefore u \frac{du}{dy}$
where u is value of velocity
at distance y from boundary.

$$\therefore y = h_0 - h$$

$$dy = -dh_0 - dh$$

$$dh_0 = \text{constant} = 0$$

$$dy = -dh$$

$$\tau = -\mu \frac{du}{dh}$$

Now

$$hL = -\mu \frac{du \Delta L}{h^2 dh}$$

$$du = \frac{-hL \tau}{2\mu L} dh$$

Integrating:

$$\int du = \frac{-hL \tau}{2\mu L} \cdot \frac{h^2}{2} + C$$

$$u = \frac{-hL \tau}{2\mu L} \cdot \frac{h^2}{2} + C$$

$$h_2 = 0 \quad h_1 = h_{\text{max}}$$

$$\therefore C = U_{\text{max}}$$

$$U = U_{\text{max}} - \frac{h_2 \gamma}{2 \mu L} \cdot \frac{h_2^2}{2}$$

$$U = U_{\text{max}} - K \gamma^2$$

Now as we know that $V=0$
when $h_2 = \gamma_0$

$$U_{\text{max}} = K \gamma_0^2 = \frac{h_2 \gamma}{4 \mu L} \cdot \gamma_0^2$$

It is also known as V_{av}

$$V_{\text{av}} = \frac{h_2 \gamma}{4 \mu L} \cdot \gamma_0^2 = \frac{h_2 \gamma}{16 \mu L} D^2$$

The average velocity may be taken as

$$V = \frac{V_{\text{av}} \cdot 0}{2} = 0.5 V_{\text{av}}$$
$$= \frac{h_2 \gamma D^2}{32 \mu L}$$

$$\text{As } \nu = \frac{\mu}{\rho}, \quad \mu/\rho = \nu$$

$$\frac{32 \mu L V}{h_1 D^2} \Rightarrow \frac{32 \mu L V}{\rho g \cdot D^2} \Rightarrow 32 \frac{\nu L}{g D^2} V$$

Q No # (01) (b)

Critical Reynold Number

Definition

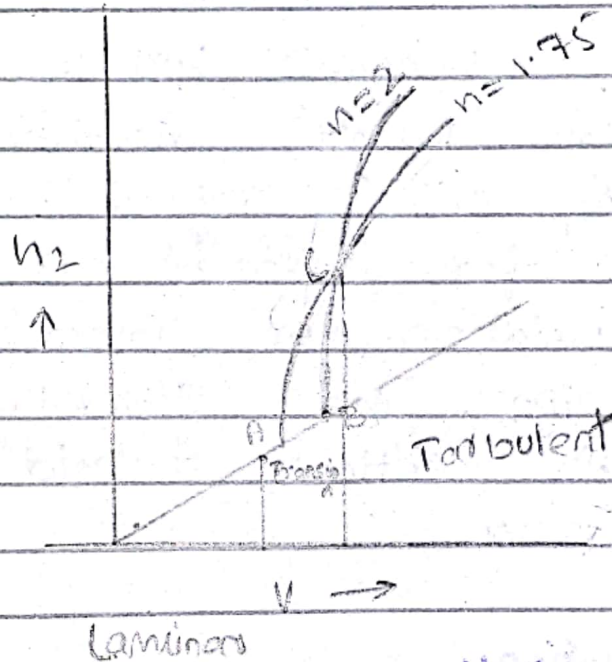
The Reynold number at the flow becomes turbulent is called the critical Reynold number, Re_{cr} . The value of critical Reynold number is different for different geometric and flow condition. For internal flow in circular pipe, The generally accepted value of critical Reynold's number is $Re_{cr} = 2300$.

Explanation:

If head loss in given length of uniform pipe is measured at different values of velocity, It will found that as long as velocity is low enough to secure laminar flow. The head loss due to friction will be directly proportional to velocity, but increase in velocity change flow from laminar to turbulent cause change in head loss. Thus if values are plotted, lines obtained with slope ranging about 1.75 to 2.

Thus for laminar drop of energy varies as U and for turbulent friction varies as U^n where n is 1.75 to 2.

Graphical Explanation:

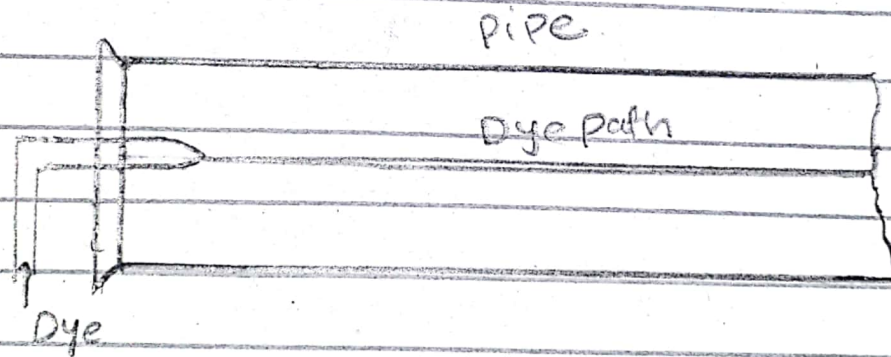


The upper critical Reynolds number corresponding to point B is intermediate and depends upon care taken to prevent initial disturbance. Its value is 4000, but normally it is possible for flow to be in straight after R at 2000. Thus lower value is much more definite than higher one and is dividing point. Thus lower value is true critical Reynolds number.

Evaluation of Critical Reynolds number

$$R = \frac{DVP}{\mu} = \frac{DV}{\nu}$$

For a circular pipe, we usually take the significant linear dimension L as the pipe diameter D .



Diagram

Q NO # 2

Given Data

Oil having $S = 0.7$
Kinematic viscosity = $1.8 \times 10^{-5} \text{ m}^2/\text{sec}$
Dia of pipe = $150 \text{ mm} = 0.15 \text{ m}$
Flow = $0.5 \text{ L/sec} = 0.0005 \text{ m}^3/\text{sec}$

Required Data

Centerline velocity =
Velocity at 10 mm from edge
Velocity at edge of pipe
Max shear stress at wall

Solution

As we know that

First we check flow is laminar or turbulent.

$$R = \frac{DV}{\nu} \rightarrow \text{A}$$

Now

$$V = \frac{Q}{A} = \frac{Q}{\frac{\pi d^2}{4}} \quad \because A = \frac{\pi d^2}{4}$$

$$V = \frac{0.0005}{\frac{\pi (0.15)^2}{4}}$$

$$V = 0.028 \text{ m/sec}$$

Now put V in eq (A)

$$R = \frac{(0.15)(0.028)}{1.8 \times 10^{-5}}$$

$$R = 233.37 < 2000 \text{ so the flow is (laminar)}$$

$$V_{cr} = 2V = 2 \times 0.028$$
$$V_{cr} = 0.056 \text{ m/sec}$$

As

$$u = u_{max} - K\gamma^2$$

at

$$\gamma = \gamma_0 = 0.075 \text{ m}, u = 0$$

Thus

$$0 = u_{max} - K\gamma^2$$

$$u_{max} = K\gamma^2$$

$$K = \frac{u_{max}}{\gamma^2} = \frac{0.056}{(0.075)^2}$$

$$K = 9.96$$

We get a equation

$$u = 0.056 - 9.96(r^2) \rightarrow \textcircled{B}$$

Velocity at 10mm from edge
 $r = 0.065 \text{ m}$

$$v = 0.056 - 9.96(0.065)^2$$

$$v = 0.014 \text{ m/sec}$$

Velocity at edge

$$r = 0.075 \text{ m}$$

$$v = 0.056 - 9.96(0.075)^2$$

$$v = -0.00002 \text{ m/sec say } v = 0$$

Similarly :-

$$f = \frac{64}{R} = \frac{64}{233.33}$$

$$f = 0.27$$

Shear stress at wall

$$\tau = \frac{f}{4} \rho \frac{V^2}{2} \quad \text{--- (1)}$$

putting values

$$\tau = \frac{0.07}{4} \times (0.7 \times 1000) \times \frac{(0.056)^2}{2}$$

$$\tau = 0.074 \text{ N/m}^2$$